

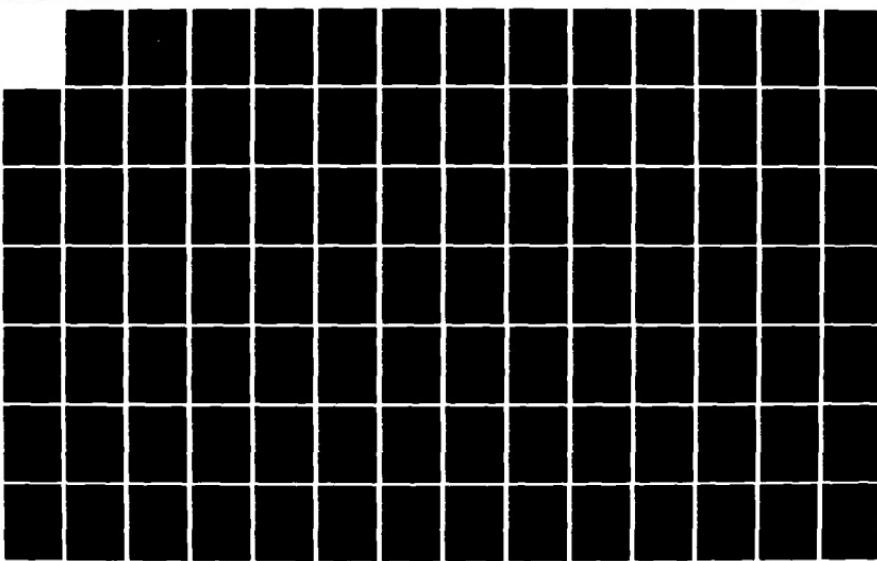
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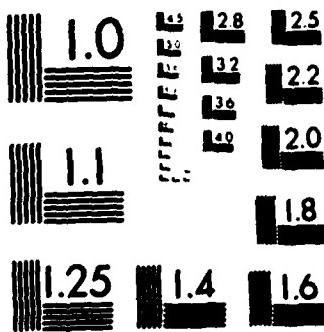
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THESIS

INFRARED DETECTION OF SURFACE VEHICLE,
CALCULATION USING ATMOSPHERIC MODEL LOWTRAN 6

by

Ioannis Eglefopoulos

December 1984

Thesis Advisor:

A. W. Cooper

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Infrared Detection of Surface Vehicle
Calculation Using Atmospheric Model LOWTRAN 6

by

Ioannis Egolfopoulos
Lieutenant, Hellenic Navy
B.S., Hellenic Naval Academy, 1975

Submitted in partial fulfillment of the
requirements for the degree of

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ABSTRACT

The thermal signature of a surface vehicle, "SPRUANCE" type destroyer, quantified by the signal to noise ratio of a thermal seeking missile head is parametrically investigated. The effects of the ship internal temperature, ship body paint emissivity, sky condition, sun elevation angle above the horizon, atmospheric profile, missile optics and flight altitude are examined in detail. Results show that both the ship body temperature and signal to noise ratio increase as the incident solar energy and the ship body paint emissivity increase and that the signal to noise ratio appears a peak for sun elevation angle in the range of 40° to 60°. Moreover the signal to noise ratio increases as the missile flight altitude decreases and keeping the other parameters constant higher values are found for the Midlatitude Summer atmospheric profile than for the Tropical atmosphere.

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I. INTRODUCTION

The basic function of a thermal-seeking-head missile is to track on a surface vehicle using the vehicle's radiant intensity. This operation is based on the fact that a moving vehicle shows a generally higher surface temperature than the environment. Reasons for this are the existence of operating engines, resulting in both increased engine room temperature and "hot" combustion products exiting into the atmosphere from the stack, the operation of a number of devices which result in heat generation, the requirement for certain compartment temperatures comfortable to the crew, and the incident solar energy on the painted metallic vehicle surfaces.

Exact information about the heat seeking missile characteristics is not available because of their classification. Hence our study will be a parametric one in order to examine the relative influence of a number of parameters on the vehicle thermal signature which is quantified by the signal to noise ratio of the thermal detector head. The range of the parameter values will be chosen using both experience and unclassified information.

Exceptional cases such as operation of the vehicle close to the coast or other vehicles, which could have similar or higher temperatures, will not be considered. At this point we must realize that the thermal interaction of the vehicle with the environment is a very complicated phenomenon, and its exact determination for the different parts of the ship body is outside of the scope of the present work. Instead we will consider a simplified case of uniform interaction for all the ship parts with the environment (convection, radiation, solar absorption energy) and will limit the source for

the thermal head detector environmental noise only to that of the sea water.

The main procedure steps of the present study will be :

To create a simplified model for the calculation of the vehicle body temperature and to show what parameters affect it and in what direction.

To determine the "hottest" vehicle area and combining with the operating characteristics of the missile to calculate the dimensions of the tracked area of the vehicle.

To calculate the total radiant intensity of the tracked area of the vehicle in a certain wavelength range, where the thermal detector head is operating.

To calculate the atmospheric transmittance, using LOWTRAN 6, for each given combination of the parameters of our interest in the given detector wavelength range and consequently the radiant heat flux detected by the thermal head.

To create an appropriate detector model that could show optimum operation for the conditions of our problem, and finally to calculate the creating signal-to-noise ratio which quantifies the vehicle thermal signature and describes the detector operational quality.

To discuss the relative influence of all the pertinent parameters and to propose certain considerations about both the vehicle and the missile.

Without expecting significant quantitative importance for our results, due to the simplifications that have been made, their qualitative importance must be more remarkable.

Parameters that will play significant roles in our study will be :

- vehicle internal temperature,
- vehicle body paint emissivity,
- sky condition,
- sun elevation angle above the horizon,

atmospheric profile,
missile flight altitude,
instantaneous Field of View (I.F.V.) created by the
detector's optic system,
detector material.

The present work consists of five Chapters. In the first Chapter we state the assumptions of the problem, we calculate the vehicle body temperature, and we determine its "hottest" area which is tracked by the detector. In the second Chapter we use the LOWTRAN 6 program in order to determine the atmospheric transmittance and in the third Chapter we calculate the total vehicle and background radiant intensity incident on the tracker aperture. In the fourth Chapter we calculate the detector signal-to-noise ratio for all the possible combinations of the parameters, and finally in the fifth Chapter we discuss the relative importance of each parameter and we suggest ways that could improve both the vehicle thermal signature and the detector operation.

II. SHIP MODELING

In order to create an exact ship model, it is necessary to acquire a significant amount of empirical data. For the present study, and for several reasons, the available data are limited, and are based only on the unclassified literature given in the first six references, with relatively reasonable assumptions.

A. ASSUMPTIONS

All compartments of the ship have a uniform temperature of $20^{\circ}\text{C}=293.15\text{K}$ except the main engine rooms 1 and 2, which have a temperature of $30^{\circ}\text{C}=303.15\text{K}$.

The stack exit temperature is uniform and we study the heat radiation from the exit plane neglecting the plume radiation. Reasons for doing this are the non-uniformity of the plume geometry and temperature distribution, and the fact that it is cooled significantly in a relatively short distance from the exit plane [Ref. 3]. We choose as stack exit temperature $306^{\circ}\text{F}=152.2^{\circ}\text{C}=425.4\text{K}$

In order to account for the heat convection from the ship to the air we will use the relative wind-speed of 30 Knots. For this case we will have forced convection with a heat transfer coefficient in the range of 5 to 10 (BTU/H FT²R) or equivalently 28.4 to 56.8 (W/M²K). We will not account for the heat leakage from the ship body to the sea water or for the heat loss due to the vaporization effect on the wet surfaces above sea level. For this reason and in order to be closer to reality we choose for our heat convection losses calculation the value of 10 (BTU/H FT²R) or 56.8 (W/M²K) as representative for the convection heat transfer coefficient.

The present study will be in Mediterranean atmospheric conditions and more specifically for the area of the Aegean Sea. We will use two atmospheric profiles i.e. Midlatitude Summer for months October through April and Tropical for months May through September. The corresponding ambient temperatures are $21^{\circ}\text{C}=294.2\text{K}$ and $26.6^{\circ}\text{C}=299.7\text{K}$ respectively [Ref. 6].

B. GEOMETRY AND MATERIAL OF THE SHIP

In Figures 2.1 and 2.2 you will find the ship side and top view respectively with the appropriate dimensions.

The material of the ship lower body construction is constructional carbon steel with thickness 13.5mm and thermal conductivity 54 (W/M K) [Ref. 9].

The material of ship body superstructure is duraluminum with thickness 3mm and thermal conductivity 164 (W/M K).

These small values of the thickness and the high values of the thermal conductivities for both materials result in negligible temperature gradients inside the materials. Hence it is reasonable to assume a uniform temperature throughout the thickness of both materials.

Both structures are painted with gray paint i.e. the emissivity and absorptivity will be spectrally independent and equal between them. A typical value for oil paint emissivity - absorptivity is 0.94 [Ref. 10:p.07] and for TiO_2 gray paint is 0.87 [Ref. 4:p.278] both for temperatures 300K. From these values we observe that the available data give us paint emissivity - absorptivity values in the range of 0.90. Based on the facts that these values play a significant role in the determination of the ship body temperature and that intensive research resulted in the production of new paints with half the values of the already existing ones [Ref. 7] we will use two (2) different values i.e. 0.80,

0.55. Therefore we cover the useful range of the existing and the newly developed paints. At this point we must note that we don't have any available information about selectively absorbing paints that could exist and give the minimum possible thermal signature and our assumption about the equality between emissivity and absorptivity is inevitable as a first approximation. Moreover the solution of the problem for two (2) different emissivities constitutes a good research point for the relative influence of this factor.

The emissivity of the interior surface of the ship body will be approximated as that of stainless steel which is 0.44 [Ref. 4:p.278].

C. SOLAR HEAT FLUX IN THE SEA LEVEL SURFACE

In Figure 2.3 you will find the illuminance levels on the surface of the earth due to the sun for three different sky conditions i.e. Unobscured Sun, Sun with Light Clouds and Sun with Heavy Storm Clouds as a function of the elevation angle above the horizon. These three sky conditions and the elevation angle above the horizon will be two more parameters in the solution of our problem.

Since the given values in Figure 2.3 are in photometric units (Lux) and we want to convert into useful radiometric units (W/M^2) we have to determine a conversion factor.

We know that the maximum incident solar flux on the earth surface is approximately $900 \text{ (W}/\text{M}^2)$ [Ref. 10:p.11]. This value corresponds to $1.15 \cdot 10^3 \text{ (Lux)}$ and therefore the conversion factor will be:

$$CF = 900 / 1.15 \cdot 10^3 = 5.826 \cdot 10^{-3} \text{ (W}/\text{M}^2\text{Lux})$$

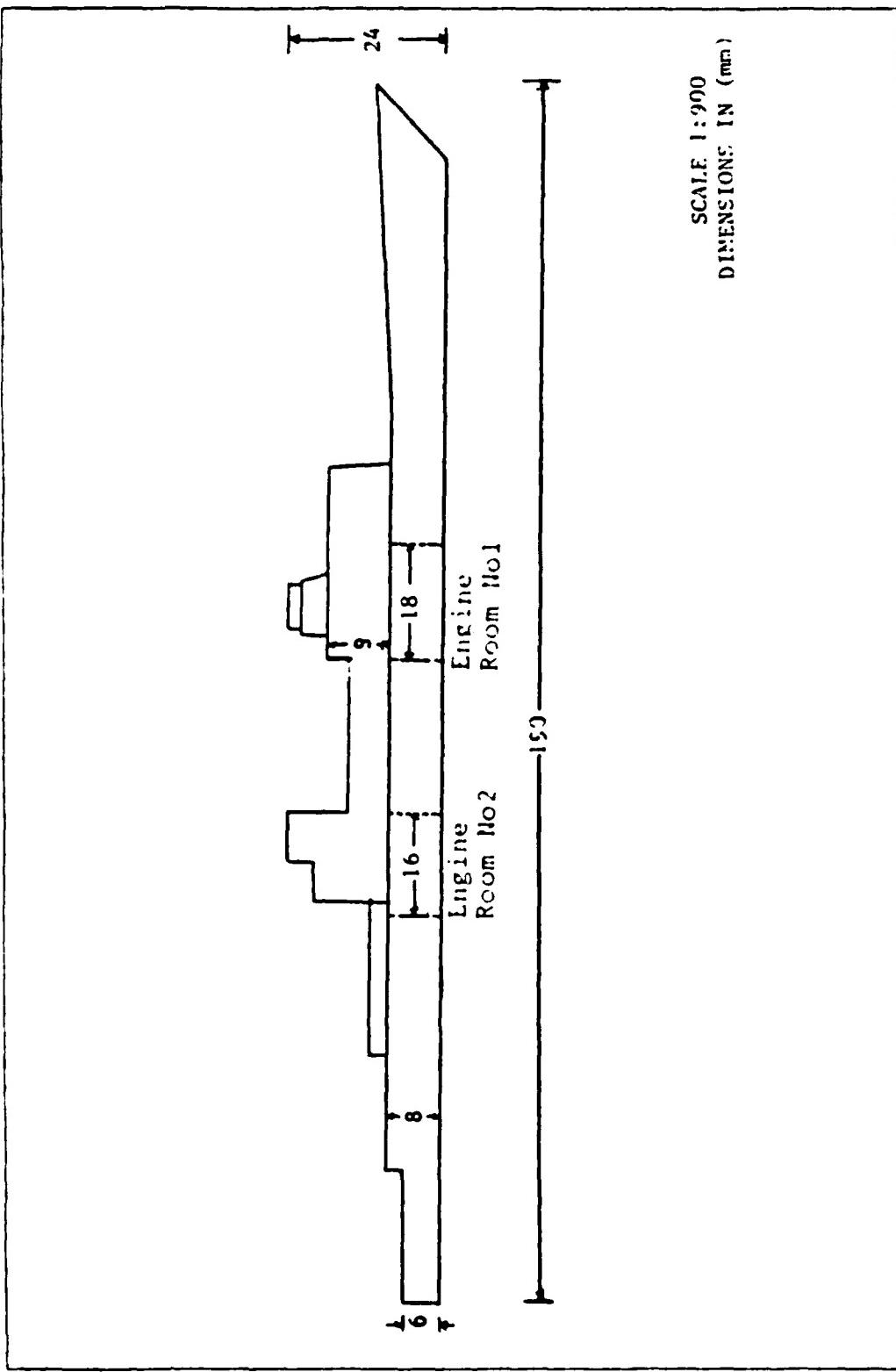


Figure 2.1 Ship Side View

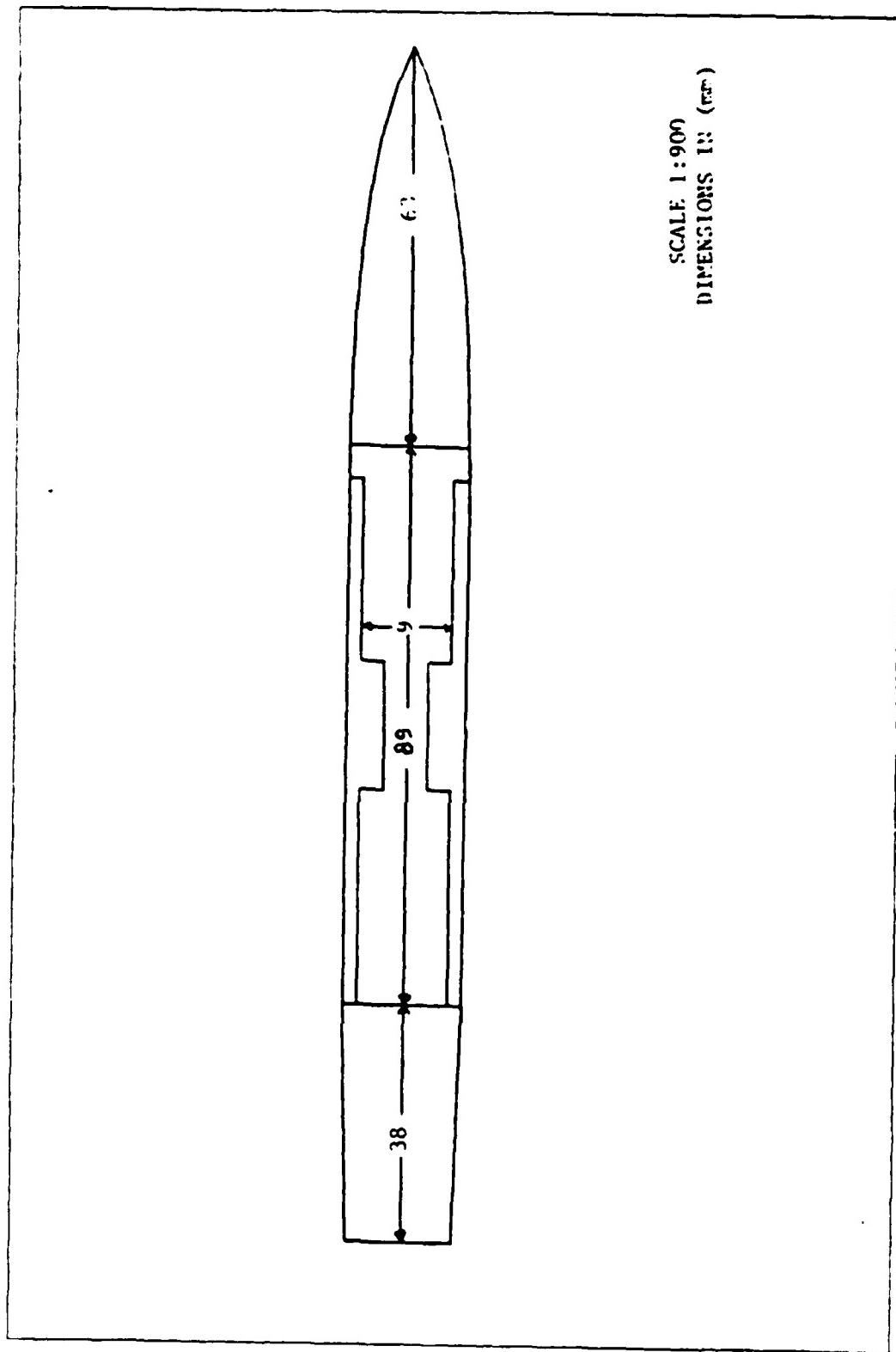


Figure 2.2 • Ship Top View

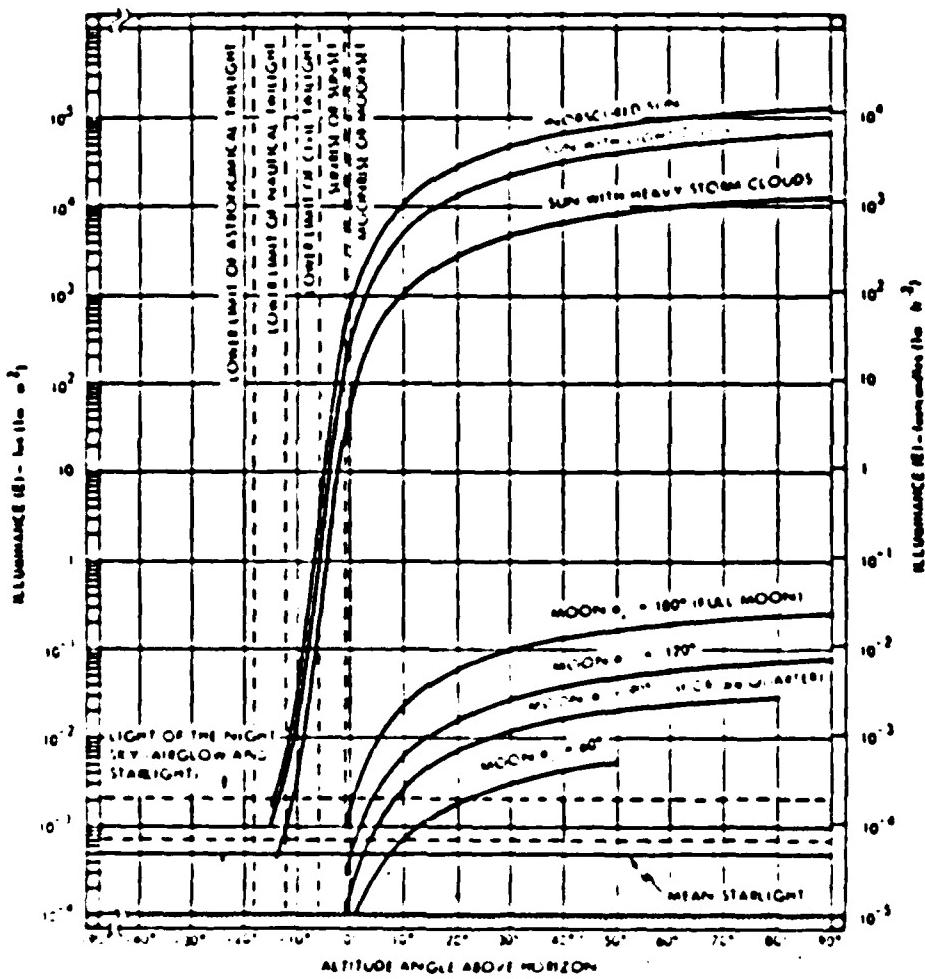


Figure 2.3 Illuminance Level on the Surface of the Earth [Ref. 5]

TABLE I
Normal Component of Solar Heat Flux on a Surface at Sea Level

ELEV.	ANGLE	UNOBSCURED SUN	SUN WITH LIGHT CLOUDS	SUN WITH HEAVY STORM CLOUDS	HEAVY STORM
34.		367.89	367.89	367.89	367.89
35.		391.67	391.67	391.67	391.67
377.02		391.67	391.67	391.67	391.67
377.02		175.70	175.70	175.70	175.70
40.98		445.45	445.45	445.45	445.45
40.98		445.45	445.45	445.45	445.45

ELEV.	ANGLE	UNOBSCURED SUN	SUN WITH LIGHT CLOUDS	SUN WITH HEAVY CLOUDS	STORM
71					
72					
73					
74					
75					
76					
77					
78					
79					
80					
81					
82					
83					
84					
85					
86					
87					
88					
89					
90					
		859.18	459.20	459.30	90.64
		864.54	464.54	464.54	91.64
		869.47	469.47	469.47	92.64
		874.40	474.40	474.40	93.64
		879.33	479.33	479.33	94.64
		884.26	484.26	484.26	95.64
		889.19	489.19	489.19	96.64
		894.12	494.12	494.12	97.64
		899.05	499.05	499.05	98.64
		904.00	504.00	504.00	99.64
		909.00	509.00	509.00	100.64
		914.00	514.00	514.00	101.64
		919.00	519.00	519.00	102.64
		924.00	524.00	524.00	103.64
		929.00	529.00	529.00	104.64
		934.00	534.00	534.00	105.64

Using this conversion factor and the values of Figure 2.3 we obtain the normal component of the solar heat flux in (W/M^2) on a surface at sea level. Table 1 demonstrates the above values.

In order to calculate the values of the solar heat flux falling obliquely on a surface we must introduce the incident angle θ and multiply the values of Table 1 by $\cos\theta$.

For a vertical face the angle θ is identical with the elevation angle above the horizon. For a horizontal face i.e. the deck in our case, the angle θ is the complement of elevation angle above the horizon.

D. CALCULATION OF SHIP BODY TEMPERATURE

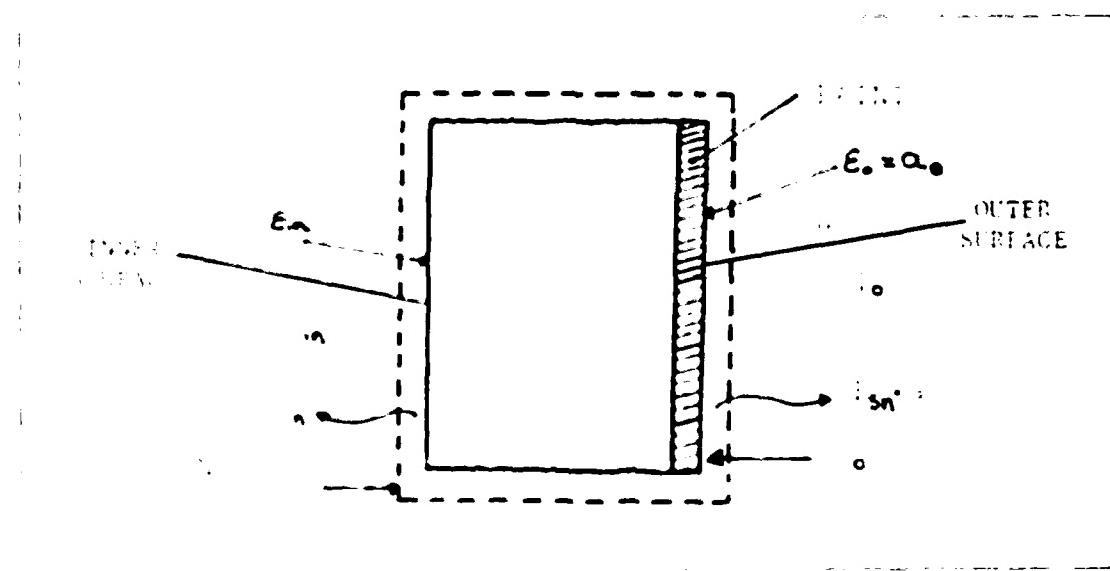


Figure 2.4 Control Volume for Determination of Ship Body Temperature

In Figure 2.4 you will find the appropriate control volume of the ship body and the pertinent energy flows that are related with it. For simplicity the control volume has

unit area exposed to both the environment and the ship interior. More specifically:

ϵ_e : paint emissivity

a_e : paint absorptivity ($\epsilon_e = a_e$)

ϵ_m : emissivity of the interior surface. ($\epsilon_m = 0.44$)

T_e : ambient temperature

T_m : ship compartment temperature

E_{in} : normal component given by Table 2

Q_o : net heat exchange of control volume with the environment

Q_m : net heat exchange of control volume with the interior of the ship

h : convection heat transfer coefficient with the air

The absorbed solar heat flux is $E_{in} \cdot a_e$. The net heat exchange with the environment is due to heat radiation and heat convection i.e.

$$Q_o = \epsilon_e \sigma (T^* - T_e^*) + h(T - T_e) \quad (2.1)$$

The net heat exchange with the interior is due only to heat radiation i.e.

$$Q_m = \epsilon_m \sigma (T^* - T_m^*) \quad (2.2)$$

Energy balance of the control volume under thermal equilibrium gives:

$$E_{in} \cdot a_e = \epsilon_e \sigma (T^* - T_e^*) + h(T - T_e) + \epsilon_m \sigma (T^* - T_m^*) \quad (2.3)$$

or

$$F(T) = (\epsilon_e + \epsilon_m) \sigma T^* + hT - \sigma(\epsilon_e T_e^* + \epsilon_m T_m^*) - hT_e - E_{in} \cdot a_e = 0 \quad (2.4)$$

Equation 2.4 must be solved numerically. We will use the Steffensen Recursive Method [Ref. 11]. The idea of this method is that starting with an x close enough to the exact solution S of the equation $F(x)=0$ then the sequence:

$$x_{n+1} = x_n - \left[f^2(x_n) / f(x_n + f(x_n)) - f(x_n) \right] \quad (2.5)$$

will converge very fast to the exact solution S .

In our case the equation is $F(T)=0$ and we are looking for the value of T that satisfies it.

In order to obtain an initial value so that our sequence will converge we use the value:

$$T_m = [(E_m \cdot a_0 + \epsilon_0 T_0^4 + \epsilon_m T_m^4) / (\epsilon_0 + \epsilon_m)]^{1/4} \quad (2.6)$$

The above initial value of T does not account for the heat convection and for our purpose gives acceptable initial values. Finally we stop the recursive formula when $|T_{m+1} - T_m|$ is less than or equal to 10^{-6} .

In Appendix A, Tables 5 through 18 demonstrates the ship body temperature for both horizontal and vertical faces, for all the combinations of two (2) atmospheric profiles, three (3) different sky conditions, two (2) internal (T_m) temperatures and two (2) paint emissivities.

E. DETERMINATION OF THE SHIP "HOTTEST" AREA

From the calculations in section D, we observe that the higher ship body temperatures happen to be in the region of the two engine rooms (higher internal temperature).

From the ship geometry, it is obvious that the dimensions of the engine room No 1 are greater than those of engine room No 2. Therefore the total radiant intensity (W/Sr) of engine room No 1 will be greater since for both regions we

have the same normal components of radiance ($\text{W/cm}^2\text{Sr}$). In addition to that to each engine room there corresponds the same stack geometry also going a 'hot' region too.

Under this simplified assumption we conclude that for a heat seeking missile head the area of engine room No 1 from the stack exit all the way down to sea level is the most significant region.

At this point it is necessary to couple the ship with some characteristics of the missile detector operation.

Without getting involved in details of the detector operation, which will be studied in later sections, we will determine the detector Instantaneous Field of View (I.F.V.), so that we will know exactly the dimensions of the 'hottest' area which will be tracked by the detector and therefore we will be able to calculate the amount of radiant energy that will be transmitted to the missile.

F. DETERMINATION OF THE DETECTOR INSTANTANEOUS FIELD OF VIEW

The following range of data is based on our experience and research, aiming to calculate a reasonable and useful I.F.V. In Figure 2.5 you will find the corresponding geometry.

The range of the missile flight altitude will be from 50m to 100m above sea level. Since we want to reduce as much as possible the background noise, the detector must have an inclination with respect to the horizontal. We choose the range for this inclination (angle φ) from 2° degrees to 5° degrees.

From general consideration for the detectors [Ref. 4] we know that I.F.V. is a function of the 'hottest' ship area that the detector has to track.

For the creation of I.F.V. it is necessary that the detector accepts radiation from inside an area which is related to an angle 2δ generated by the detector mirror system as in Figure 2.5 . From the same figure we conclude the following:

$$\theta = 90^\circ - \varphi - \delta$$

$$BC = AB + \tan\theta$$

$$AC = (AB^2 + BC^2)^{1/2}$$

$$CF = AC \cdot \sin\delta$$

$$CG = 2CF = 2AC \cdot \sin\delta = SIDE$$

$$AF = CG/2\tan\delta = PATH$$

Where:

CG represents the side of the I.F.V.

AF represents the distance or path length between detector and the I.F.V.

AB = H : flight altitude

For simplicity we will consider that the shape of I.F.V. is square with side equal to CG of Figure 2.5

In Appendix B Tables 19 through 29 demonstrate the values of PATH and SIDE for all the possible combinations of φ , δ and H. Program SHIP 1, in Appendix E, has been used for the calculation of the above values.

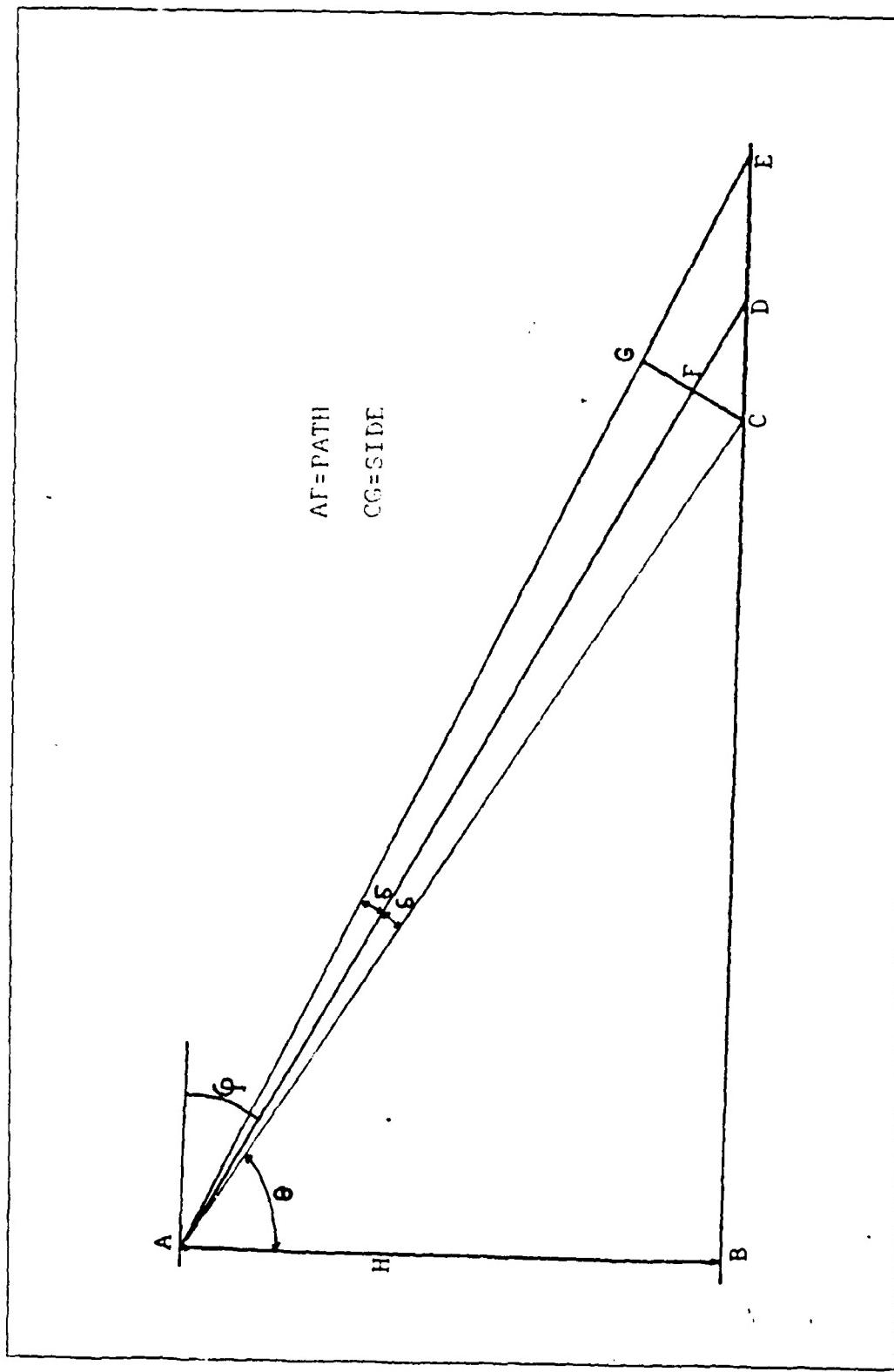


Figure 2.5 Determination of Instantaneous Field of View

G. DETERMINATION OF THE DIMENSIONS OF THE TRACKED SHIP AREA

In the previous section we have shown that the detector will track on the area of engine room No 1.

In Figure 2.6 you will find a three-dimensional sketch for the engine room No 1 area. Before we start the calculations of the different surfaces we will explain why the side of I.F.V. is greater than the width of the engine room No 1 which is $W_1 = 1,620\text{cm}$. We consider as useful that the I.F.V. will include the whole area between sea level and the stack exit plane. The total height of this dimension is:

$$H_1 + H_3 + H_4 + H_5 = 720 + 900 + 360 + 180 = 2,160\text{cm}$$

From the Tables 13 to 24, we choose for each flight altitude and for constant $\delta = 0.56^\circ$ the minimum SIDE that is greater than or equal to 2.160cm

Table 2 demonstrates the results of this choice

Based on Figure 2.6 we will now determine the characteristics of each individual surface.

$$A_{1L} = W_1 \cdot H_1$$

Vertical face

$$T_m = 303.15\text{K}$$

$$A_{1L} = (\text{SIDE} - W_1) \cdot H_1$$

Vertical face

$$T_m = 293.15\text{K}$$

$$A_{1L} = W_1 \cdot H_1$$

Horizontal face

$$T_m = 303.15\text{K}$$

$$A_{1L} = (\text{SIDE} - W_1) \cdot H_1$$

Horizontal face

$$T_m = 293.15\text{K}$$

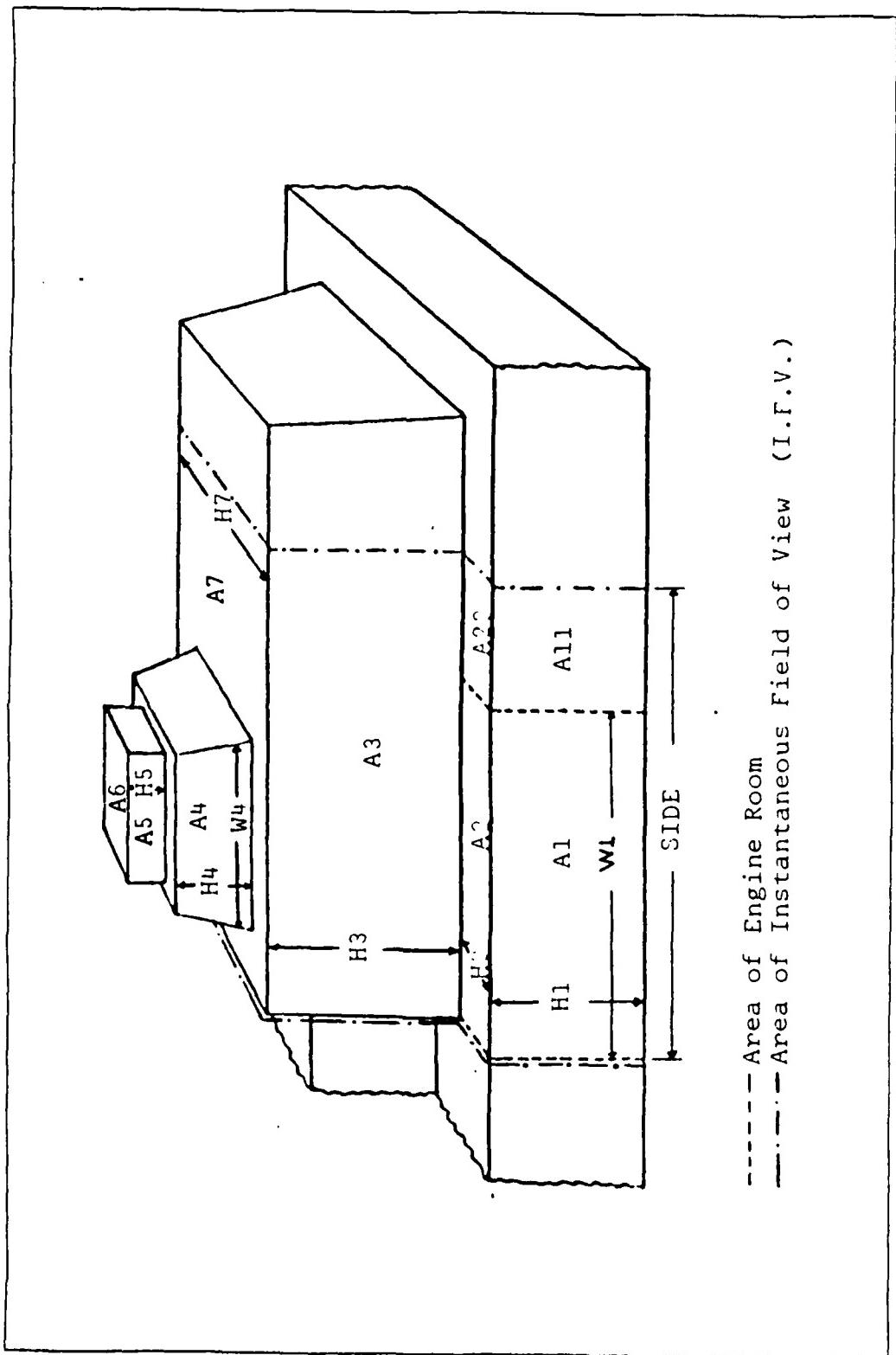


Figure 2.6 Three-Dimensional Sketch of the "Hottest" Area

TABLE 2
Final Choice for the Dimensions of I.F.V.

HEIGHT FLIGHT	SIDE (m)	PATH (m)	DELTA	PHI
50 (m)	21.88	1119.34	0.56°	2.0°
60 (m)	21.97	1123.89	0.56°	2.5°
70 (m)	22.04	1127.25	0.56°	3.0°
80 (m)	22.09	1129.84	0.56°	3.5°
90 (m)	22.13	1131.96	0.56°	4.0°
100 (m)	22.16	1133.72	0.56°	4.5°

$$A_3 = \text{SIDE} \cdot H_3$$

Vertical face

$$T_{in} = 293.15K$$

$$A_4 = W_4 \cdot H_4$$

Vertical face

$$T_{in} = 303.15K$$

At this point we must notice that surface A_5 corresponds to the stack and, since we don't have information about the insulation of this structure, we approximate its temperature with that of the surface with $T = 303.15^{\circ}K$

$$A_6 = 4.43 \cdot 10^4 \text{ cm}^2$$

Horizontal face

$$T_{in} = 425.4K$$

$$A_7 = \text{SIDE} \cdot H_7 - (\pi/4) \cdot W_4^2$$

Horizontal face

$$T_{in} = 293.15K$$

For the order of magnitude of PATH i.e. the distance between missile and ship we will approximate the thermal emission of each individual surface as point source emission. These "point sources" will be:

$(A_1 + A_{11})$, $(A_2 + A_{21})$, A_3 , A_4 , A_5 , A_6 , A_7 .

with corresponding mean altitudes from sea level:

$$H_{m1} = H_{m11} = H_1 / 2$$

$$H_{m2} = H_{m21} = H_1$$

$$H_{m3} = H_1 + H_3 / 2$$

$$H_{m4} = H_1 + H_3 + H_4 / 2$$

$$H_{m5} = H_1 + H_3 + H_4 + H_5 / 2$$

$$H_{m6} = H_1 + H_3 + H_4 + H_5$$

$$H_{m7} = H_1 + H_3$$

The data for the above calculations are:

$$W_1 = 1,620 \text{cm}$$

$$H_1 = 720 \text{cm}$$

$$H_3 = 210 \text{cm}$$

$$H_4 = 900 \text{cm}$$

$$H_5 = 1,260 \text{cm}$$

$$W_4 = 900 \text{cm}$$

$$H_6 = 360 \text{cm}$$

$$H_7 = 180 \text{cm}$$

III. ATMOSPHERIC TRANSMITTANCE USING LOWTRAN 6

A. DESCRIPTION OF LOWTRAN 6

LOWTRAN 6 is the latest issue of LOWTRAN code that has been designed to calculate the transmittance of the atmosphere over spectral bands from $0.25\mu\text{m}$ to $28.5\mu\text{m}$ along paths of various inclinations and, of course, through different atmospheric profiles and parameters.

In order to use LOWTRAN 6 program we have to fill four (4) cards (groups) of input data.

Cards 1 and 2 require the significant meteorological parameters for the atmospheric profile. Card 3 requires the geometry of the problem i.e. initial and final altitude and the path length. Card 4 is related to the wavelength range and limits.

B. DATA USED

For card 1 we used data for the Mediterranean Sea and more specifically for the Aegean Sea area and the region from Crete to North Africa.

For this area we have two different atmospheric profiles, namely Midlatitude Summer for months October through April, and Tropical for months May through September [Ref. 6].

For card 2 we used again data from [Ref. 6] for each month individually and for card 3 we used all the possible combinations between final altitude and PATH from Table 2 and for the initial altitude the mean altitudes of each "point source" of the "hottest" ship area as given in Section G of Chapter II.

At this point we have to choose one more parameter that will be needed for our problem : the wavelength range of the detector operation. We decide that this range will be, in first approximation, in the 5um window (i.e. $\lambda_1=4.20\mu\text{m}$ - $\lambda_2=5.18\mu\text{m}$) [Ref. 4:ch 5, p.89].

Lowtran 6 has the advantage of giving us, besides the average transmittance between two limits, the spectral transmittance as well. From this spectral distribution we can reject certain values of very low spectral transmittance and therefore the wavelength range can be narrower. This peculiarity will have an influence on card 4.

C. RESULTS

Running LOWTRAN 6 for all the flight altitudes of the missile and for both Midlatitude and Tropical atmospheric profiles, we observed that, for wavelengths less than $4.435\mu\text{m}$ and greater than $5.0\mu\text{m}$, the spectral transmittance generally shows values less than about 0.10. Deciding to reject these low values we choose as the final spectral range of the detector operation to be from $4.435\mu\text{m}$ to $5.0\mu\text{m}$.

In Figure 3.1 you will find the spectral transmittance in the above range for both atmospheric profiles.

Using the final wavelength range we calculated the average transmittance for all the possible combinations between atmospheric profile, missile flight altitude and individual "point sources" of the ship's tracked area. Tables 23 to 46 in Appendix C demonstrate both the data and total average transmittance for each month.

From the data of the above tables, it is obvious that the influences of both missile flight altitude (from 50m to 100m) and the relative differences between the individual "point source" altitudes (4m to 16m) are very weak on the

average transmittance. Averaging in both directions in this two dimensional table results in a unique representative value for each month. Checking these average values we observe again a relatively small difference between months belonging in the same atmospheric profile. Since a unique value of average transmittance of each atmospheric profile would be very convenient for the calculations to follow, we average over several months yielding the following final values:

Midlatitude summer : 0.4711

Tropical: 0.4259

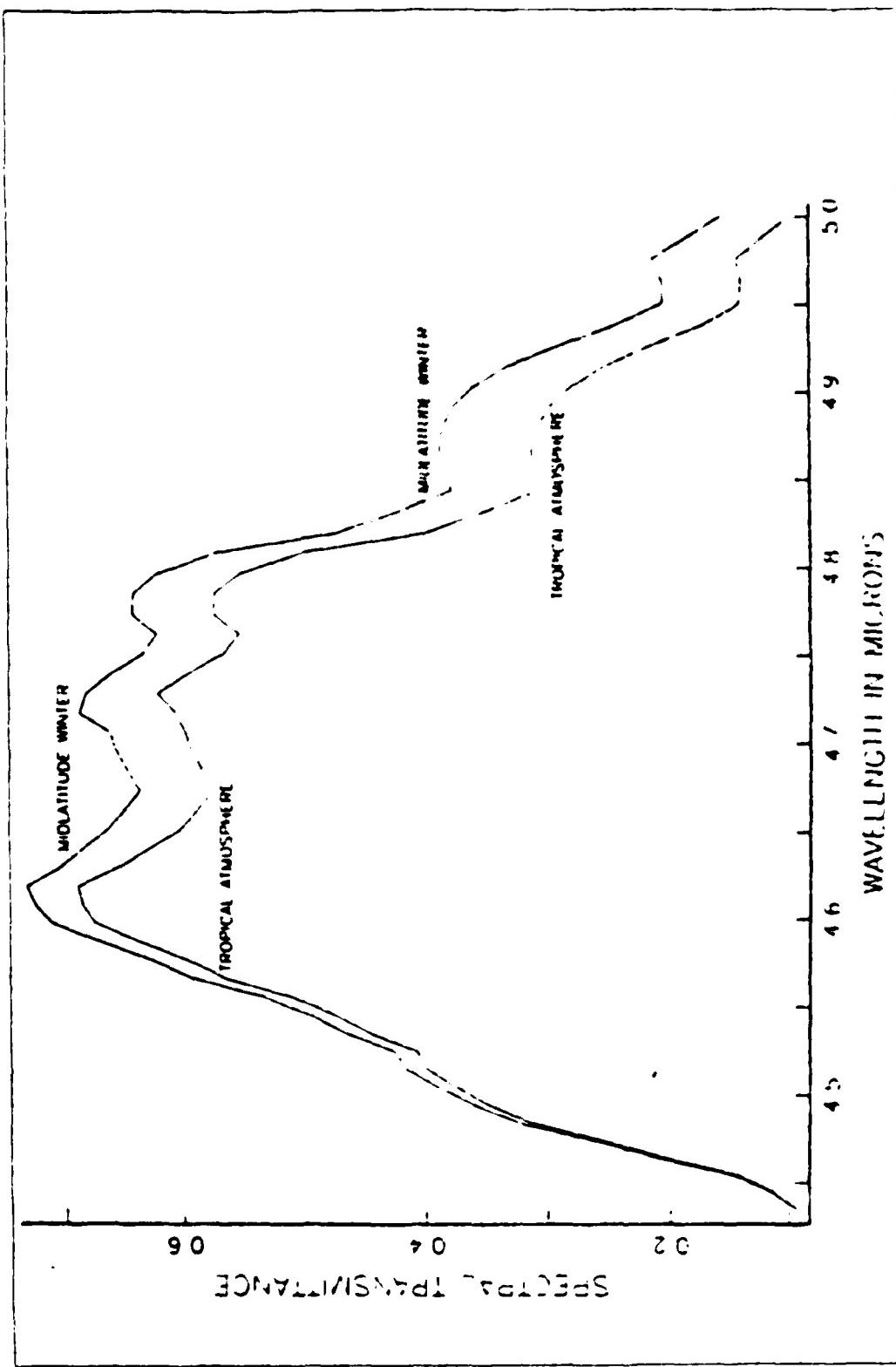


Figure 3.1 Spectral Transmittance in the Window of $4.435\mu\text{m}$ - $5.00\mu\text{m}$ Through LOWTRAN 6

IV. CALCULATION OF SHIP AND BACKGROUND RADIANCE

A. NORMAL COMPONENT OF SHIP RADIANCE DUE TO THERMAL RADIATION

The purpose of this calculation is to find, by integrating the Plank's radiation law, the radiance in the given wavelength range.

From Plank's radiation law we know that:

$$M_\lambda(T) = [2\pi c^3 h \lambda^3] \cdot [1/(e^{\frac{hc}{kT}} - 1)] \quad (\text{W/cm}^2 \text{ cm})$$

or

$$M(T) = C_1 \lambda^3 \cdot (e^{\frac{C_2}{\lambda T}} - 1)$$

where:

$$c_1 = 2\pi c^3 h = 3.7415 \cdot 10^{13} \quad (\text{W/cm}^2)$$

$$c_2 = hc/k = 1.4388 \quad (\text{cm} \cdot \text{K})$$

The above formula is in the form of:

$$M(T) = C \cdot x^m (e^x - 1)$$

Integrating this between the band limits λ_1 , λ_2 , we will find the in-band energy radiance [Ref. 4:p. 19].

Therefore:

$$L_B = \epsilon_B \pi \int_{\lambda_1}^{\lambda_2} M_\lambda(T) d\lambda = [C_1 \epsilon_B / \pi C_2] \cdot T^4 \cdot \sum_i \quad (\text{W/cm}^2 \cdot \text{Sr})$$

Where:

$$\sum_i = \left[\sum_{m=1}^{\infty} m^4 [(m \cdot x)^3 + 3(m \cdot x)^2 + 6(m \cdot x)^1 + 6m \cdot x + 6] e^{-mx} \right]_{\lambda_1}^{\lambda_2}$$

and $x = C_2 / \lambda T$

$$\sum_i = \sum_{m=1}^{\infty} m^4 [(m \cdot x)^3 + 3(m \cdot x)^2 + 6m \cdot x + 6] - \\ e^{-mx} [(m \cdot x)^3 + 3(m \cdot x)^2 + 6m \cdot x + 6]$$

In the above relations, the temperature T is taken from Tables 5, 18 in Appendix A.

B. NORMAL COMPONENT OF SHIP RADIANCE DUE TO REFLECTED SOLAR ENERGY

We have already calculated the normal component of solar irradiance E_{in} incident on each ship surface. In the energy balance calculation we considered that the solar energy absorbed by the surface is $E_{\text{in}} \cdot a_s$. Obviously the amount $(1-a_s) \cdot E_{\text{in}}$ is reflected.

Defining as $D(0-\lambda)$ the percentage of the solar constant associated with wavelengths shorter than λ we take [Ref. 4:p. 17].

$$D(0-4.435) = 99.3075454\%$$

$$D(0-5.000) = 99.511500\%$$

Therefore

$$\Delta D = D(0-5.000) - D(0-4.435) = 0.2039546\%$$

In conclusion we have that the normal reflected solar energy per m^2 from $\lambda_1=4.435\mu\text{m}$ to $\lambda_2=5.00\mu\text{m}$ is:

$$(1-a_s) \cdot \Delta D \cdot E_{\text{in}} = (1-a_s) \cdot 2.04 \cdot 10^{-3} \cdot E_{\text{in}} \quad (\text{W/m}^2)$$

In order to determine the normal component of radiance due to reflection we must divide the above value by π and in order to transform into cm^2 we must divide by 10^4 . Finally the normal component of radiance due to reflection in $(\text{W/cm}^2 \cdot \text{Sr})$ from $\lambda_1=4.435\mu\text{m}$ to $\lambda_2=5.00\mu\text{m}$ will be:

$$L_{\text{ref}} = (1-a_s) \cdot 2.04 \cdot 10^{-3} \cdot E_{\text{in}} / \pi \cdot 10^4 = (1-a_s) \cdot 4.08 \cdot 10^{-8} \cdot E_{\text{in}} \quad (\text{W/cm}^2 \cdot \text{Sr})$$

C. NORMAL COMPONENT OF STACK EXIT PLANE RADIANCE

So far we know that the stack exit plane area is $4.43 \cdot 10^4 \text{ cm}^2$ and that the corresponding uniform temperature is 425.4K.

For simplicity we consider that the greater part of the exhaust gas is CO_2 . Since CO_2 is a selective emitter we need to carry out the approximation shown in Figure 4.1 which partially accounts for atmospheric absorption and approximates the peak with a constant emissivity over a limited

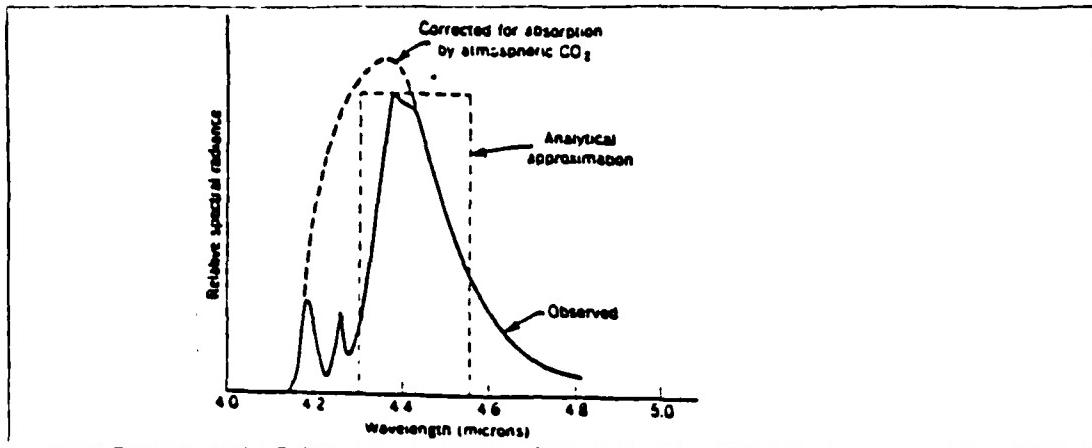


Figure 4.1 The 4.4 μ m Emission Band of CO₂ [Ref. 12]

range. Taking the band to be from $\lambda_1=4.35\mu\text{m}$ to $\lambda_2=4.55\mu\text{m}$ and the emissivity 0.5 we find for the radiance:

$$\begin{aligned} L &= \epsilon_0 \int_{4.35}^{4.55} M_\lambda d\lambda = (\epsilon \sigma T^4 / \pi) \int_{4.35}^{4.55} M_\lambda d\lambda / \int_{4.35}^{\infty} M_\lambda d\lambda = (\epsilon \sigma T^4 / \pi) \cdot (0.05) \\ &= (0.5) \cdot (5.67 \cdot 10^{-12}) \cdot (425.4)^4 \cdot (0.05) / \pi \\ &= 1.478 \cdot 10^{-3} \text{ (W/cm}^2 \cdot \text{Sr)} \end{aligned}$$

Since our band of interest overlaps with half of the 4.4 μm emission band of CO₂ the above value must be divided by 2. Therefore:

$$L = 7.39 \cdot 10^{-4} \text{ (W/cm}^2 \cdot \text{Sr)}$$

D. TOTAL SHIP RADIANT INTENSITY IN THE DIRECTION FROM SHIP TO MISSILE

As mentioned in Section G of Chapter II, we will consider the surfaces (A₁+A₁₁), (A₂+A₂₂), A₃, A₄, A₅, A₆, A₈, as "point sources". For each source we know the corresponding radiance due to thermal radiation and reflection i.e.

$$L_{T1} = L_{\Delta\lambda_1} + L_{R\Delta\lambda_1}$$

$$L_{T11} = L_{\Delta\lambda_{11}} + L_{R\Delta\lambda_{11}}$$

$$L_{T2} = L_{\Delta\lambda_2} + L_{R\Delta\lambda_2}$$

$$L_{T22} = L_{\Delta\lambda_{22}} + L_{R\Delta\lambda_{22}}$$

$$L_{T3} = L_{\Delta\lambda_3} + L_{R\Delta\lambda_3}$$

$$L_{T4} = L_{\Delta\lambda_4} + L_{R\Delta\lambda_4}$$

$$L_{T5} = L_{\Delta\lambda_5} + L_{R\Delta\lambda_5}$$

$$L_{T6} = L_{\Delta\lambda_6} + L_{R\Delta\lambda_6}$$

For L_{T6} we have the unique value given in section C of chapter III

In order to obtain the normal radiant intensity (W/Sr) of each surface we must multiply the corresponding values of each area and radiance. Doing this we take:

$$I_{1N} = L_{T1} \cdot A_1 + L_{T2} \cdot A_{11}$$

$$I_{2N} = L_{T2} \cdot A_2 + L_{T3} \cdot A_{22}$$

$$I_{3N} = L_{T3} \cdot A_3$$

$$I_{4N} = L_{T4} \cdot A_4$$

$$I_{5N} = L_{T5} \cdot A_5$$

$$I_{6N} = L_{T6} \cdot A_6$$

$$I_{7N} = L_{T7} \cdot A_7$$

The important magnitude for us is the radiant intensity in the direction of source - missile.

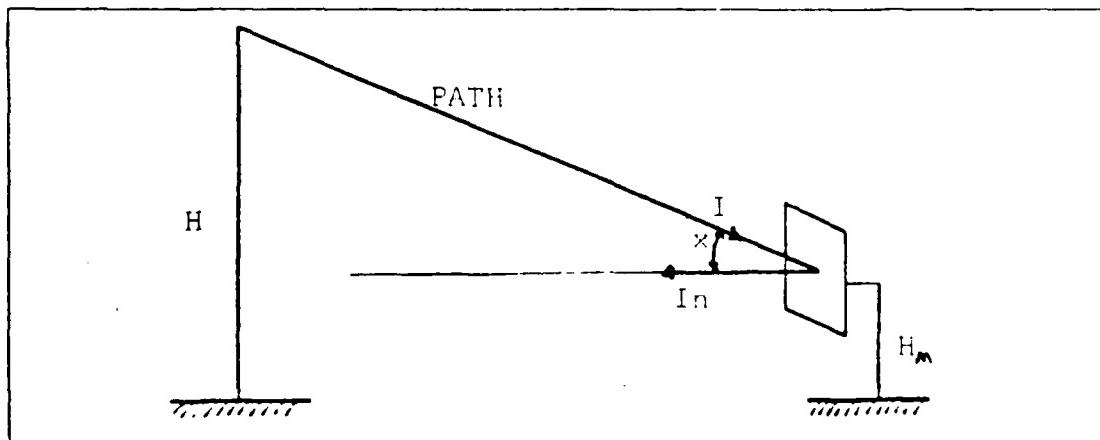


Figure 4.2 Relative Position Between Surface- Missile

In Figure 4.2 you will find a sketch representing the relative position between the surface and the missile. Based on that we conclude this the radiant intensity will be:

$$I = I_N \cdot \cos x \text{ for a vertical face}$$

$I = I_N \cdot \sin x$ for a horizontal face
where $x = \sin^{-1}(H - H_N / \text{PATH})$

The resulting relations will be:

$$I_1 = I_{N1} \cdot \cos x$$

$$I_2 = I_{N2} \cdot \sin x$$

$$I_3 = I_{N3} \cdot \cos x$$

$$I_4 = I_{N4} \cdot \cos x$$

$$I_5 = I_{N5} \cdot \cos x$$

$$I_6 = I_{N6} \cdot \sin x$$

$$I_7 = I_{N7} \cdot \sin x$$

E. BACKGROUND RADIANT INTENSITY

In the ideal case in which the detector tracks perfectly on the ships "hottest" area and the square I.F.V. will fit perfectly on that the detector will still see some areas outside of the ship. This is due to the fact that the upper construction of the ship has not a perfect rectangular shape. Therefore the detector will receive signal from an area of the sea that radiates behind the ship. Since this signal constitutes noise for the detector, it has important meaning.

In Figure 4.3 you will find the geometry from which we will determine the background radiant intensity. We must notice that the sketch in Figure 4.3 has been magnified for convenience in the ship region. In other words angle ($\theta + \delta$) is very close to 90° and the side of I.F.V. AE is ASE is practically equal to the ship dimension ABC.

AB represents the height of surface A₂ [Figure 2.6] from the sea level or $AB = H_1 + H_2$

BC then represents the path through which the detector will see the sea area FG past the stack.

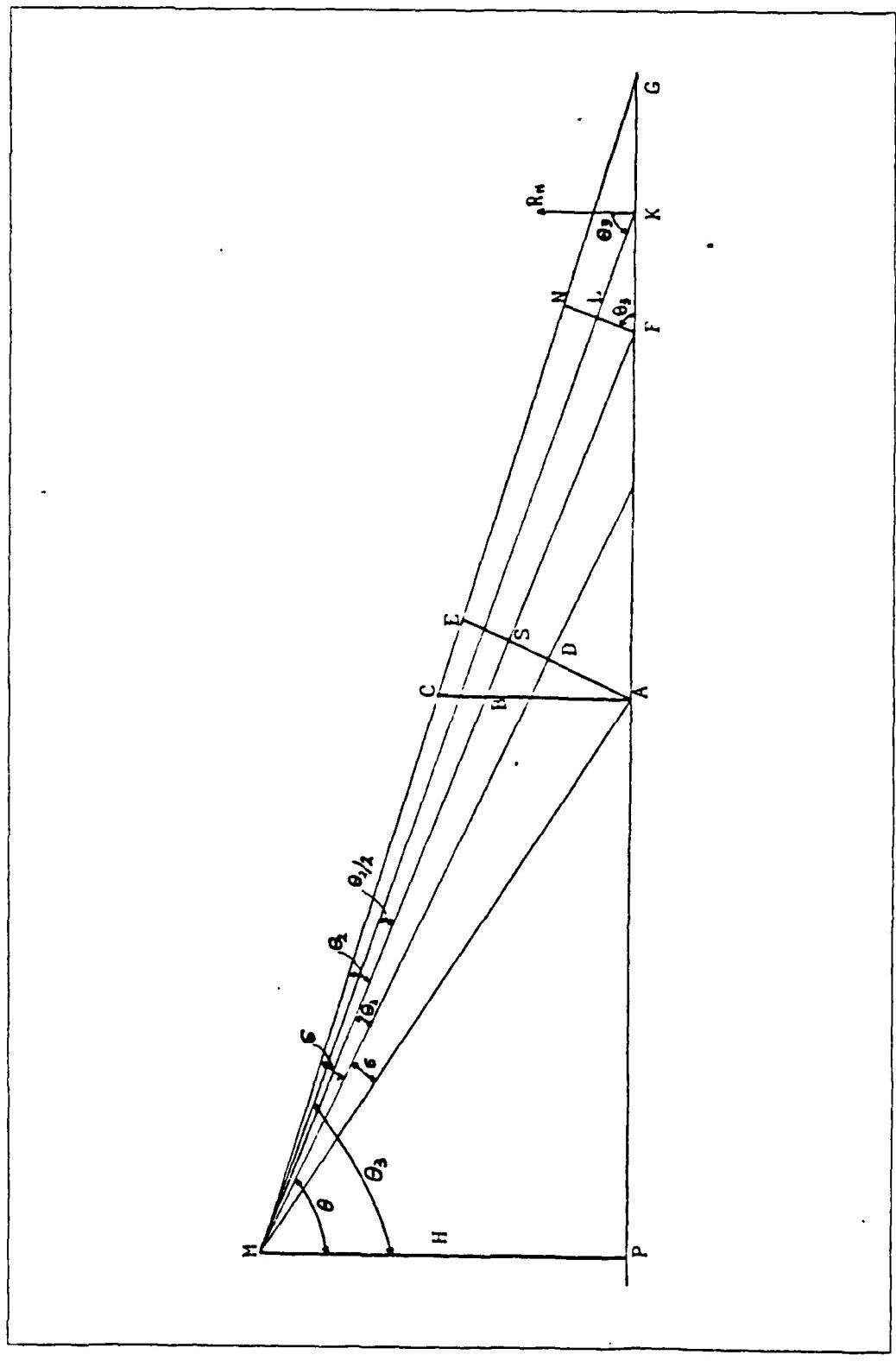


Figure 4.3 Geometry for Determination of Background Radiant Intensity

The width of this area is ($SIDE - W_1$) which represents the free horizontal dimension over the surface A_2 .

From the geometry of Figure 4.3 we have:

$$DS = AS - SIDE$$

$$\theta_1 = \tan^{-1} (DS/PATH)$$

$$\theta_2 = \delta - \theta_1$$

$$MA = PATH / \cos \delta$$

$$\theta = \cos^{-1} (H/MA)$$

$$PF = H \cdot \tan(\theta + \delta + \theta_1)$$

$$PG = H \cdot \tan(\theta + 2\delta)$$

$$FG = PG - PF$$

$$\theta_3 = \theta + \delta + \theta_1 + \theta_2 / 2$$

$$MK = H \cdot \tan \theta_3$$

$$AREA = FG \cdot [SIDE - W_1] \cdot \cos \theta_3$$

Table 3 demonstrates the result of the above calculations for six different flight altitudes.

TABLE 3
Geometrical Parameters for Observed Sea Area

H (m)	FG (m)	θ (deg)	MK (m)	AREA (m ²)
50	333.67	88.415	1806.98	144.90
60	230.54	87.915	1648.07	131.68
70	180.09	87.410	1547.47	127.76
80	144.50	86.91	1481.9-	122.29
90	116.07	86.4	1430.50	114.42
100	99.42	85.9	1395.07	111.60

For the sea water we know that the average temperature in the geographical area of our interest is $13^\circ\text{C}=286.15\text{K}$ for Midlatitude Summer and $18^\circ\text{C}=291.15\text{K}$ for Tropical atmosphere [Ref. 6].

For the sea water emissivity we chose the value of 0.96 [Ref. 10:p.07].

For the calculation of the normal component of the sea radiance ($\text{W/cm}^2 \cdot \text{Sr}$) in the range $\lambda_1=4.435\mu\text{m}$ to $\lambda_2=5.00\mu\text{m}$ we will use again the formulas from section A of chapter IV. This value must be multiplied by the water emissivity =0.96. For convenience we consider the observed area of sea as a point source.

The normal component of the radiant intensity of this point source it will be:

$$I_{wN}=L_{N\delta}\cdot \text{AREA}$$

And in the direction leading to the missile will be:

$$I_w=I_{wN}\cdot \cos\theta$$

V. DETECTOR MODELING AND CALCULATION OF S/N RATIO

A. DETECTOR SYNTHESIS

We already studied the geometry of the detection problem which shows us what the detector is required to do, and of course from this side we are interested in the most realistic performance of the detector. Since we do not have available experimental data that could show us the performance of the detector element we have to choose this according to the literature.

From Figure 5.1 [Ref. 10] the possible choices for the detector include photovoltaic or photocontactive Indium Antimonide and Lead Selenide. The detailed data sheets for these detectors [Ref. 12:pp.365-368] show the relative advantages and disadvantages of each. When they are compared on the basis of D^* , photovoltaic Indium Antimonide is clearly the best detector. A possible difficulty with this conclusion arises when the responsivities are compared and it is noted that value of Lead Selenide is 10 times higher than it is for the other two types of detector. In an ideal operating environment, that is, in which there are no extraneous sources of noise, the designer is interested in the signal - to - noise ratio of the detector, as described by D^* and not the signal level, as described by responsivity. In advance we know that during the experimental trials photovoltaic Indium Antimonide had the better performance. Therefore we are led to the decision to use one element of photovoltaic Indium Antimonide, and we can see the detailed data for this material in Figure 5.2 [Ref. 12:p.368].

This one element will be placed in the middle of an assumed two mirror system as in Figure 5.3, with

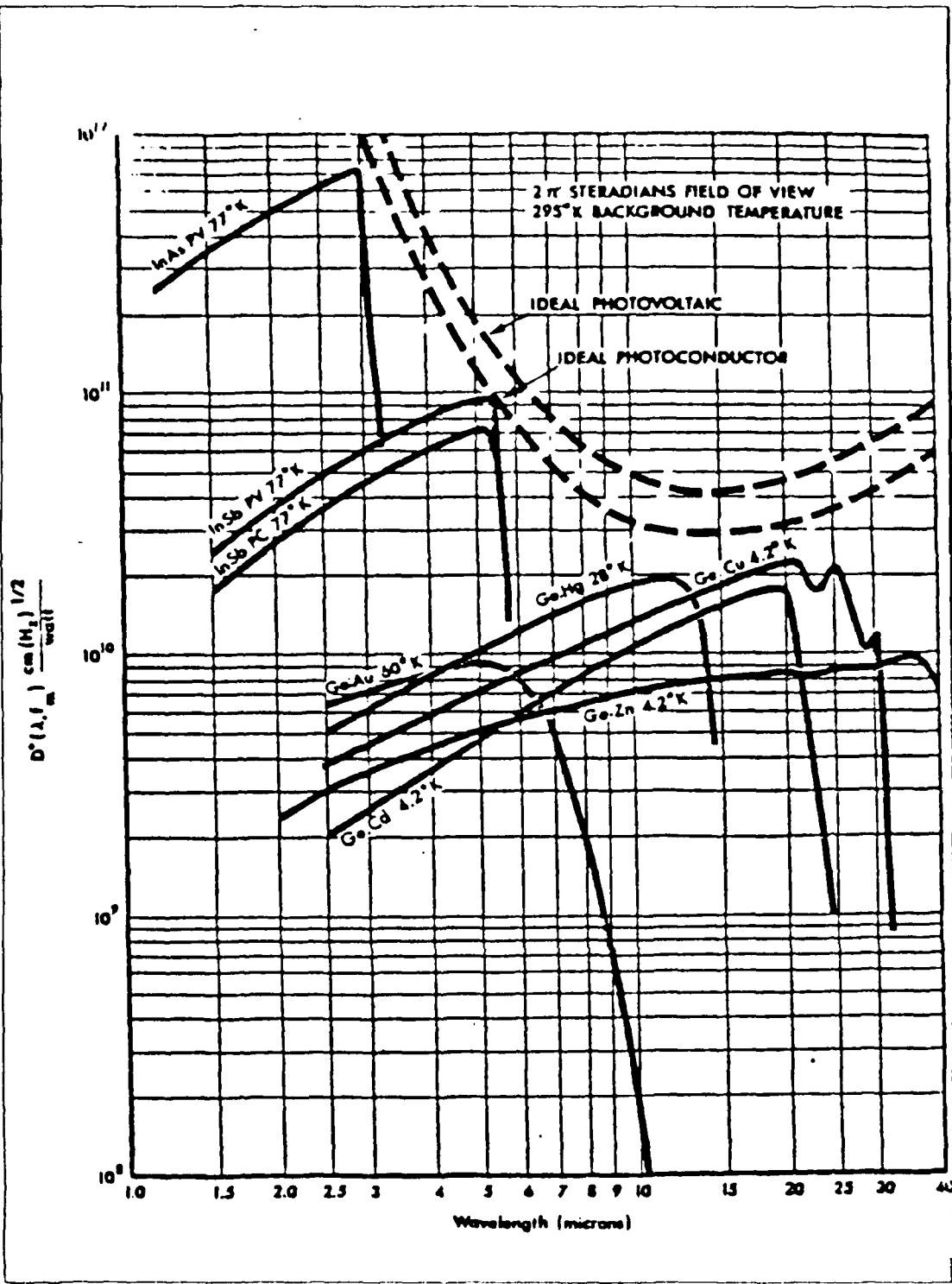


Figure 5.1 The D^* of the Detectors in Each Atmospheric Window at the Most Common Operation Temperature [Ref. 10]

Indium Antimonide (InSb)

Type: Photovoltaic.
 Element sizes: From 0.1×0.1 to 10×10 mm.
 available: Rectangular elements from
 0.1 mm wide. Circular elements
 with diameters from 0.1 to 10 mm.
 Time constant: Less than 1 μ sec.
 Dynamic impedance: $2-5 \times 10^4$ ohm.
 Responsivity: 2×10^4 VW^{-1} .
 $D^*(\text{peak})$
 $D^*(500^\circ\text{K})$: 6.2

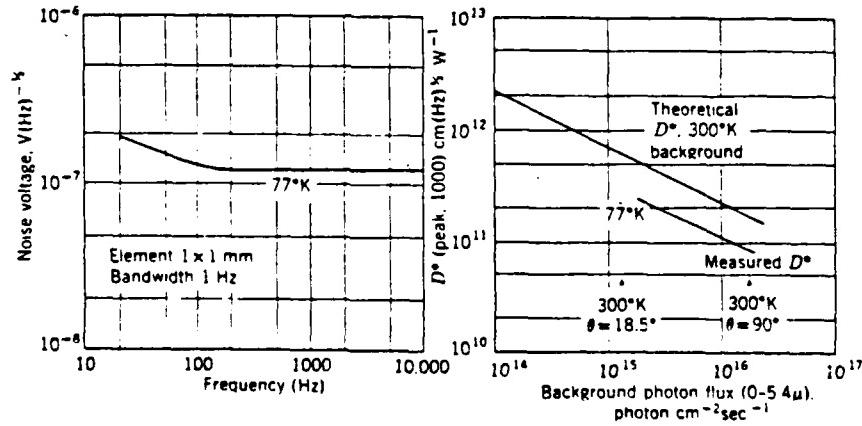
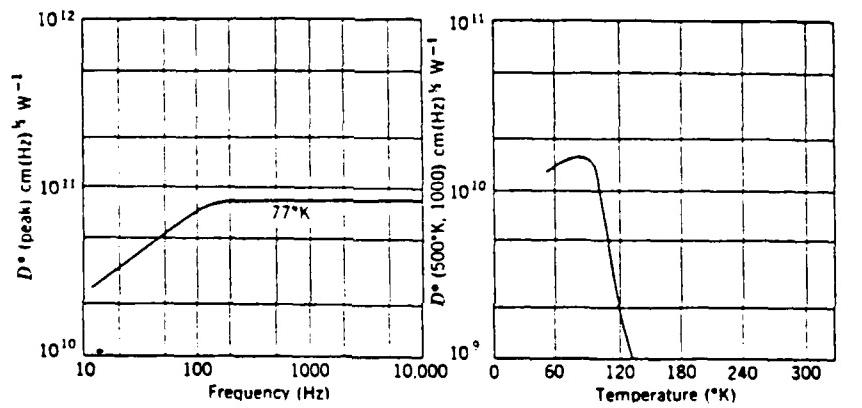
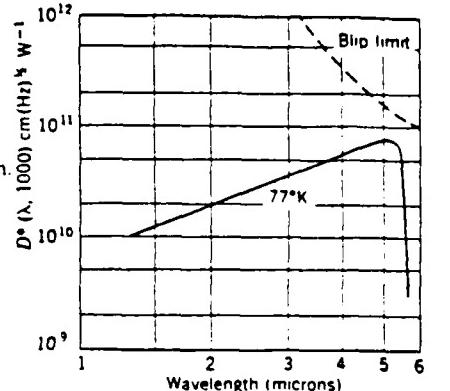


Figure 5.2 Data Sheet for Photovoltaic Indium Antimonide Detector [Ref. 12]

reflectivity 0.9 each mirror and total of two $r = 0.9 \cdot 0.9 = 0.81$. The detector will be considered to be used with a two-mirror system designed so that the detector element will subtend area equal to the I.F.V. at a distance of PATH. Thus we have to determine the size of the sensitive element (detector).

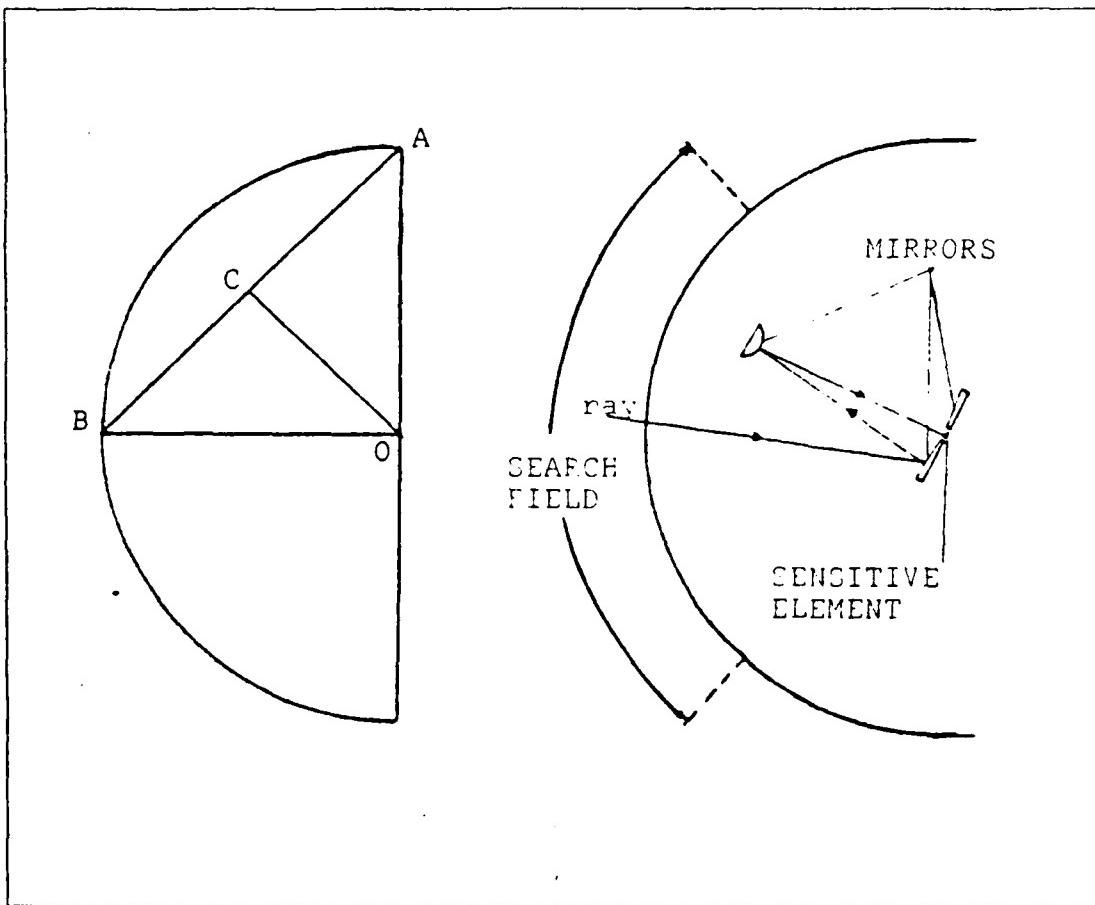


Figure 5.3 System of Two Mirrors in the Receiver

We know that the area of the detector A is given by:

$$A_d = w \cdot f^2$$

where

w=Instantaneous Field of View in (Sr)

$$w = (\text{SIDE} \cdot \text{SIDE}) / (\text{PATH})^2 = (2200 \cdot 2200) / (\text{PATH})^2 \text{ (Sr)}$$

f =equivelant focal length

and for PATH we have to use the values from Table 2 which is function of the flight altitude.

For the our purpose of work a very good and accurate approach to calculate the equivalent focal length of the system of two mirrors is the following.

First we have to choose the diameter of the entrance aperture $D_0 = 12\text{cm}$. This dimension is good enough to have an aerodynamic shape in the nose of the missile. Thus from the Figure 5.3 we have:

$$(AB)^2 = (AO)^2 + (OB)^2 = 72 \text{ and } AB = 8.4\text{cm}$$

$$(AO)^2 = (AC)^2 + (CO)^2 \text{ or}$$

$$(CO)^2 = (AO)^2 - (AC)^2 = (AO)^2 - (AB/2)^2 = \\ = (CO)^2 = 6^2 - (8.49/2)^2 = 17.98\text{cm}^2$$

and $CO = 4.24\text{cm}$

We can define the distance CO as the equivalent focal length $f = 4.24\text{cm}$

And

$$A_d = w \cdot f^2 = (2200)^2 \cdot (4.24)^2 / (\text{PATH})^2$$

Substituting the values of PATH for each flight altitude gives A and the square root of this gives us the side of the detector element. Table 4 demonstrates the dimensions of the detector element shows that with reasonable detector size the I.F.V. at about 1125 meters slant path range will include the hot region of the ship regardless of the flight altitude.

B. SIGNAL VOLTAGE

The general formula that gives the signal voltage is:

$$V_s = [A_0 \cdot r_0 \cdot R / (\text{PATH})^2] \cdot \left[\int_{\lambda_1}^{\lambda_2} I_\lambda T_\lambda(\lambda) d\lambda \right]$$

Where

A_0 : Area of entrance aperture

$$A_0 = \pi D^2 / 4 = 113.1\text{cm}^2$$

r_0 : Optics reflectance

TABLE 4
Dimensions of the Detector Element

FLIGHT ALTITUDE (m)	PATH (m)	DIMENSION OF DETECTOR ELEMENT (cm)
50	1,120	0.83 x 0.83
60	1,124	0.83 x 0.83
70	1,127	0.83 x 0.83
80	1,130	0.83 x 0.83
90	1,131	0.83 x 0.83
100	1,134	0.83 x 0.83

R : responsivity = $2 \cdot 10^4$ (V/W)

I_λ : Spectral radiant intensity of the source

T_a : Spectral atmospheric transmittance

The above relation can be further simplified using instead of the spectral transmittance an average one as we determined in Section C of Chapter III i.e. 0.4711 for Midlatitude summer and 0.4259 for Tropical atmosphere.

Taking $T_a(\lambda)$ outside of the integral then the quantity $\int_{\lambda_1}^{\lambda_2} I_\lambda d\lambda$ becomes simply the inband radiant intensity of the source which has been calculated in section D of chapter III. Hence the signal voltage formula becomes:

$$Vs = A_s r R T_a I_{\Delta\lambda} / (\text{PATH})^2$$

In order to determine the signal voltage due to all the available point sources inside the I.F.V. we use the superposition principle:

$$Vs = [A_s r R T_a / (\text{PATH})^2] \cdot [I_1 + I_2 + I_3 + I_4 + I_5 + I_6 + I_7]$$

C. NOISE VOLTAGE

The noise voltage is due to two main factors, namely the background noise that is created from the background radiant intensity and the noise from the optical system.

In Section E of Chapter III we determined the radiant intensity of the sea region that the detector observes and we named I_w (W/Sr). The noise voltage of this reason will be:

$$V_{ns} = A_s \cdot r_s \cdot R \cdot L_s \cdot I_w / (MK)^2$$

where MK is determined in Figure 4.3 and indicates the average path between the observed sea surface and the detector.

The noise signal due to the optical system is given by: [Ref. 12].

$$V_{no} = R \cdot (A_s \Delta f)^{1/2} / D^*$$

where

D^* : Detectivity with average value $9 \cdot 10^{11}$ [cm(Hz) $^{1/2}$ /W] in the wavelength range $\lambda=4.435\mu\text{m}$ to $\lambda=5.00\mu\text{m}$

A_s : Area of the detector as determined in previous section of the present chapter.

R : Responsivity

Δf : equivalent noise bandwidth

The equivalent noise bandwidth is given by

$$\Delta f = 1.2 T_d$$

where

T_d : dwell time (time required for the image of the target to pass across the detector).

and $L_s = w \cdot c \cdot \dot{\Omega}$

where

c: number of detector elements (1 for our case)

$\dot{\Omega}$: Time rate of search.

and

$$\dot{\Omega} = \Omega / T_f$$

where

Ω : size of search field in (Sr)

t_f : frame time (time required to scan the entire search field). For our case we chose 10msec.

Therefore

$$\Delta f = 1/2t_f = \dot{\Omega}/2 \cdot w \cdot c = \Omega/2 \cdot w \cdot c \cdot t_f$$

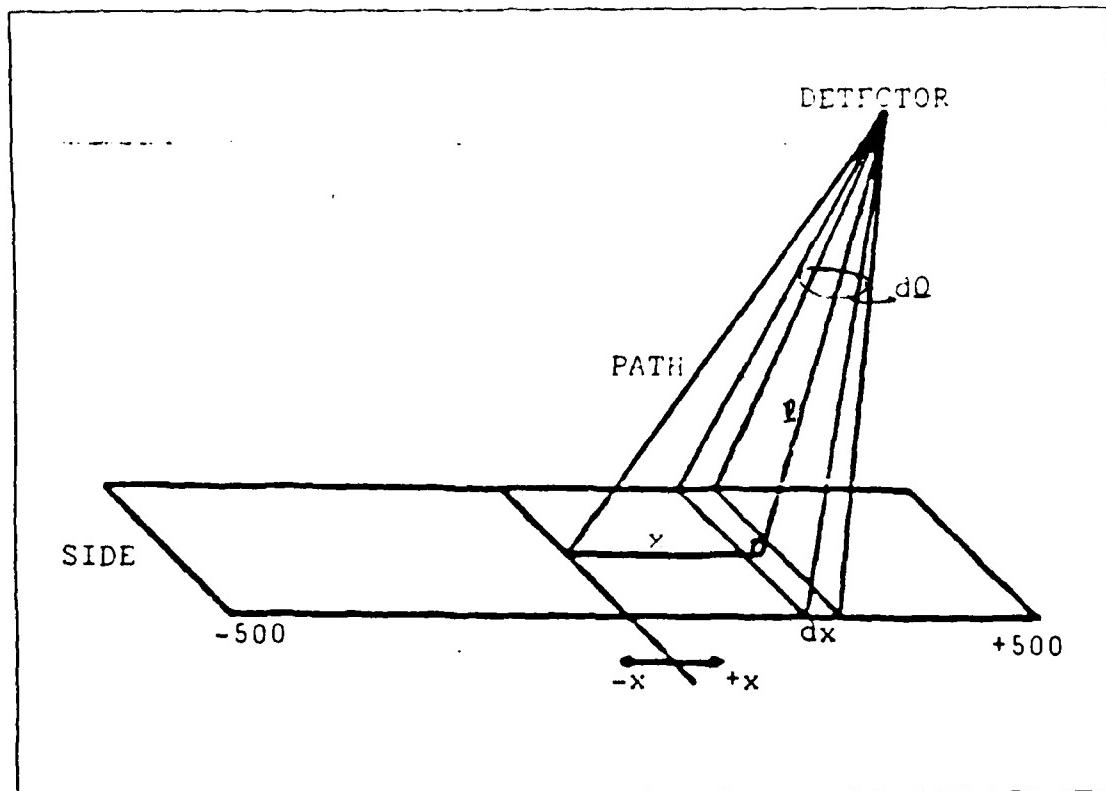


Figure 5.4 Total Detector Search Field

In Figure 5.4 you will find the geometrical representation of the total search field which will have total length of 1000m and is scanned by the detector in =10msec.

The total solid angle that the search field subtends at the detector is:

$$\Omega = dQ$$

$$dQ = dx \cdot SIDE / l^2$$

$$l^2 = (\text{PATH})^2 - x^2$$

$$dQ = [SIDE / (PATH)^2 - x^2] \cdot dx$$

$$Q = (SIDE) \int dx / (PATH)^2 - x^2$$

Since $(PATH)^2 > 500^2$ always, from the tables we get:

$$\begin{aligned} Q &= (SIDE)/2(PATH)[\ln((PATH+x)/(PATH-x))] \\ &= (SIDE)/2(PATH)[\ln((PATH+500)/(PATH-500)) \\ &\quad - \ln((PATH-500)/(PATH+500))] \\ &= ((SIDE)/2(PATH)) \cdot \ln[(PATH+500)/(PATH-500)]^2 \end{aligned}$$

$$Q = ((SIDE)/(PATH)) \cdot \ln[(PATH+500)/(PATH-500)]$$

Therefore

$$\Delta f = (1/2(SIDE) \cdot c \cdot T_f) \cdot \ln[(PATH+500)/(PATH-500)]$$

The total noise voltage is

$$V_n = V_{n_0} + V_{n_e}$$

And the signal to noise ratio is:

$$V_s/V_n = V_s / (V_{n_0} + V_{n_e})$$

In figures 5.5 to 5.22 we can see the summary of the results of the above calculation for all the possible combinations of the relative parameters.

In Appendix D we can see the influence of sky condition to S/N ratio for the two extremes flight altitude i.e. 50 (m) - 100 (m) and for all the combinations between atmospheric profile and paint emissivity.

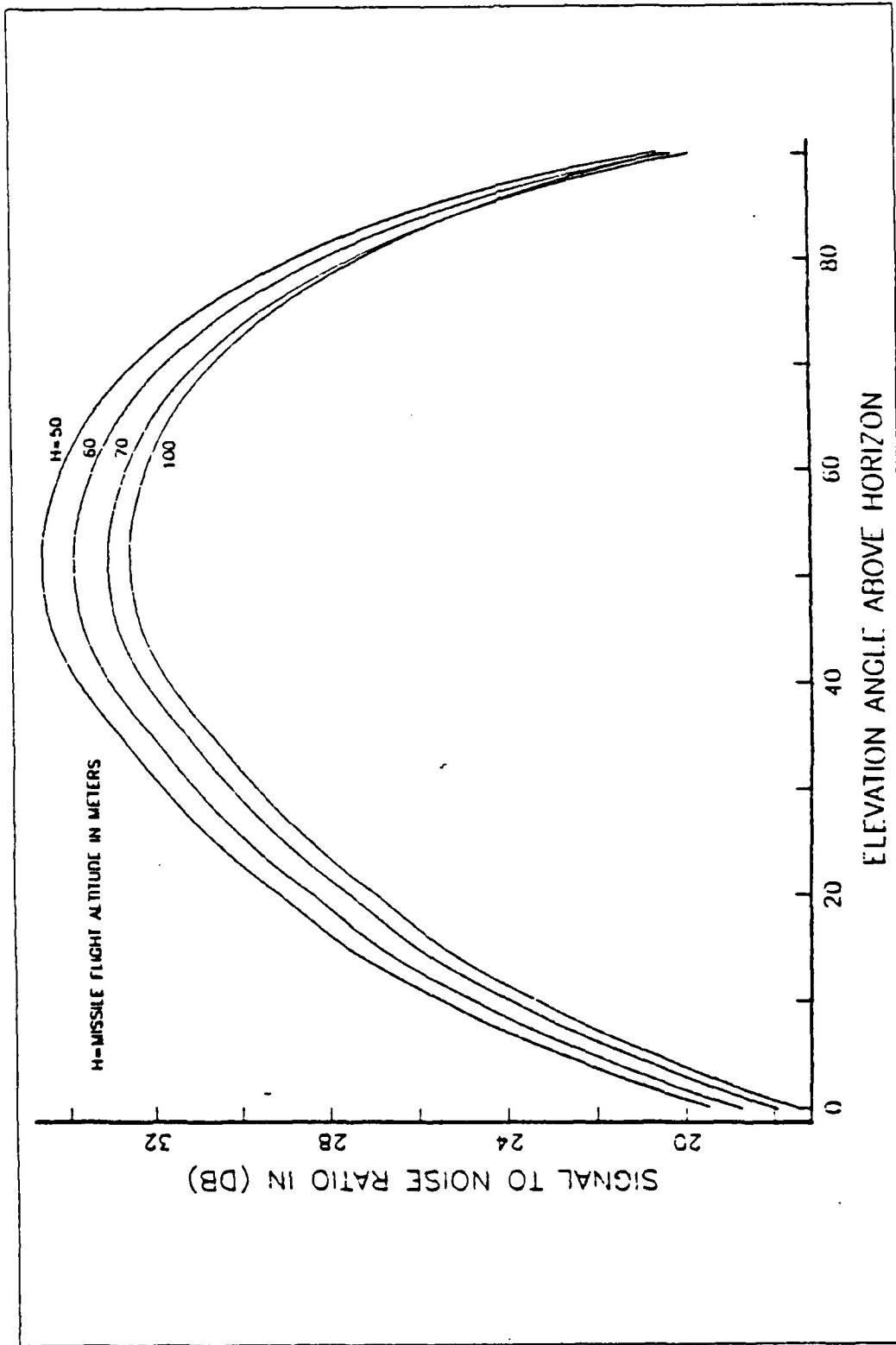


Figure 5.5 Influence of Missile Flight Altitude for Unobscured Sun, $\epsilon_0=0.80$, Midlatitude Summer

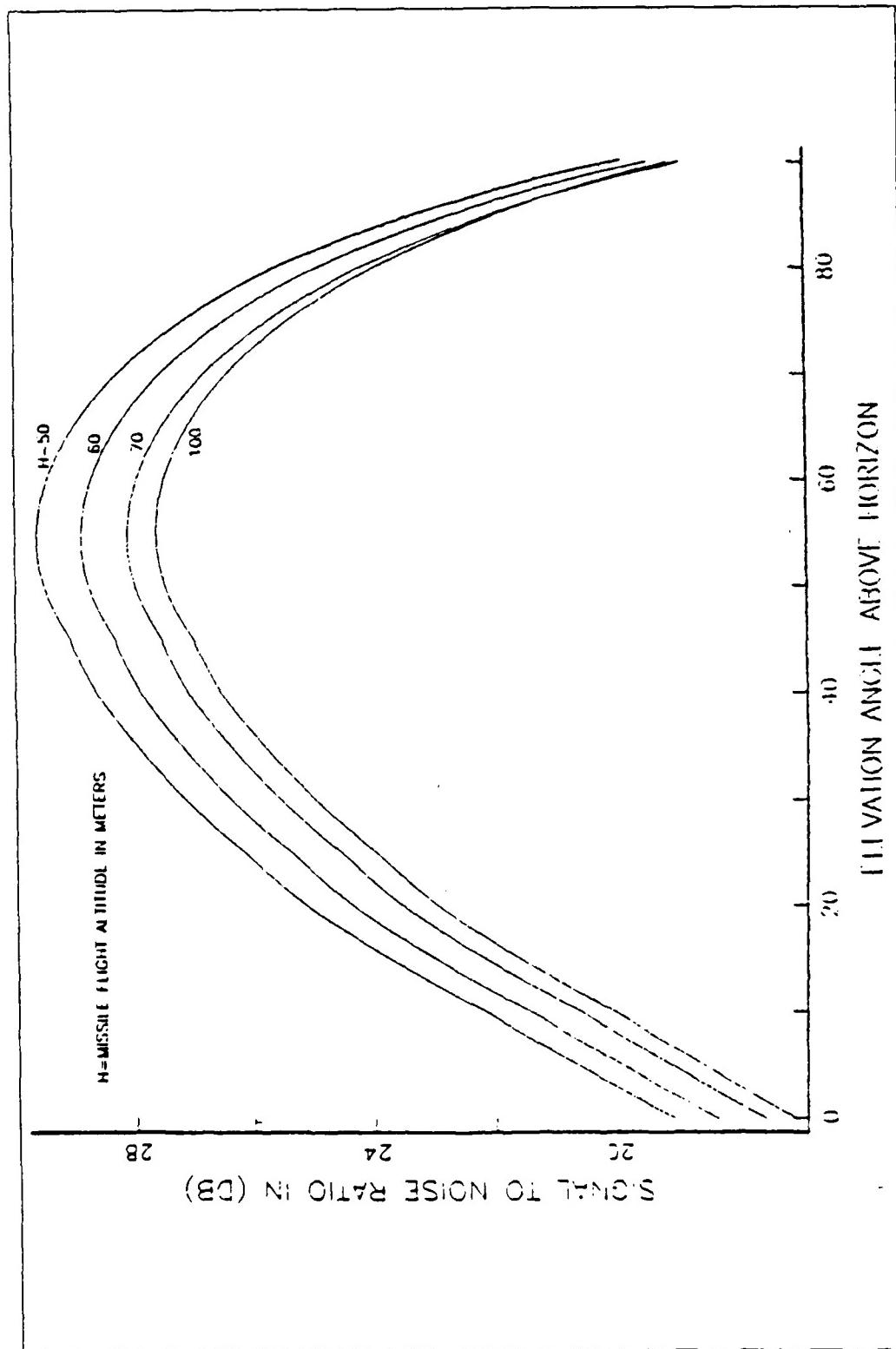


Figure 5.6 Influence of Missile Flight Altitude for Sun with Light Clouds, $\epsilon_0 = 0.80$, Midlatitude Summer

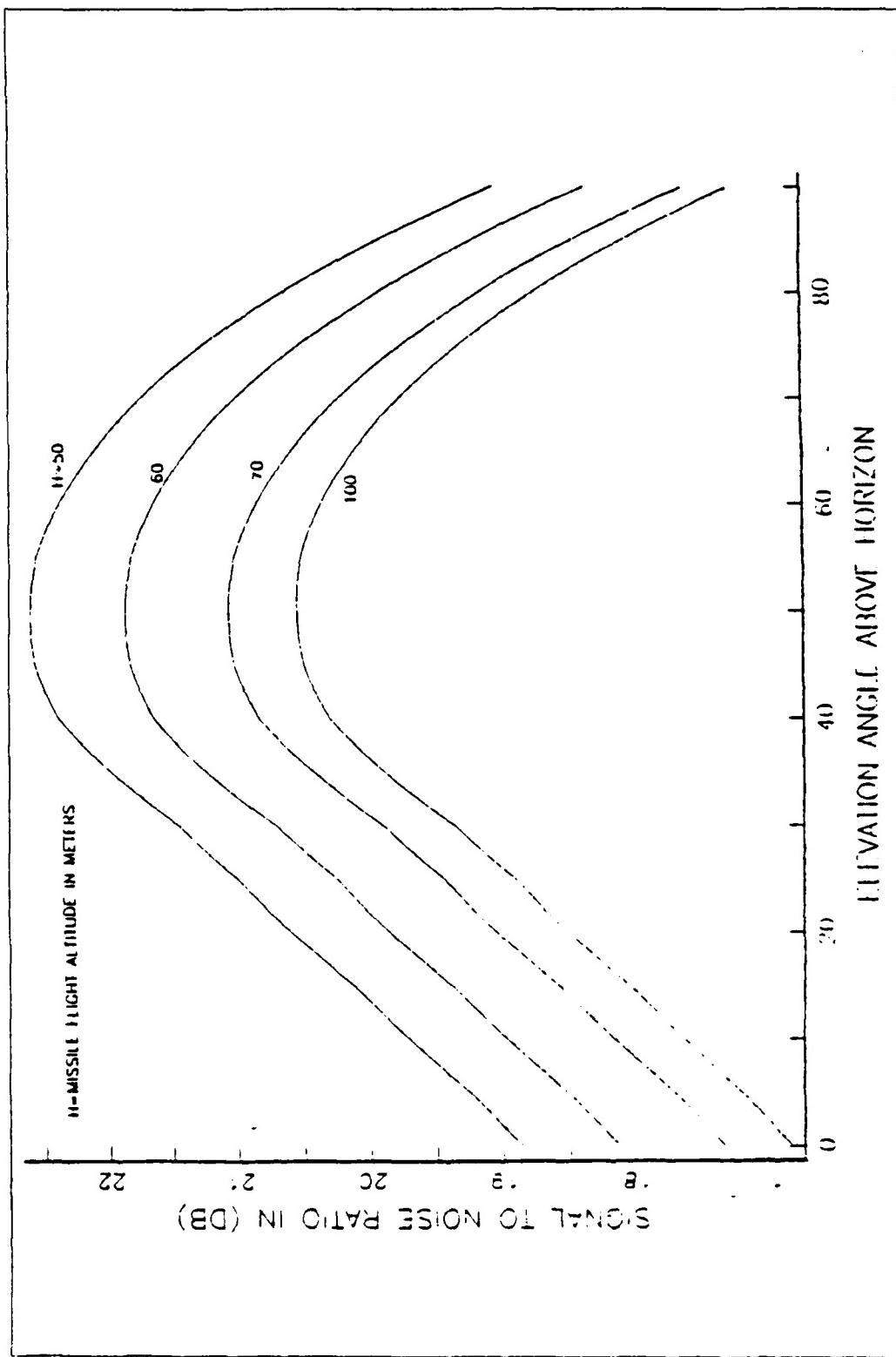


Figure 5.7 Influence of Missile Flight Altitude for Sun with Heavy Storm Clouds, $\epsilon_0=0.80$, Midlatitude Summer

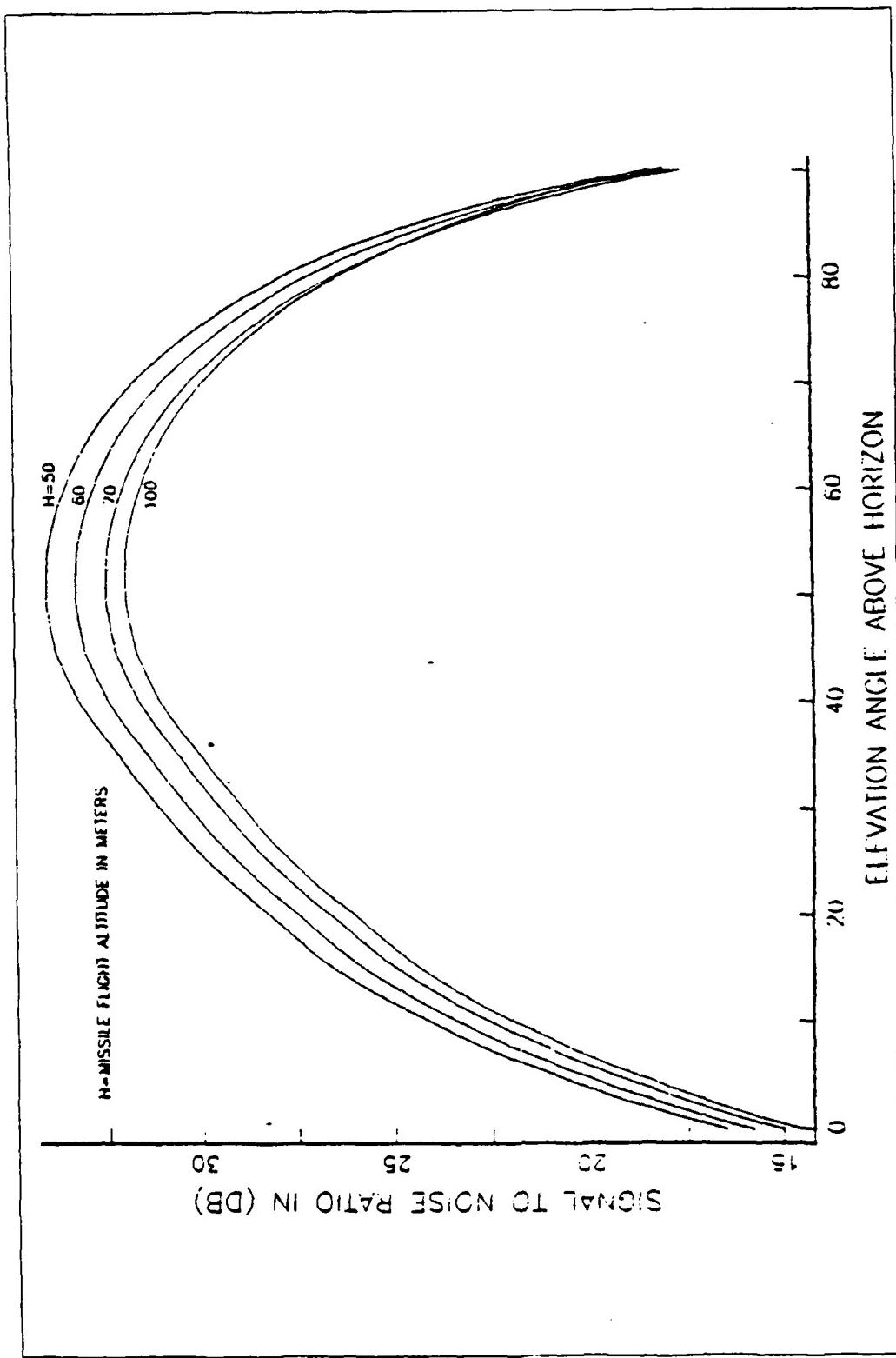


Figure 5.8 Influence of Missile Flight Altitude for Unobscured Sun, $\epsilon_0 = 0.55$, Midlatitude Summer

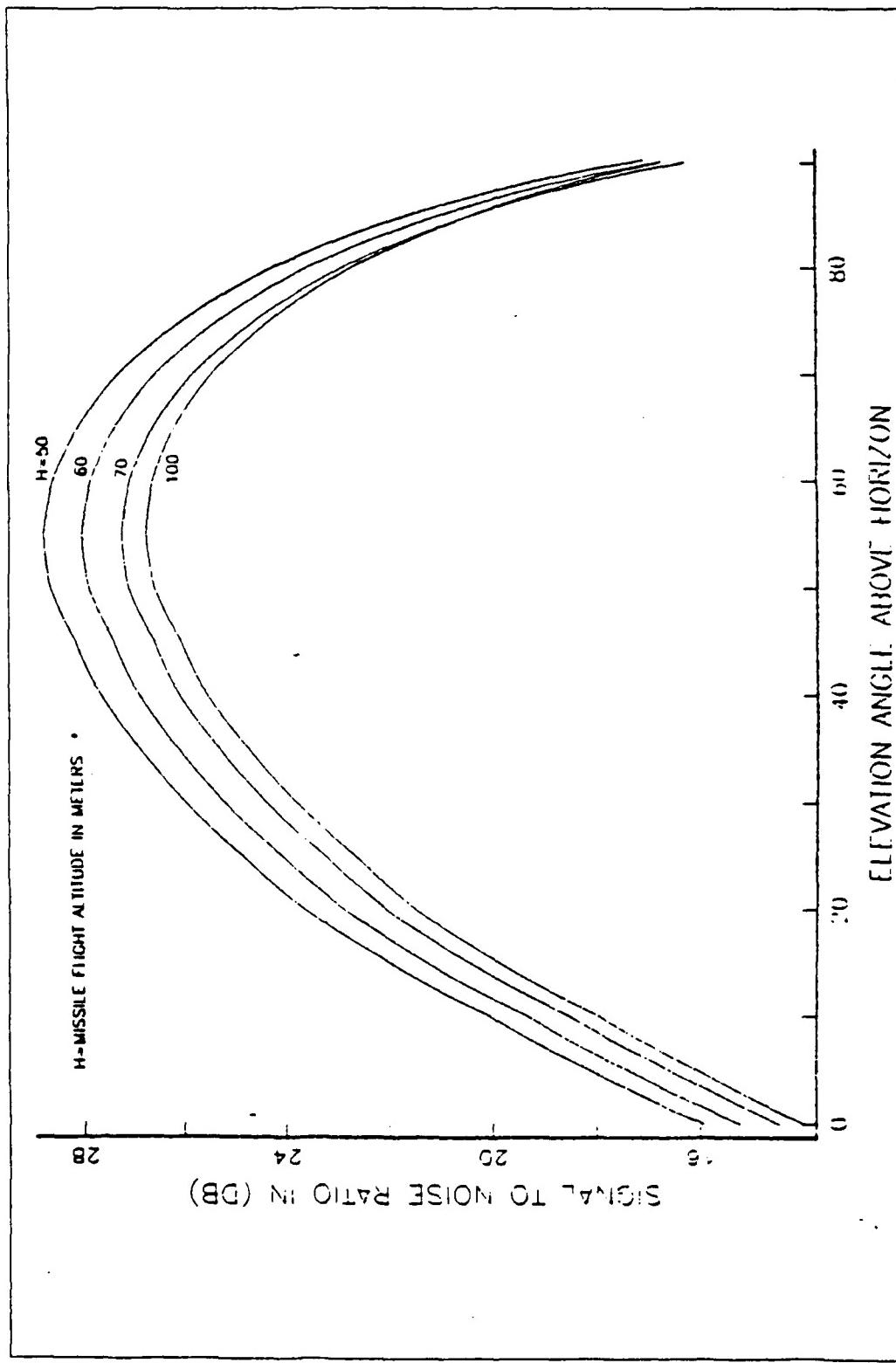


Figure 5.9 Influence of Missile Flight Altitude for Sun with Light Clouds, $\epsilon_0 = 0.55$, Midlatitude Summer

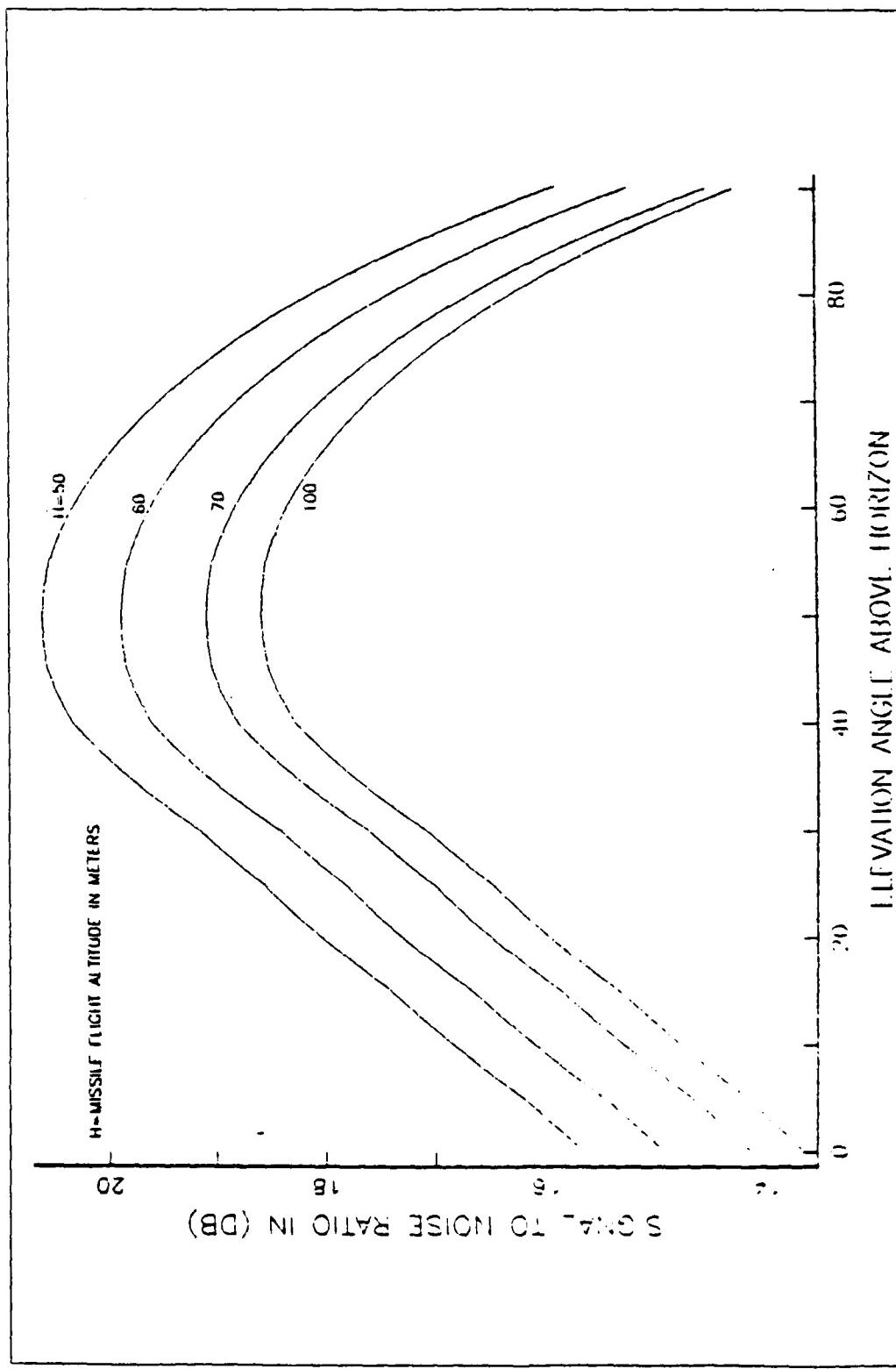


Figure 5.10 Influence of Missile Flight Altitude for Sun with Heavy Storm Clouds, $\epsilon_0 = 0.55$, Midlatitude Summer

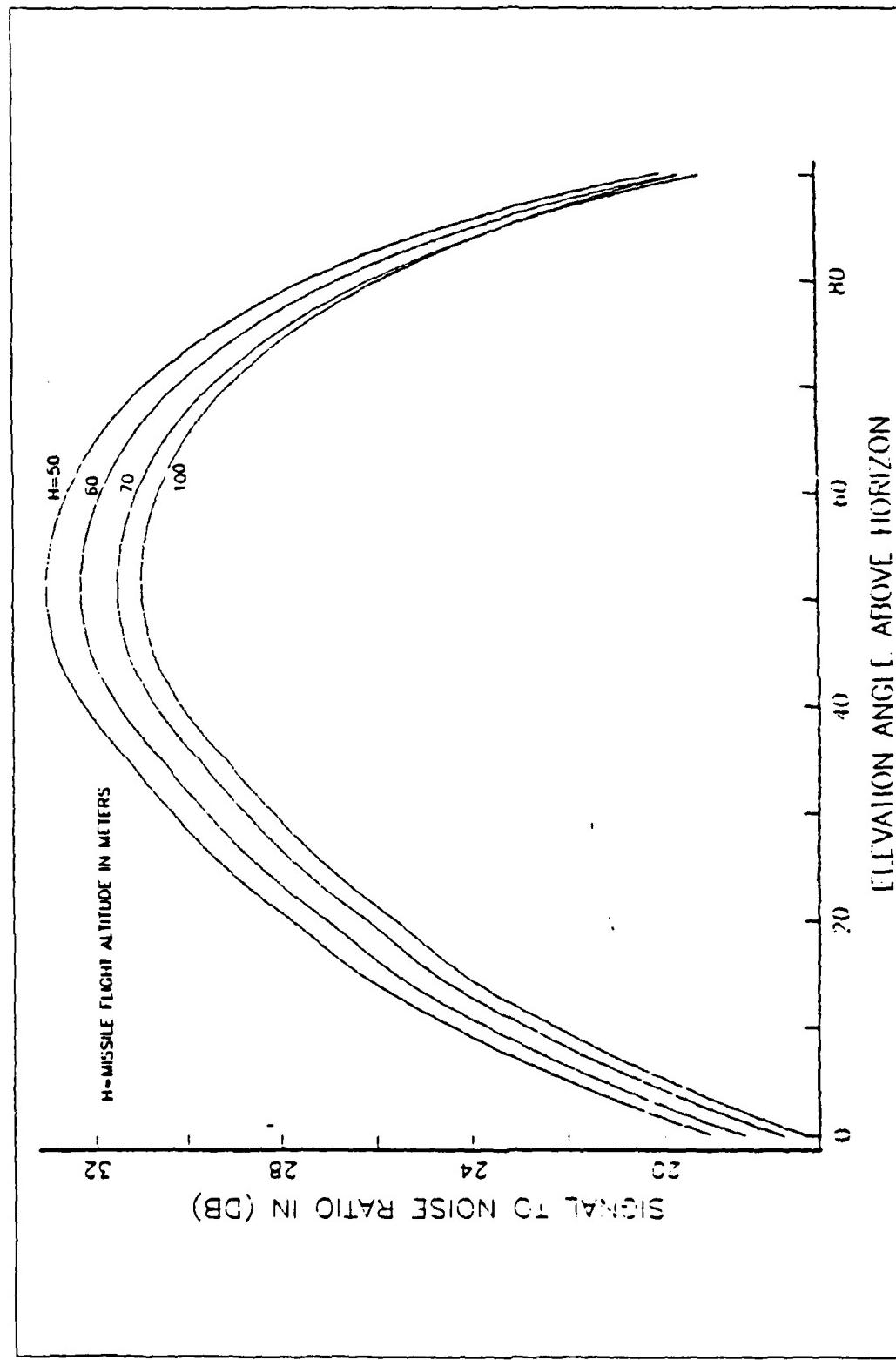


Figure 5.11 Influence of Missile Flight Altitude for Unobscured Sun,
 $\epsilon_0 = 0.80$, Tropical Atmosphere

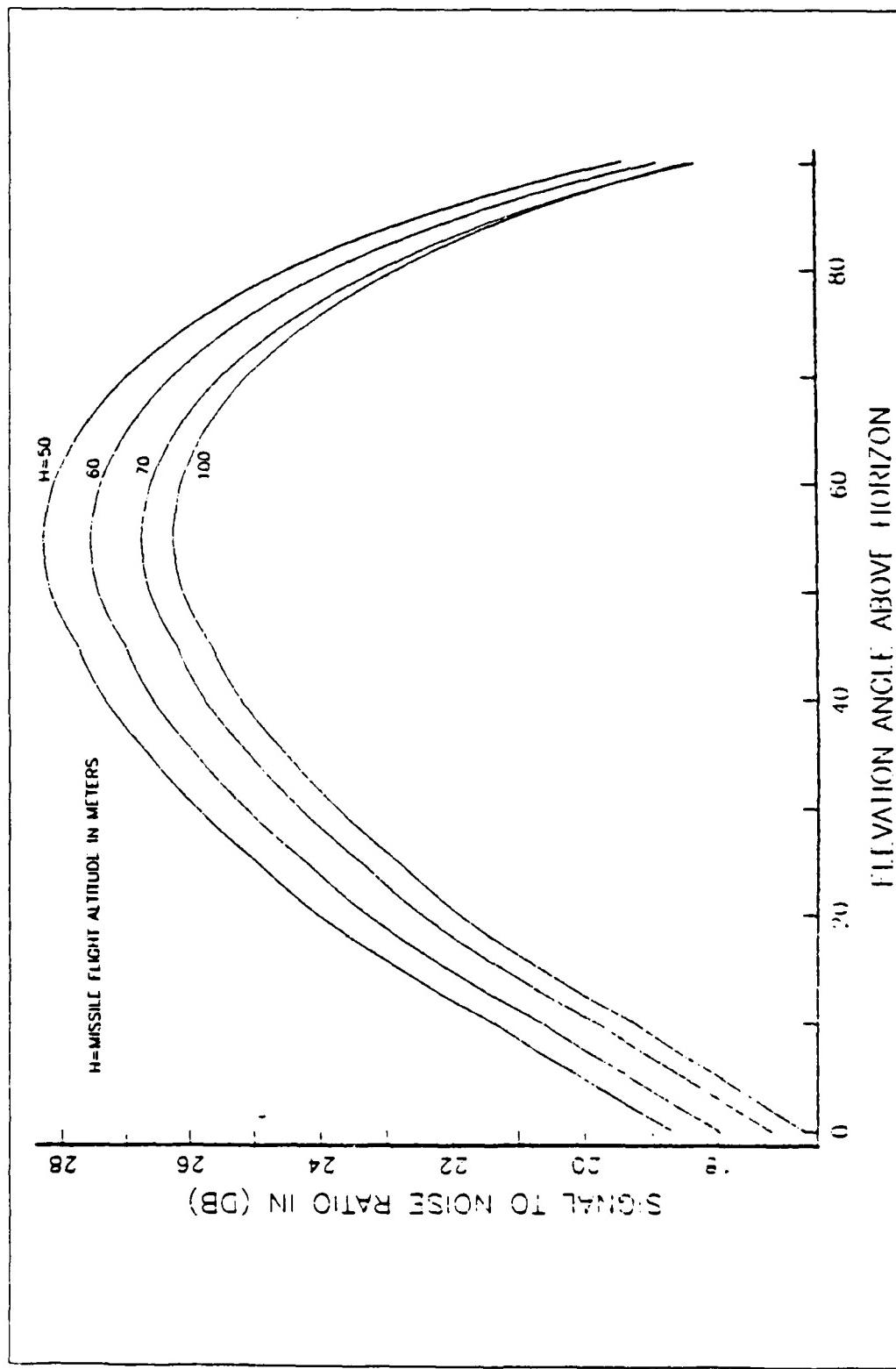


Figure 5.12 Influence of Missile Flight Altitude for Sun with Light Clouds, $\epsilon_s = 0.80$, Tropical Atmosphere

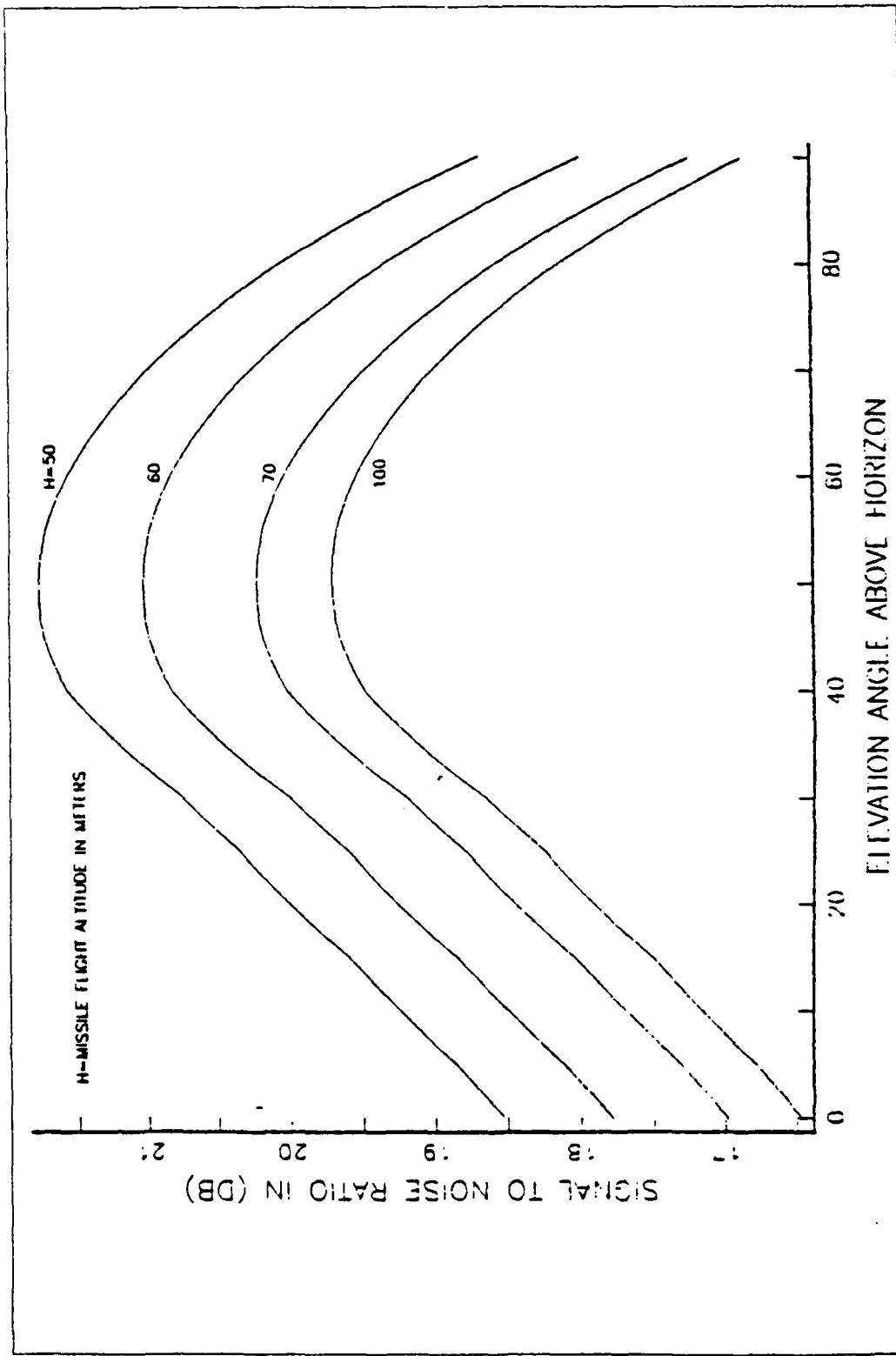


Figure 5.13 Influence of Missile Flight Altitude for Sun with Heavy Storm Clouds, $\epsilon=0.80$, Tropical Atmosphere

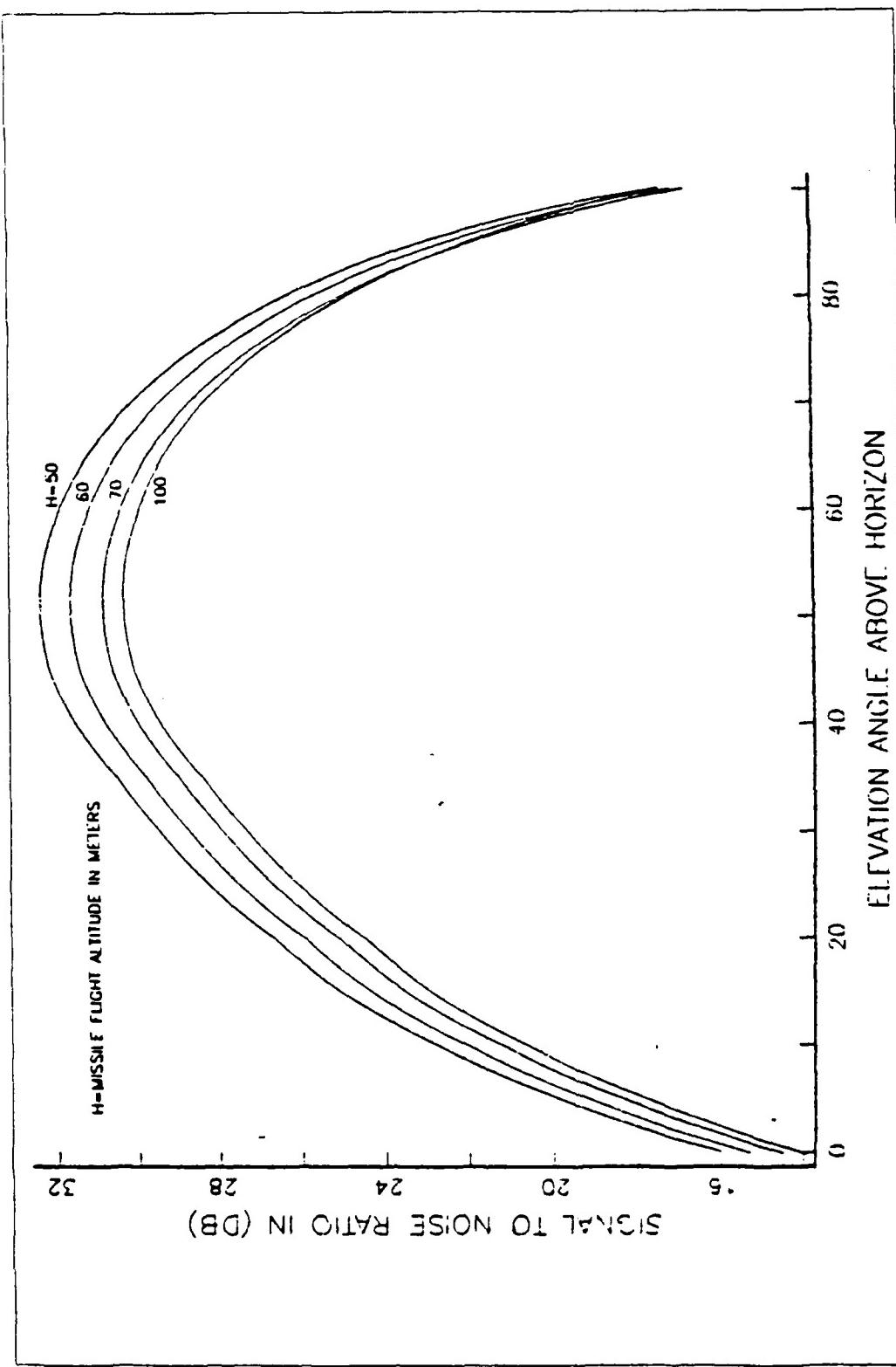


Figure 5.14 Influence of Missile Flight Altitude for Unobscured Sun, $\epsilon_0 = 0.55$, Tropical Atmosphere

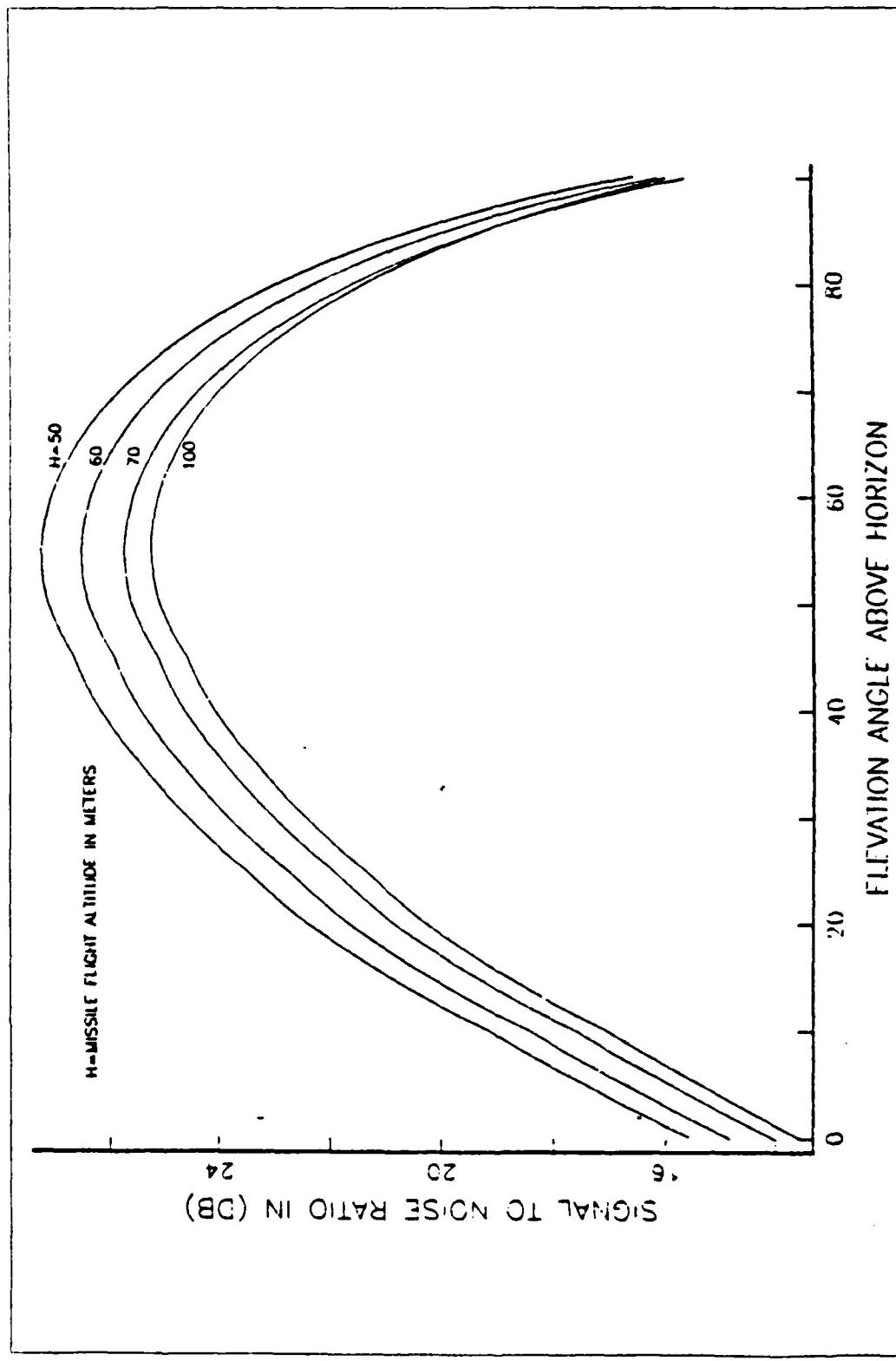


Figure 5.15 Influence of Missile Flight Altitude for Sun with Light Clouds, $\epsilon_0 = 0.55$, Tropical Atmosphere

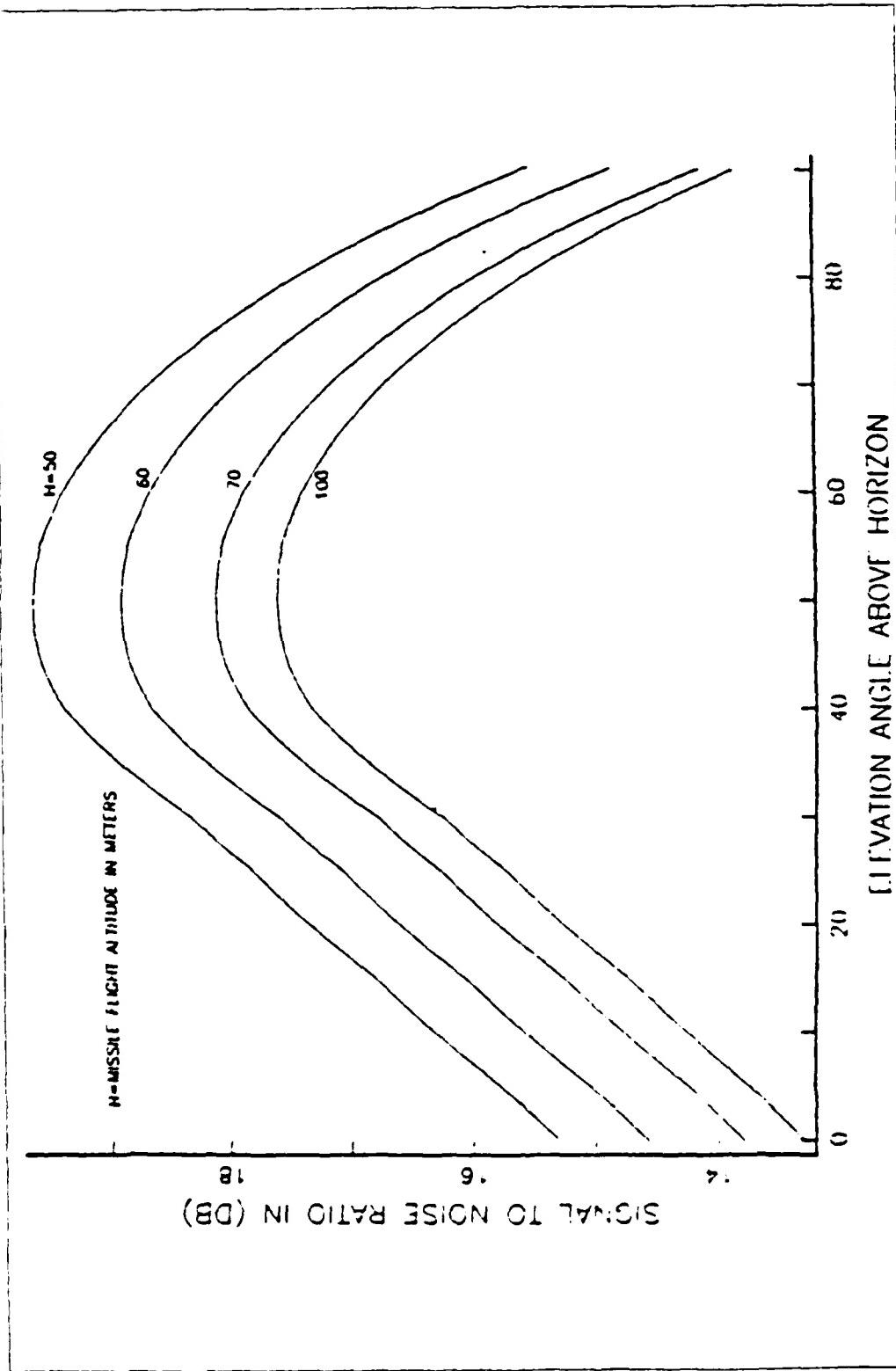


Figure 5.16 Influence of Missile Flight Altitude for Sun with Heavy Storm Clouds, $\epsilon=0.55$, Tropical Atmosphere

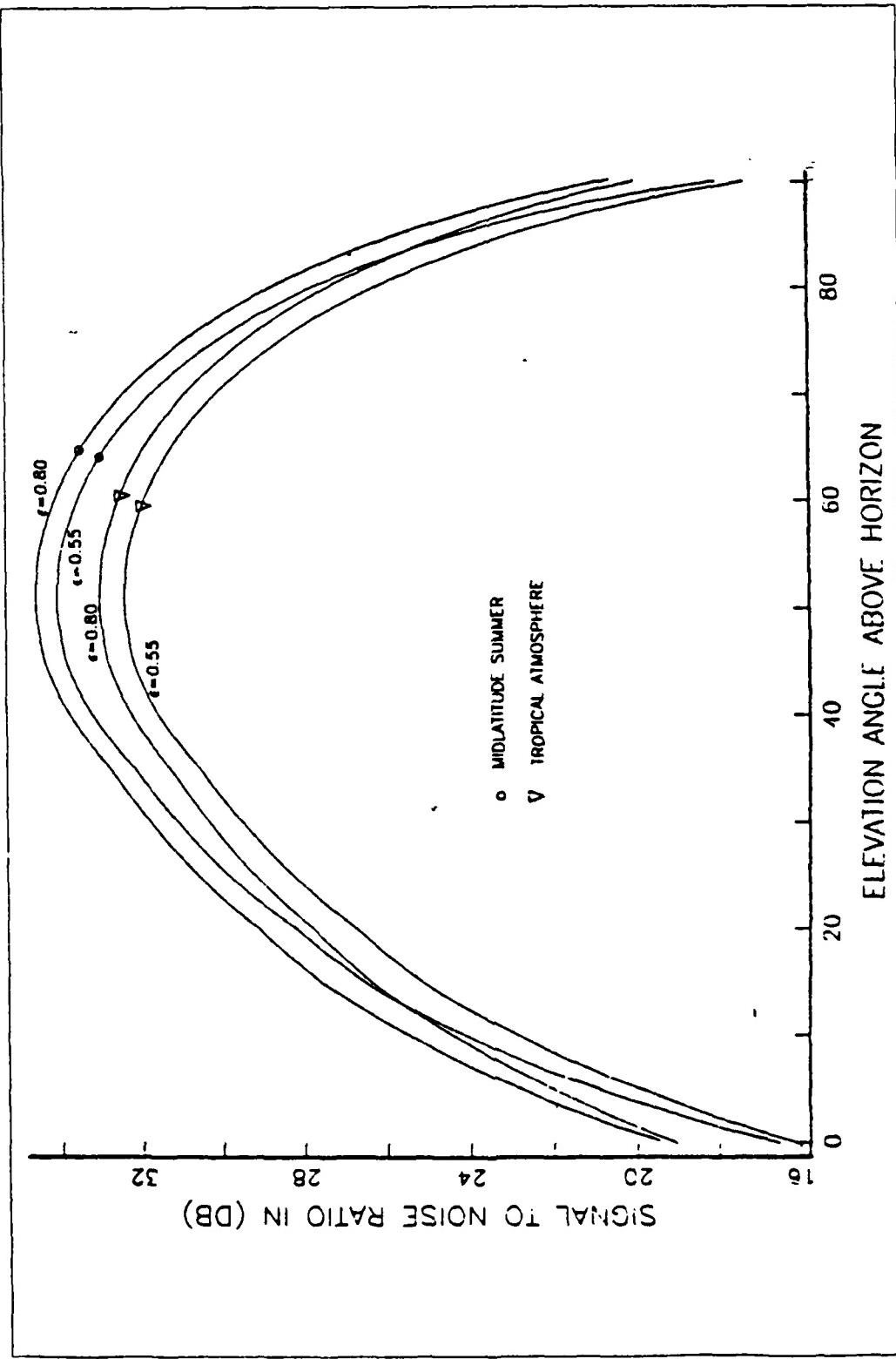


Figure 5.17 Influence of Emissivity and Atmospheric Profile for Flight Altitude 50m and Unobscured Sun

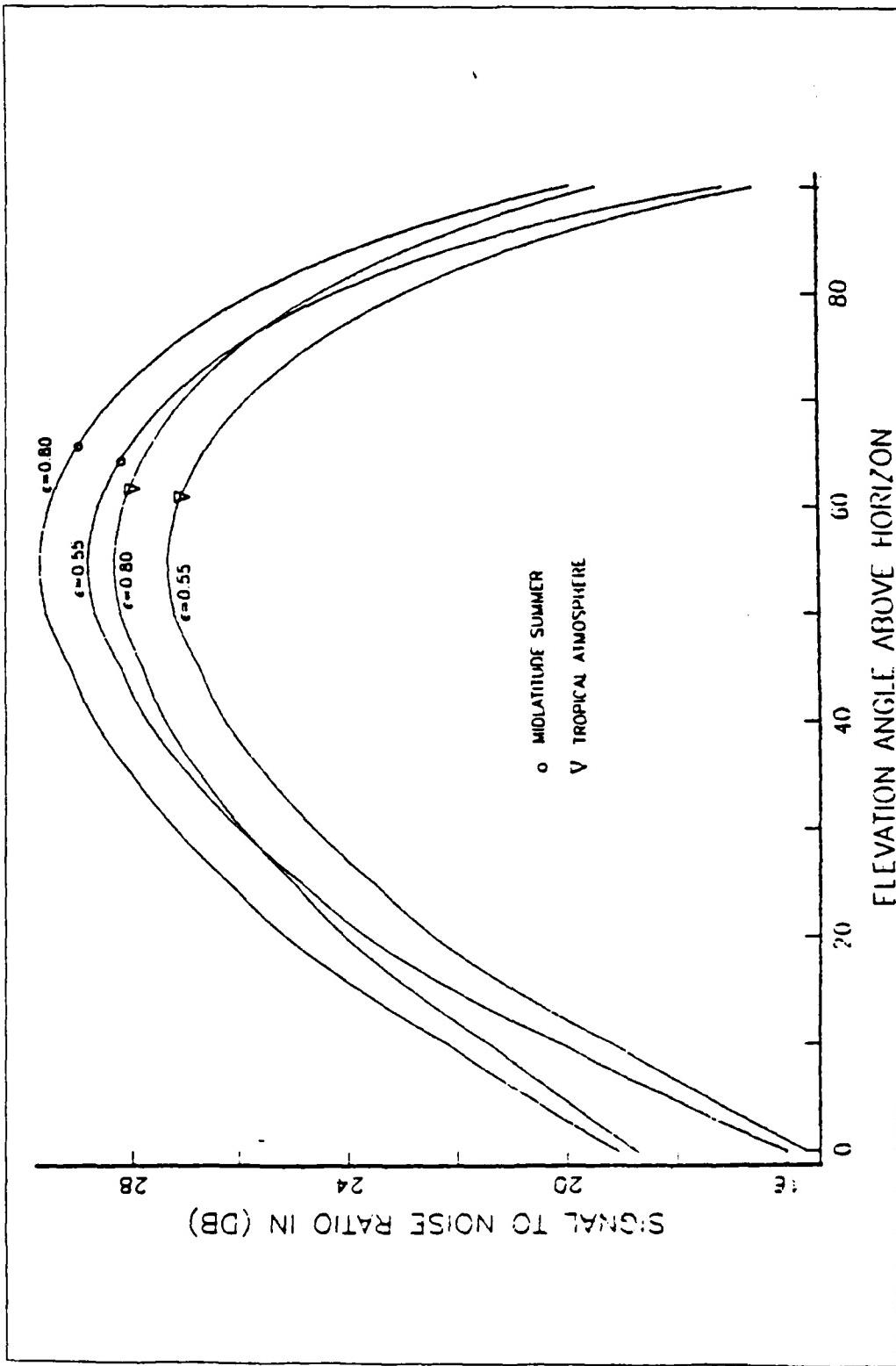


Figure 5.18 Influence of Emissivity and Atmospheric Profile for Flight Altitude 50m and Sun with Light Clouds

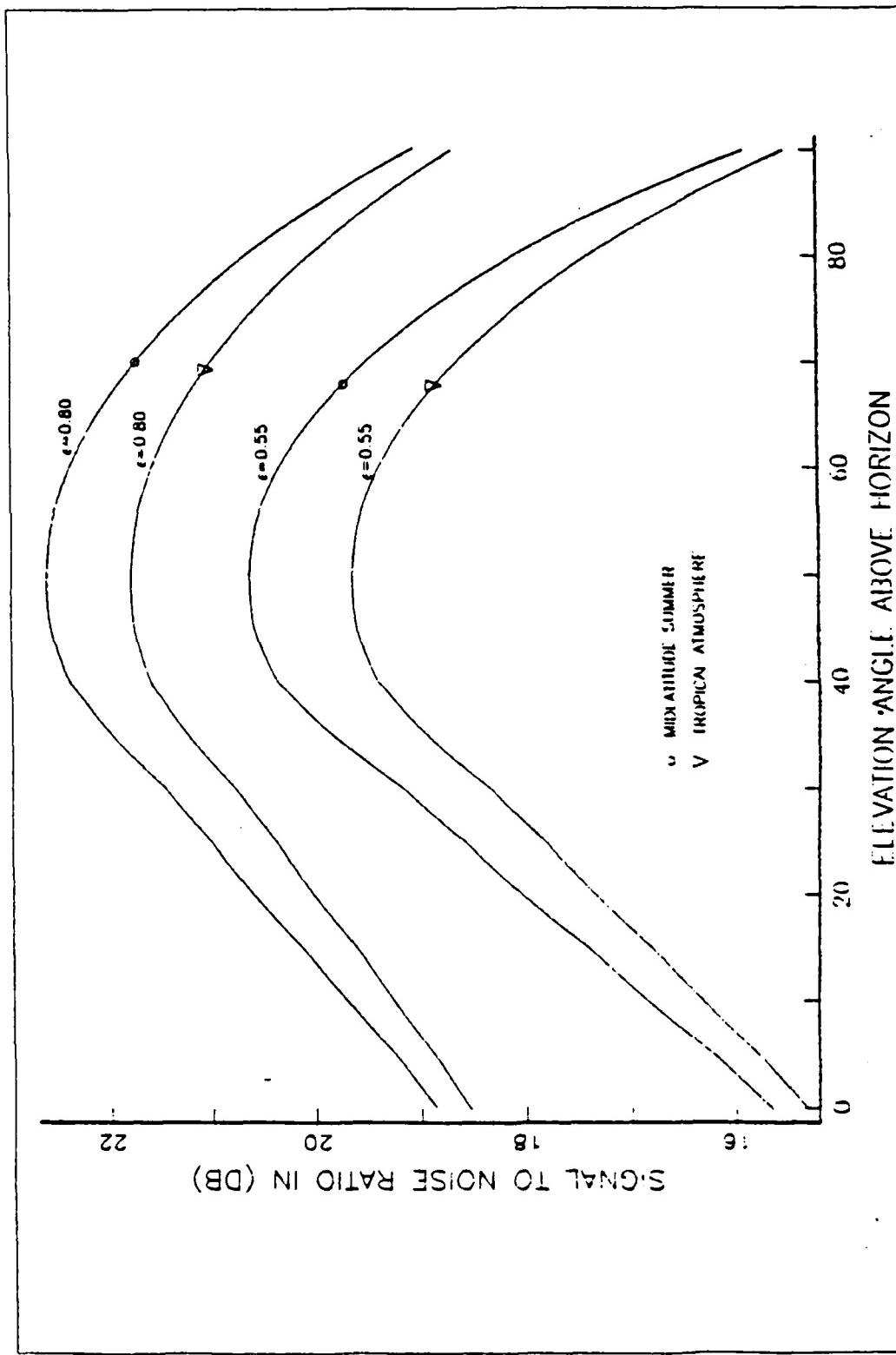


Figure 519 Influence of Emissivity and Atmospheric Profile for Flight Altitude 50m and Sun with Heavy Storm Clouds

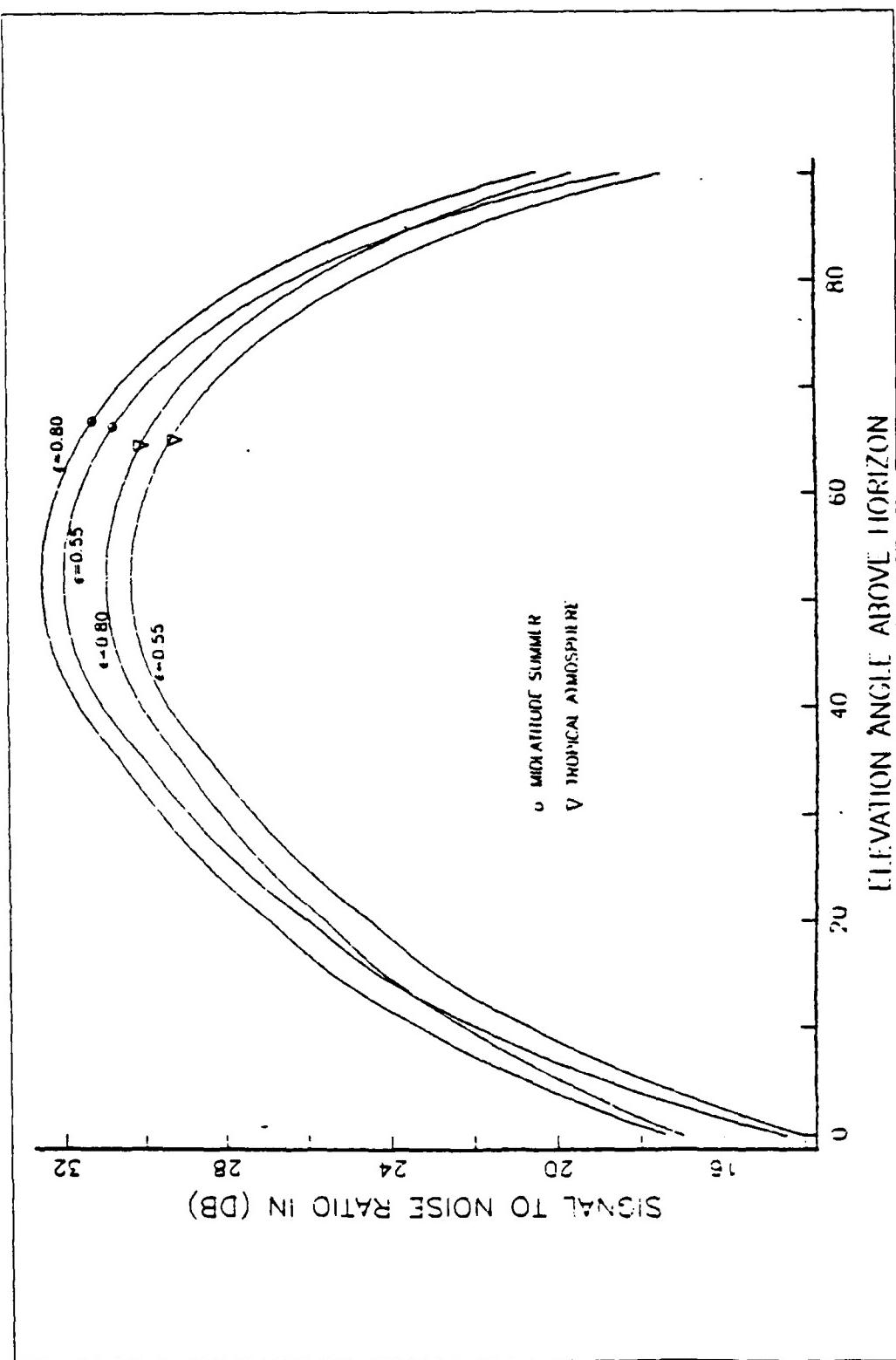


Figure 5.20 Influence of Emissivity and Atmospheric Profile
for Flight Altitude 100m and Unobscured Sun

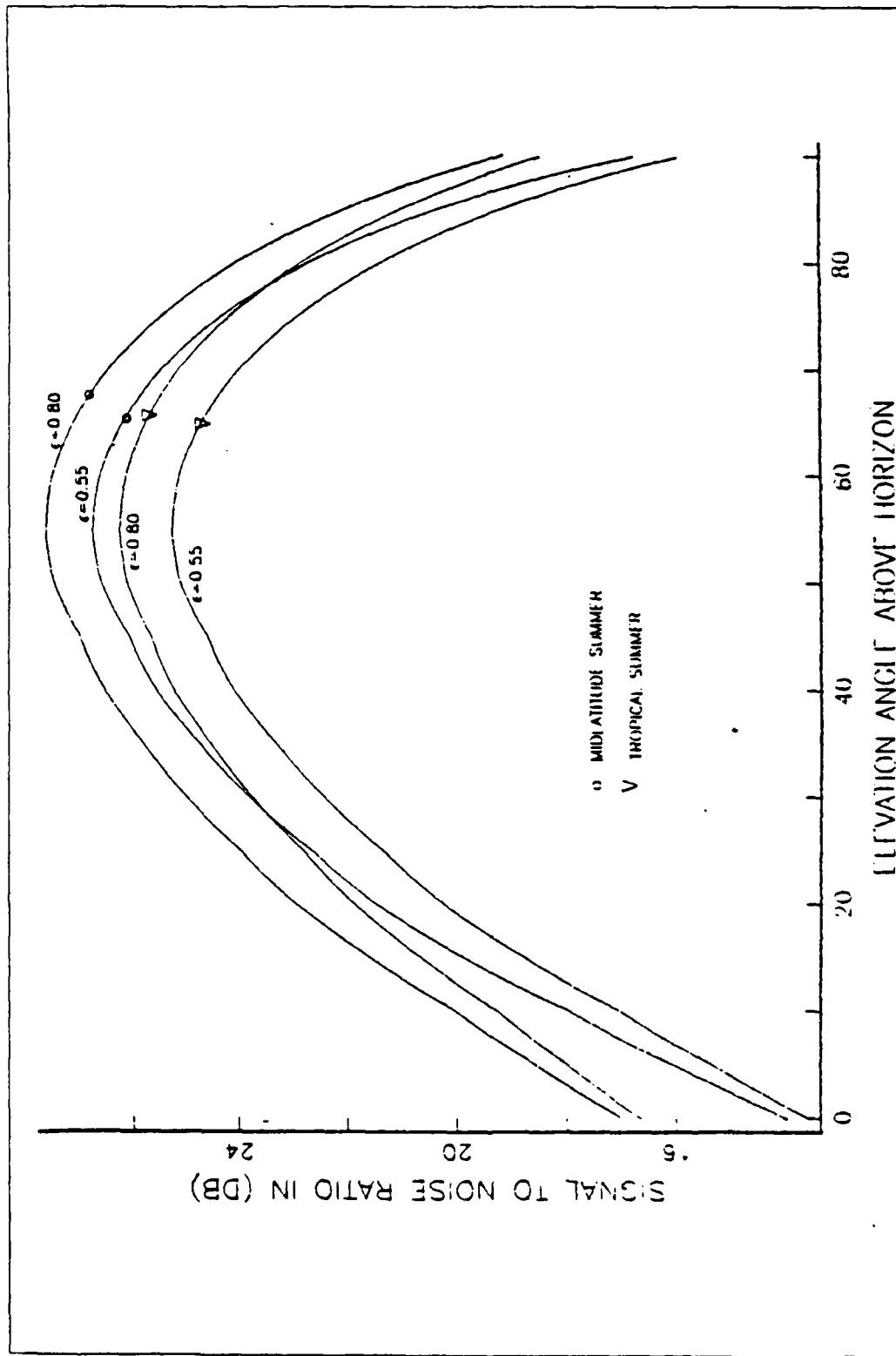


Figure 5.21 Influence of Emissivity and Atmospheric Profile for Flight Altitude 100m and Sun with Light Clouds

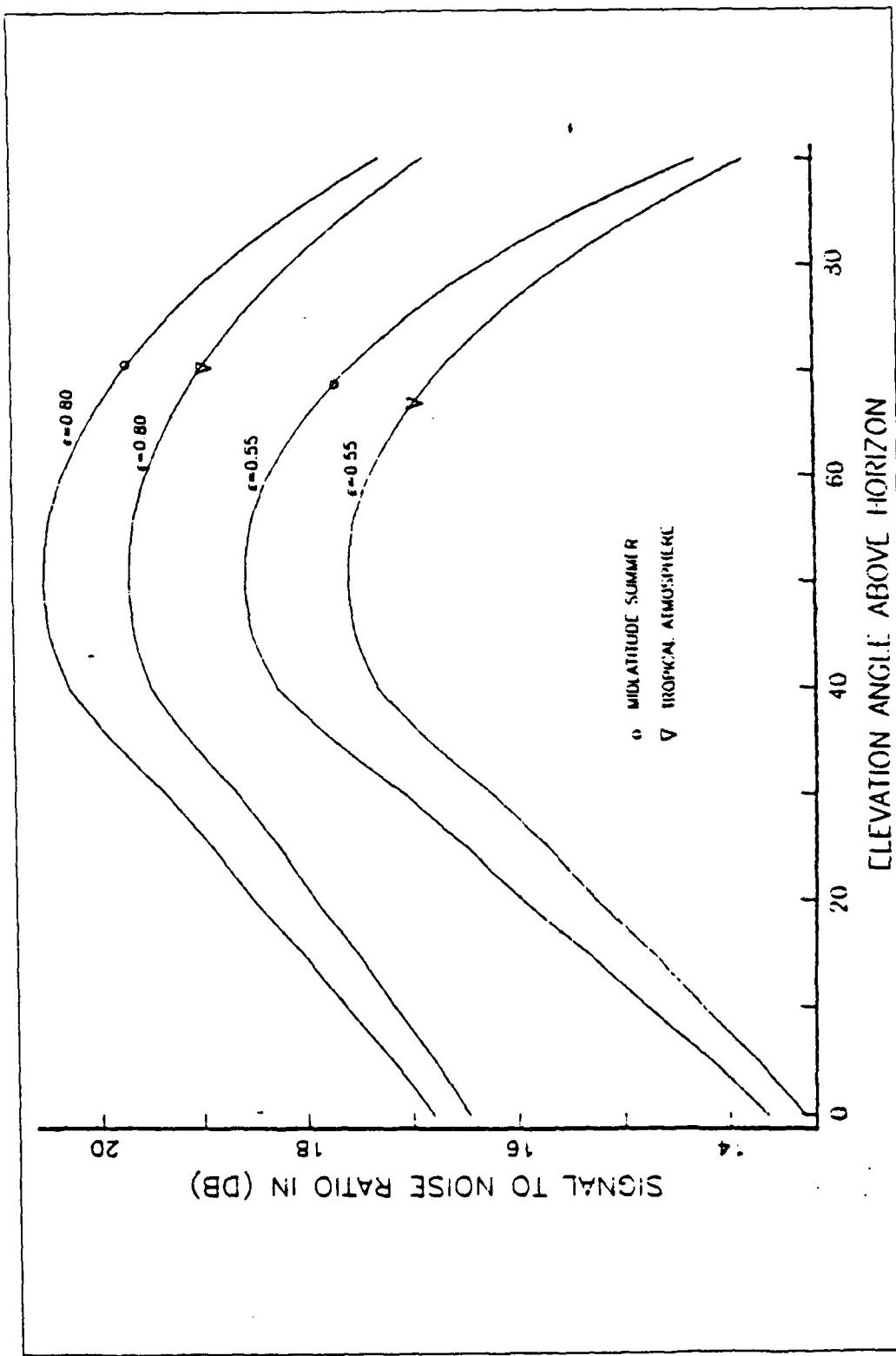


Figure 5.22 Influence of Emissivity and Atmospheric Profile for Flight Altitude 100m and Sun with Heavy Storm Clouds

VI. RESULTS AND DISCUSSION

A. SHIP BODY TEMPERATURE

Tables 5 to 18 of Appendix A show the effects of the ship internal temperature, atmospheric profile, ship body paint emissivity, sun elevation angle and sky condition on the ship body temperature.

For both horizontal and vertical faces higher paint emissivity and ship internal temperature result in higher ship body temperature. Higher internal temperature corresponds to less ship body heat losses towards the internal compartments. Moreover higher paint emissivity (absorptivity for gray paint) corresponds in higher solar energy absorption and, for the 300K ship body temperature level, the dominant mechanism of heat transfer is that of convection. Hence the increase of the paint emissivity does not correspond to a significant increase of the energy radiated by the ship body, and the net result is an increase of the ship body temperature.

The sky condition appears to be significant as well. The more clear it is the more solar energy there is incident on the ship, resulting in an increase of its body temperature.

As far as the atmospheric profile is concerned, the Tropical Atmosphere results in higher body temperature than that of Midlatitude Summer since the ambient temperature is higher and the ship body heat losses towards the environment are less.

The sun elevation angle above the horizon affects vertical and horizontal faces in different ways. We know that the incident solar energy at sea level is an increasing function of the elevation angle. For an horizontal face the

temperature appears to be an increasing function of the elevation angle, the sine of which is used for the calculation of the incident angle effect. For vertical faces the cosine of the elevation angle is used and the temperature appears a maximum in the vicinity of 50° . The explanation is that, up to this value, the increasing effect of the solar energy dominates the decreasing one of the cosine function. After the value of about 50° the first derivative of the solar energy function becomes small and the cosine behavior dominates so that the net result is a decreasing temperature function from 50° to 90° .

As we see in the ship tracked area geometric analysis is constituted from both vertical and horizontal faces, but the vertical ones have much larger projected surface areas. Therefore the radiant energy from the tracked area is due mainly to that of the vertical faces and as we will see from the S/N ratio results they appear maximum at a value of elevation angle close to 50° .

B. DETECTOR S/N RATIO

Figures 5.5 to 5.22 show the effects of the sky condition, sun elevation angle and missile flight altitude on the detector S/N ratio.

In general we observe that the variation of the S/N ratio is similar to that of the ship body temperature. That means that the radiant energy from the ship body due to its temperature is the main thermal source for the detector. At this point, it is useful to discuss the influence of the stack exit plane thermal radiation. From our calculation we know that the radiant intensity of the CO in the stack exit plane has a magnitude of the order of 10^4 ($\text{W/cm}^2\text{Sr}$) and that of the ship body of the order of 10^3 to 10^2 . Moreover the stack exit plane area is much smaller than the total tracked

area and the stack exit radiant flux is significantly smaller than this of the ship body tracked surface.

Sky condition and the paint emissivity affect S/N ratio similarly to the ship body temperature. The more clear is the sky and the higher is the paint emissivity the higher is the S/N ratio.

The atmospheric profile affects S/N in a different way. While the Tropical Atmosphere gives higher ship body temperature, it results in lower S/N ratios comparing with the Midlatitude Summer. The explanation for that is that the increase of the ship body radiance, due to its temperature increase, is less than the background noise increase (higher sea temperature). Moreover the atmospheric transmittance in the 5um window for the Tropical Atmosphere is about 10% less than this for Midlatitude Summer.

The missile flight altitude appears to be important as a parameter. The higher it is the lower is the S/N ratio and this decrease is higher between 50 to 70 (m) than between 70 to 100 (m). The explanation is that although higher missile flight altitude corresponds to lower background area (sea), the path between missile head and ship becomes larger and its square appears in the denominator of the signal voltage formula. Therefore the net result is to give lower values of S/N ratio.

The influence of the sun elevation angle above the horizon is similar to that on the ship body temperature and it has been discussed earlier.

C. RECOMMENDATIONS

Based on the relative influence of each parameter which has been varied in our analysis we recommend:

From the ship design point of view, lower internal temperature or, when it is not feasible, better wall

insulation, especially in the area where significant heat generation exist. In addition to that, paint of low emissivity and absorptivity is desirable.

From the missile point of view we recommend lower flight altitude. At this point we must notice that, lowering the flight altitude, we will reach a point where the background area (sea) will be very large, resulting in higher background noise and the coast or even the sun itself could constitute background noise for the detector. Therefore the flight altitude must be determined, in a certain design, accounting for all these conflicting parameters. Another solution could be the use of higher flight altitude, that assures us low background area, and to improve the detector optics design that will balance the loss of the signal voltage due to the greater path length.

APPENDIX A
TABLES FOR SHIP BODY TEMPERATURE

TABLE 5
Ship Body Temperature for Midlatitude Summer
 $T = 293.15K$ and $\phi = 0.80$

ELEVATION	HORIZONTAL FACE	VERTICAL FACE								
71.0	300.16		299.62		299.730		299.525		299.438	
72.0	300.20		299.623		299.737		299.520		299.437	
73.0	300.27		299.623		299.736		299.520		299.436	
74.0	300.35		299.623		299.735		299.520		299.435	
75.0	300.40		299.623		299.734		299.520		299.434	
76.0	300.45		299.623		299.733		299.520		299.433	
77.0	300.54		299.623		299.732		299.520		299.432	
78.0	300.64		299.623		299.731		299.520		299.431	
79.0	300.65		299.623		299.730		299.520		299.430	
80.0	300.67		299.623		299.729		299.520		299.429	
81.0	300.69		299.623		299.728		299.520		299.428	
82.0	300.71		299.623		299.727		299.520		299.427	
83.0	300.73		299.623		299.726		299.520		299.426	
84.0	300.74		299.623		299.725		299.520		299.425	
85.0	300.75		299.623		299.724		299.520		299.424	
86.0	300.75		299.623		299.723		299.520		299.423	
87.0	300.76		299.623		299.722		299.520		299.422	
88.0	300.77		299.623		299.721		299.520		299.421	
89.0	300.77		299.623		299.720		299.520		299.420	

TABLE 6
Ship Body Temperature for Midlatitude Summer
 $T = 303.15K$ and $\phi = 0.80$

UNOBSURED SUN		ELEVATION ANGLE	HORIZONTAL	FACE	
SUN WITH LIGHT	SUN WITH HEAVY STORM CLOUDS			VERTICAL	VERTICAL
0	0	0	0	0	0
1	1	1	1	1	1
2	2	2	2	2	2
3	3	3	3	3	3
4	4	4	4	4	4
5	5	5	5	5	5
6	6	6	6	6	6
7	7	7	7	7	7
8	8	8	8	8	8
9	9	9	9	9	9
10	10	10	10	10	10
11	11	11	11	11	11
12	12	12	12	12	12
13	13	13	13	13	13
14	14	14	14	14	14
15	15	15	15	15	15
16	16	16	16	16	16
17	17	17	17	17	17
18	18	18	18	18	18
19	19	19	19	19	19
20	20	20	20	20	20
21	21	21	21	21	21
22	22	22	22	22	22
23	23	23	23	23	23
24	24	24	24	24	24
25	25	25	25	25	25
26	26	26	26	26	26
27	27	27	27	27	27
28	28	28	28	28	28
29	29	29	29	29	29
30	30	30	30	30	30
31	31	31	31	31	31
32	32	32	32	32	32
33	33	33	33	33	33
34	34	34	34	34	34
35	35	35	35	35	35
36	36	36	36	36	36
37	37	37	37	37	37
38	38	38	38	38	38
39	39	39	39	39	39
40	40	40	40	40	40
41	41	41	41	41	41
42	42	42	42	42	42
43	43	43	43	43	43
44	44	44	44	44	44
45	45	45	45	45	45
46	46	46	46	46	46
47	47	47	47	47	47
48	48	48	48	48	48
49	49	49	49	49	49
50	50	50	50	50	50
51	51	51	51	51	51
52	52	52	52	52	52
53	53	53	53	53	53
54	54	54	54	54	54
55	55	55	55	55	55
56	56	56	56	56	56
57	57	57	57	57	57
58	58	58	58	58	58
59	59	59	59	59	59
60	60	60	60	60	60
61	61	61	61	61	61
62	62	62	62	62	62
63	63	63	63	63	63
64	64	64	64	64	64
65	65	65	65	65	65
66	66	66	66	66	66
67	67	67	67	67	67
68	68	68	68	68	68
69	69	69	69	69	69
70	70	70	70	70	70
71	71	71	71	71	71
72	72	72	72	72	72
73	73	73	73	73	73
74	74	74	74	74	74
75	75	75	75	75	75
76	76	76	76	76	76
77	77	77	77	77	77
78	78	78	78	78	78
79	79	79	79	79	79
80	80	80	80	80	80
81	81	81	81	81	81
82	82	82	82	82	82
83	83	83	83	83	83
84	84	84	84	84	84
85	85	85	85	85	85
86	86	86	86	86	86
87	87	87	87	87	87
88	88	88	88	88	88
89	89	89	89	89	89
90	90	90	90	90	90
91	91	91	91	91	91
92	92	92	92	92	92
93	93	93	93	93	93
94	94	94	94	94	94
95	95	95	95	95	95
96	96	96	96	96	96
97	97	97	97	97	97
98	98	98	98	98	98
99	99	99	99	99	99
100	100	100	100	100	100

TABLE 7
Ship Body Temperature $T = 293.15K$ for Midlatitude Summer
 $\phi = 0.55$

UNOBSCURED SUN		SUN WITH LIGHT CLOUDS		SUN WITH HEAVY STORM CLOUDS	
ELEVATION ANGLE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE	VERTICAL FACE
0.0	294.16	294.20	294.16	294.18	294.16
1.2	294.17	294.25	294.17	294.17	294.17
3.4	294.17	294.34	294.17	294.34	294.34
5.6	294.17	294.40	294.17	294.40	294.40
7.8	294.17	294.45	294.17	294.45	294.45
10.0	294.17	294.50	294.17	294.50	294.50
12.2	294.17	294.55	294.17	294.55	294.55
14.4	294.17	294.60	294.17	294.60	294.60
16.7	294.17	294.65	294.17	294.65	294.65
19.0	294.17	294.70	294.17	294.70	294.70
21.3	294.17	294.74	294.17	294.74	294.74
23.5	294.17	294.78	294.17	294.78	294.78
25.6	294.17	294.82	294.17	294.82	294.82
27.8	294.17	294.85	294.17	294.85	294.85
30.0	294.17	294.88	294.17	294.88	294.88
32.2	294.17	294.91	294.17	294.91	294.91
34.4	294.17	294.94	294.17	294.94	294.94
36.7	294.17	294.97	294.17	294.97	294.97
39.0	294.17	295.00	294.17	295.00	295.00
41.2	294.17	295.03	294.17	295.03	295.03
43.5	294.17	295.06	294.17	295.06	295.06
45.6	294.17	295.09	294.17	295.09	295.09
47.8	294.17	295.12	294.17	295.12	295.12
50.0	294.17	295.15	294.17	295.15	295.15
52.2	294.17	295.18	294.17	295.18	295.18
54.4	294.17	295.21	294.17	295.21	295.21
56.7	294.17	295.24	294.17	295.24	295.24
59.0	294.17	295.27	294.17	295.27	295.27
61.2	294.17	295.30	294.17	295.30	295.30
63.5	294.17	295.33	294.17	295.33	295.33
65.6	294.17	295.36	294.17	295.36	295.36
67.8	294.17	295.39	294.17	295.39	295.39
70.0	294.17	295.42	294.17	295.42	295.42
72.2	294.17	295.45	294.17	295.45	295.45
74.4	294.17	295.48	294.17	295.48	295.48
76.7	294.17	295.51	294.17	295.51	295.51
79.0	294.17	295.54	294.17	295.54	295.54
81.2	294.17	295.57	294.17	295.57	295.57
83.5	294.17	295.60	294.17	295.60	295.60
85.6	294.17	295.63	294.17	295.63	295.63
87.8	294.17	295.66	294.17	295.66	295.66
90.0	294.17	295.69	294.17	295.69	295.69
92.2	294.17	295.72	294.17	295.72	295.72
94.4	294.17	295.75	294.17	295.75	295.75
96.7	294.17	295.78	294.17	295.78	295.78
99.0	294.17	295.81	294.17	295.81	295.81
101.2	294.17	295.84	294.17	295.84	295.84
103.5	294.17	295.87	294.17	295.87	295.87
105.6	294.17	295.90	294.17	295.90	295.90
107.8	294.17	295.93	294.17	295.93	295.93
110.0	294.17	295.96	294.17	295.96	295.96
112.2	294.17	295.99	294.17	295.99	295.99
114.4	294.17	296.02	294.17	296.02	296.02
116.7	294.17	296.05	294.17	296.05	296.05
119.0	294.17	296.08	294.17	296.08	296.08
121.2	294.17	296.11	294.17	296.11	296.11
123.5	294.17	296.14	294.17	296.14	296.14
125.6	294.17	296.17	294.17	296.17	296.17
127.8	294.17	296.20	294.17	296.20	296.20
130.0	294.17	296.23	294.17	296.23	296.23
132.2	294.17	296.26	294.17	296.26	296.26
134.4	294.17	296.29	294.17	296.29	296.29
136.7	294.17	296.32	294.17	296.32	296.32
139.0	294.17	296.35	294.17	296.35	296.35
141.2	294.17	296.38	294.17	296.38	296.38
143.5	294.17	296.41	294.17	296.41	296.41
145.6	294.17	296.44	294.17	296.44	296.44
147.8	294.17	296.47	294.17	296.47	296.47
150.0	294.17	296.50	294.17	296.50	296.50
152.2	294.17	296.53	294.17	296.53	296.53
154.4	294.17	296.56	294.17	296.56	296.56
156.7	294.17	296.59	294.17	296.59	296.59
159.0	294.17	296.62	294.17	296.62	296.62
161.2	294.17	296.65	294.17	296.65	296.65
163.5	294.17	296.68	294.17	296.68	296.68
165.6	294.17	296.71	294.17	296.71	296.71
167.8	294.17	296.74	294.17	296.74	296.74
170.0	294.17	296.77	294.17	296.77	296.77
172.2	294.17	296.80	294.17	296.80	296.80
174.4	294.17	296.83	294.17	296.83	296.83
176.7	294.17	296.86	294.17	296.86	296.86
179.0	294.17	296.89	294.17	296.89	296.89
181.2	294.17	296.92	294.17	296.92	296.92
183.5	294.17	296.95	294.17	296.95	296.95
185.6	294.17	296.98	294.17	296.98	296.98
187.8	294.17	297.01	294.17	297.01	297.01
190.0	294.17	297.04	294.17	297.04	297.04
192.2	294.17	297.07	294.17	297.07	297.07
194.4	294.17	297.10	294.17	297.10	297.10
196.7	294.17	297.13	294.17	297.13	297.13
199.0	294.17	297.16	294.17	297.16	297.16
201.2	294.17	297.19	294.17	297.19	297.19
203.5	294.17	297.22	294.17	297.22	297.22
205.6	294.17	297.25	294.17	297.25	297.25
207.8	294.17	297.28	294.17	297.28	297.28
210.0	294.17	297.31	294.17	297.31	297.31
212.2	294.17	297.34	294.17	297.34	297.34
214.4	294.17	297.37	294.17	297.37	297.37
216.7	294.17	297.40	294.17	297.40	297.40
219.0	294.17	297.43	294.17	297.43	297.43
221.2	294.17	297.46	294.17	297.46	297.46
223.5	294.17	297.49	294.17	297.49	297.49
225.6	294.17	297.52	294.17	297.52	297.52
227.8	294.17	297.55	294.17	297.55	297.55
230.0	294.17	297.58	294.17	297.58	297.58
232.2	294.17	297.61	294.17	297.61	297.61
234.4	294.17	297.64	294.17	297.64	297.64
236.7	294.17	297.67	294.17	297.67	297.67
239.0	294.17	297.70	294.17	297.70	297.70
241.2	294.17	297.73	294.17	297.73	297.73
243.5	294.17	297.76	294.17	297.76	297.76
245.6	294.17	297.79	294.17	297.79	297.79
247.8	294.17	297.82	294.17	297.82	297.82
250.0	294.17	297.85	294.17	297.85	297.85
252.2	294.17	297.88	294.17	297.88	297.88
254.4	294.17	297.91	294.17	297.91	297.91
256.7	294.17	297.94	294.17	297.94	297.94
259.0	294.17	297.97	294.17	297.97	297.97
261.2	294.17	298.00	294.17	298.00	298.00
263.5	294.17	298.03	294.17	298.03	298.03
265.6	294.17	298.06	294.17	298.06	298.06
267.8	294.17	298.09	294.17	298.09	298.09
270.0	294.17	298.12	294.17	298.12	298.12
272.2	294.17	298.15	294.17	298.15	298.15
274.4	294.17	298.18	294.17	298.18	298.18
276.7	294.17	298.21	294.17	298.21	298.21
279.0	294.17	298.24	294.17	298.24	298.24
281.2	294.17	298.27	294.17	298.27	298.27
283.5	294.17	298.30	294.17	298.30	298.30
285.6	294.17	298.33	294.17	298.33	298.33
287.8	294.17	298.36	294.17	298.36	298.36
290.0	294.17	298.39	294.17	298.39	298.39
292.2	294.17	298.42	294.17	298.42	298.42
294.4	294.17	298.45	294.17	298.45	298.45
296.7	294.17	298.48	294.17	298.48	298.48
299.0	294.17	298.51	294.17	298.51	298.51
301.2	294.17	298.54	294.17	298.54	298.54
303.5	294.17	298.57	294.17	298.57	298.57
305.6	294.17	298.60	294.17	298.60	298.60
307.8	294.17	298.63	294.17	298.63	298.63
310.0	294.17	298.66	294.17	298.66	298.66
312.2	294.17	298.69	294.17	298.69	298.69
314.4	294.17	298.72	294.17	298.72	298.72
316.7	294.17	298.75	294.17	298.75	298.75
319.0	294.17	298.78	294.17	298.78	298.78
321.2	294.17	298.81	294.17	298.81	298.81
323.5	294.17	298.84	294.17	298.84	298.84
325.6	294.17	298.87	294.17	298.87	298.87
327.8	294.17	298.90	294.17	298.90	298.90
330.0	294.17	298.93	294.17	298.93	298.93

TABLE 8
Ship Body Temperature for Midlatitude Summer
 $T = 303.15K$ and $\delta = 0.55$

ELEVATION HORIZONTAL FACE ANGLE	UNOBSCURED SUN	SUN WITH LIGHT CLOUDS			SUN WITH HEAVY STORM CLOUDS		
		HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE
0°	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1°	1.2	1.2	1.2	1.2	1.2	1.2	1.2
2°	2.3	2.3	2.3	2.3	2.3	2.3	2.3
3°	3.4	3.4	3.4	3.4	3.4	3.4	3.4
4°	4.5	4.5	4.5	4.5	4.5	4.5	4.5
5°	5.6	5.6	5.6	5.6	5.6	5.6	5.6
6°	6.7	6.7	6.7	6.7	6.7	6.7	6.7
7°	7.8	7.8	7.8	7.8	7.8	7.8	7.8
8°	8.9	8.9	8.9	8.9	8.9	8.9	8.9
9°	10.1	10.1	10.1	10.1	10.1	10.1	10.1
10°	11.2	11.2	11.2	11.2	11.2	11.2	11.2
11°	12.3	12.3	12.3	12.3	12.3	12.3	12.3
12°	13.4	13.4	13.4	13.4	13.4	13.4	13.4
13°	14.5	14.5	14.5	14.5	14.5	14.5	14.5
14°	15.6	15.6	15.6	15.6	15.6	15.6	15.6
15°	16.7	16.7	16.7	16.7	16.7	16.7	16.7
16°	18.0	18.0	18.0	18.0	18.0	18.0	18.0
17°	20.1	20.1	20.1	20.1	20.1	20.1	20.1
18°	22.2	22.2	22.2	22.2	22.2	22.2	22.2
19°	24.3	24.3	24.3	24.3	24.3	24.3	24.3
20°	26.4	26.4	26.4	26.4	26.4	26.4	26.4
21°	28.6	28.6	28.6	28.6	28.6	28.6	28.6
22°	30.7	30.7	30.7	30.7	30.7	30.7	30.7
23°	32.9	32.9	32.9	32.9	32.9	32.9	32.9
24°	35.1	35.1	35.1	35.1	35.1	35.1	35.1
25°	37.3	37.3	37.3	37.3	37.3	37.3	37.3
26°	39.5	39.5	39.5	39.5	39.5	39.5	39.5
27°	41.7	41.7	41.7	41.7	41.7	41.7	41.7
28°	43.9	43.9	43.9	43.9	43.9	43.9	43.9
29°	46.1	46.1	46.1	46.1	46.1	46.1	46.1
30°	48.3	48.3	48.3	48.3	48.3	48.3	48.3
31°	50.5	50.5	50.5	50.5	50.5	50.5	50.5
32°	52.7	52.7	52.7	52.7	52.7	52.7	52.7
33°	54.9	54.9	54.9	54.9	54.9	54.9	54.9
34°	57.1	57.1	57.1	57.1	57.1	57.1	57.1
35°	59.3	59.3	59.3	59.3	59.3	59.3	59.3
36°	61.5	61.5	61.5	61.5	61.5	61.5	61.5
37°	63.7	63.7	63.7	63.7	63.7	63.7	63.7
38°	65.9	65.9	65.9	65.9	65.9	65.9	65.9
39°	68.1	68.1	68.1	68.1	68.1	68.1	68.1
40°	70.3	70.3	70.3	70.3	70.3	70.3	70.3
41°	72.5	72.5	72.5	72.5	72.5	72.5	72.5
42°	74.7	74.7	74.7	74.7	74.7	74.7	74.7
43°	76.9	76.9	76.9	76.9	76.9	76.9	76.9
44°	79.1	79.1	79.1	79.1	79.1	79.1	79.1
45°	81.3	81.3	81.3	81.3	81.3	81.3	81.3
46°	83.5	83.5	83.5	83.5	83.5	83.5	83.5
47°	85.7	85.7	85.7	85.7	85.7	85.7	85.7
48°	87.9	87.9	87.9	87.9	87.9	87.9	87.9
49°	90.1	90.1	90.1	90.1	90.1	90.1	90.1
50°	92.3	92.3	92.3	92.3	92.3	92.3	92.3
51°	94.5	94.5	94.5	94.5	94.5	94.5	94.5
52°	96.7	96.7	96.7	96.7	96.7	96.7	96.7
53°	98.9	98.9	98.9	98.9	98.9	98.9	98.9
54°	101.1	101.1	101.1	101.1	101.1	101.1	101.1
55°	103.3	103.3	103.3	103.3	103.3	103.3	103.3
56°	105.5	105.5	105.5	105.5	105.5	105.5	105.5
57°	107.7	107.7	107.7	107.7	107.7	107.7	107.7
58°	110.0	110.0	110.0	110.0	110.0	110.0	110.0
59°	112.2	112.2	112.2	112.2	112.2	112.2	112.2
60°	114.4	114.4	114.4	114.4	114.4	114.4	114.4
61°	116.6	116.6	116.6	116.6	116.6	116.6	116.6
62°	118.8	118.8	118.8	118.8	118.8	118.8	118.8
63°	121.0	121.0	121.0	121.0	121.0	121.0	121.0
64°	123.2	123.2	123.2	123.2	123.2	123.2	123.2
65°	125.4	125.4	125.4	125.4	125.4	125.4	125.4
66°	127.6	127.6	127.6	127.6	127.6	127.6	127.6
67°	130.0	130.0	130.0	130.0	130.0	130.0	130.0
68°	132.2	132.2	132.2	132.2	132.2	132.2	132.2
69°	134.4	134.4	134.4	134.4	134.4	134.4	134.4
70°	136.6	136.6	136.6	136.6	136.6	136.6	136.6
71°	138.8	138.8	138.8	138.8	138.8	138.8	138.8
72°	141.0	141.0	141.0	141.0	141.0	141.0	141.0
73°	143.2	143.2	143.2	143.2	143.2	143.2	143.2
74°	145.4	145.4	145.4	145.4	145.4	145.4	145.4
75°	147.6	147.6	147.6	147.6	147.6	147.6	147.6
76°	150.0	150.0	150.0	150.0	150.0	150.0	150.0
77°	152.2	152.2	152.2	152.2	152.2	152.2	152.2
78°	154.4	154.4	154.4	154.4	154.4	154.4	154.4
79°	156.6	156.6	156.6	156.6	156.6	156.6	156.6
80°	158.8	158.8	158.8	158.8	158.8	158.8	158.8
81°	161.0	161.0	161.0	161.0	161.0	161.0	161.0
82°	163.2	163.2	163.2	163.2	163.2	163.2	163.2
83°	165.4	165.4	165.4	165.4	165.4	165.4	165.4
84°	167.6	167.6	167.6	167.6	167.6	167.6	167.6
85°	170.0	170.0	170.0	170.0	170.0	170.0	170.0
86°	172.2	172.2	172.2	172.2	172.2	172.2	172.2
87°	174.4	174.4	174.4	174.4	174.4	174.4	174.4
88°	176.6	176.6	176.6	176.6	176.6	176.6	176.6
89°	178.8	178.8	178.8	178.8	178.8	178.8	178.8
90°	181.0	181.0	181.0	181.0	181.0	181.0	181.0
91°	183.2	183.2	183.2	183.2	183.2	183.2	183.2
92°	185.4	185.4	185.4	185.4	185.4	185.4	185.4
93°	187.6	187.6	187.6	187.6	187.6	187.6	187.6
94°	190.0	190.0	190.0	190.0	190.0	190.0	190.0
95°	192.2	192.2	192.2	192.2	192.2	192.2	192.2
96°	194.4	194.4	194.4	194.4	194.4	194.4	194.4
97°	196.6	196.6	196.6	196.6	196.6	196.6	196.6
98°	198.8	198.8	198.8	198.8	198.8	198.8	198.8
99°	201.0	201.0	201.0	201.0	201.0	201.0	201.0
100°	203.2	203.2	203.2	203.2	203.2	203.2	203.2
101°	205.4	205.4	205.4	205.4	205.4	205.4	205.4
102°	207.6	207.6	207.6	207.6	207.6	207.6	207.6
103°	210.0	210.0	210.0	210.0	210.0	210.0	210.0
104°	212.2	212.2	212.2	212.2	212.2	212.2	212.2
105°	214.4	214.4	214.4	214.4	214.4	214.4	214.4
106°	216.6	216.6	216.6	216.6	216.6	216.6	216.6
107°	218.8	218.8	218.8	218.8	218.8	218.8	218.8
108°	221.0	221.0	221.0	221.0	221.0	221.0	221.0
109°	223.2	223.2	223.2	223.2	223.2	223.2	223.2
110°	225.4	225.4	225.4	225.4	225.4	225.4	225.4
111°	227.6	227.6	227.6	227.6	227.6	227.6	227.6
112°	230.0	230.0	230.0	230.0	230.0	230.0	230.0
113°	232.2	232.2	232.2	232.2	232.2	232.2	232.2
114°	234.4	234.4	234.4	234.4	234.4	234.4	234.4
115°	236.6	236.6	236.6	236.6	236.6	236.6	236.6
116°	238.8	238.8	238.8	238.8	238.8	238.8	238.8
117°	241.0	241.0	241.0	241.0	241.0	241.0	241.0
118°	243.2	243.2	243.2	243.2	243.2	243.2	243.2
119°	245.4	245.4	245.4	245.4	245.4	245.4	245.4
120°	247.6	247.6	247.6	247.6	247.6	247.6	247.6
121°	250.0	250.0	250.0	250.0	250.0	250.0	250.0
122°	252.2	252.2	252.2	252.2	252.2	252.2	252.2
123°	254.4	254.4	254.4	254.4	254.4	254.4	254.4
124°	256.6	256.6	256.6	256.6	256.6	256.6	256.6
125°	258.8	258.8	258.8	258.8	258.8	258.8	258.8
126°	261.0	261.0	261.0	261.0	261.0	261.0	261.0
127°	263.2	263.2	263.2	263.2	263.2	263.2	263.2
128°	265.4	265.4	265.4	265.4	265.4	265.4	265.4
129°	267.6	267.6	267.6	267.6	267.6	267.6	267.6
130°	270.0	270.0	270.0	270.0	270.0	270.0	270.0
131°	272.2	272.2	272.2	272.2	272.2	272.2	272.2
132°	274.4	274.4	274.4	274.4	274.4	274.4	274.4
133°	276.6	276.6	276.6	276.6	276.6	276.6	276.6
134°	278.8	278.8	278.8	278.8	278.8	278.8	278.8

TABLE 9
Ship Body Temperature for Tropical
 $T = 293.15K$ and
 $= 0.80$

ELEVATION	HORIZONTAL	VERTICAL	FACE	7.2
			FACE	2.99
ANGLE	FACE	FACE	6.4	
			6.4	
ELEVATION	HORIZONTAL	VERTICAL	FACE	1.15
			FACE	0.30
ANGLE	FACE	FACE	0.64	
			0.64	
ELEVATION	HORIZONTAL	VERTICAL	FACE	0.75
			FACE	0.75
ANGLE	FACE	FACE	0.70	
			0.70	
ELEVATION	HORIZONTAL	VERTICAL	FACE	0.89
			FACE	0.89
ANGLE	FACE	FACE	0.86	
			0.86	
ELEVATION	HORIZONTAL	VERTICAL	FACE	0.68
			FACE	0.68

TABLE 10
Ship Body Temperature for Tropical
 $T = 303.15\text{K}$ and $\theta = 0.80$

AD-A153 834

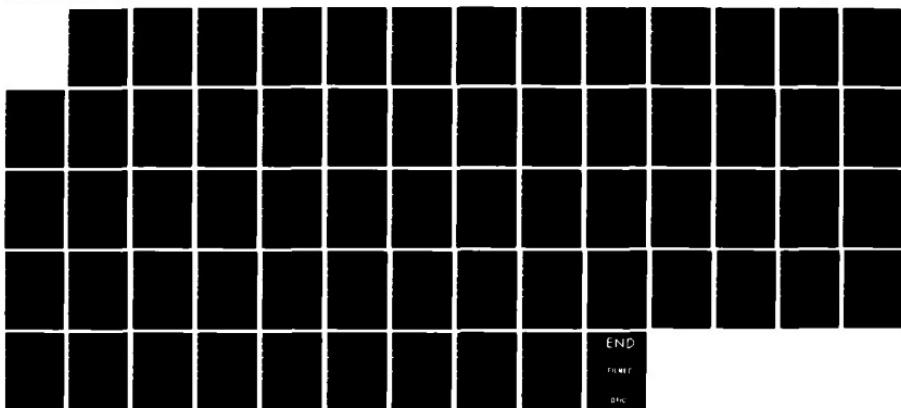
INFRARED DETECTION OF SURFACE VEHICLE CALCULATION USING 2/2
ATMOSPHERIC MODEL LOWTRAN 6(U) NAVAL POSTGRADUATE

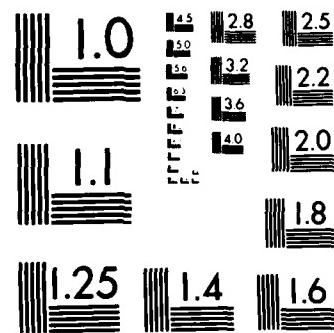
SCHOOL MONTEREY CA I EGOLFOPOULOS DEC 84

UNCLASSIFIED

F/G 17/5

NL





MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

ELEVATION ANGLE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE
71.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
72.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
73.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
74.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
75.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
76.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
77.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
78.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
79.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
80.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
81.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
82.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
83.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
84.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
85.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
86.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
87.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
88.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
89.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70
90.0	304.32	304.38	304.48	304.54	304.63	304.67	304.69	304.70

TABLE 11
Ship Body Temperature for Tropical
 $T = 293$, $T_5 = 15$ and $\theta = 0^{\circ}$

ELEVATION	HORIZONTAL ANGLE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE								
71.0	304.16	301.06	300.99	300.94	300.78	300.69	300.60	300.42	300.33	300.25	300.18	300.11
72.0	304.22	301.95	300.91	300.84	300.79	300.73	300.65	300.51	300.42	300.33	300.24	300.14
73.0	304.39	301.91	300.90	300.84	300.79	300.73	300.65	300.52	300.43	300.34	300.25	300.15
74.0	304.44	301.88	300.89	300.84	300.79	300.73	300.66	300.53	300.44	300.35	300.26	300.16
75.0	304.49	301.85	300.88	300.83	300.78	300.72	300.65	300.52	300.43	300.34	300.25	300.15
76.0	304.52	301.82	300.87	300.82	300.77	300.71	300.64	300.51	300.42	300.33	300.24	300.14
77.0	304.57	301.79	300.86	300.81	300.76	300.70	300.63	300.50	300.41	300.32	300.23	300.13
78.0	304.62	301.76	300.85	300.80	300.75	300.69	300.62	300.49	300.40	300.31	300.22	300.12
79.0	304.67	301.73	300.84	300.79	300.74	300.68	300.61	300.48	300.39	300.30	300.21	300.11
80.0	304.71	301.70	300.83	300.78	300.73	300.67	300.60	300.47	300.38	300.29	300.20	300.10
81.0	304.76	301.67	300.82	300.77	300.72	300.66	300.59	300.46	300.37	300.28	300.19	300.09
82.0	304.80	301.64	300.81	300.76	300.71	300.65	300.58	300.45	300.36	300.27	300.18	300.08
83.0	304.85	301.61	300.80	300.75	300.70	300.64	300.57	300.44	300.35	300.26	300.17	300.07
84.0	304.88	301.58	300.79	300.74	300.69	300.63	300.56	300.43	300.34	300.25	300.16	300.06
85.0	304.92	301.55	300.78	300.73	300.68	300.62	300.55	300.42	300.33	300.24	300.15	300.05
86.0	304.96	301.52	300.77	300.72	300.67	300.61	300.54	300.41	300.32	300.23	300.14	300.04
87.0	305.00	301.49	300.76	300.71	300.66	300.60	300.53	300.40	300.31	300.22	300.13	300.03
88.0	305.04	301.46	300.75	300.70	300.65	300.59	300.52	300.39	300.30	300.21	300.12	300.02
89.0	305.08	301.43	300.74	300.69	300.64	300.58	300.51	300.38	300.29	300.20	300.11	300.01
90.0	305.12	301.40	300.73	300.68	300.63	300.57	300.50	300.37	300.28	300.19	300.10	300.00

TABLE 12
Ship Body Temperature for Tropical
 $\frac{T}{T} = 303$; 15°K and
 $= 0.55$

ELEVATION ANGLE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE	HORIZONTAL FACE	VERTICAL FACE
71.0	303.01	300.94	300.42	300.21	300.22	299.97	299.96	299.95
72.0	303.05	300.89	300.37	300.24	300.24	299.92	299.91	299.90
73.0	303.09	300.82	300.32	300.25	300.25	299.87	299.86	299.85
74.0	303.14	300.76	300.27	300.21	300.21	299.82	299.81	299.80
75.0	303.19	300.70	300.22	300.15	300.15	299.77	299.76	299.75
76.0	303.23	300.65	300.17	300.11	300.11	299.72	299.71	299.70
77.0	303.28	300.60	300.12	300.05	300.05	299.67	299.66	299.65
78.0	303.33	300.55	300.07	300.01	300.01	299.62	299.61	299.60
79.0	303.38	300.50	300.02	300.00	300.00	299.57	299.56	299.55
80.0	303.43	300.45	300.07	300.01	300.01	299.52	299.51	299.50
81.0	303.48	300.40	300.12	300.06	300.06	299.47	299.46	299.45
82.0	303.53	300.35	300.17	300.10	300.10	299.42	299.41	299.40
83.0	303.58	300.30	300.22	300.09	300.09	299.37	299.36	299.35
84.0	303.63	300.25	300.27	300.10	300.10	299.32	299.31	299.30
85.0	303.68	300.20	300.32	300.09	300.09	299.27	299.26	299.25
86.0	303.73	300.15	300.37	300.08	300.08	299.22	299.21	299.20
87.0	303.78	300.10	300.42	300.07	300.07	299.17	299.16	299.15
88.0	303.83	300.05	300.47	300.06	300.06	299.12	299.11	299.10
89.0	303.88	300.00	300.52	300.07	300.07	299.07	299.06	299.05
90.0	303.93	300.00	300.57	300.08	300.08	299.02	299.01	299.00

APPENDIX B
TABLES FOR DETERMINATION OF I.F.V.

TABLE 13
Side of Instantaneous Field of View in Meters
for Flight Altitude = 50 m

DELTA	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
0.220	9.09	7.41	6.25	5.41	4.77	4.26	3.85
0.224	9.91	8.09	6.84	5.92	5.22	4.67	4.22
0.226	10.51	9.76	8.42	7.42	6.67	5.95	5.59
0.228	11.23	10.42	9.42	8.42	7.67	6.95	6.31
0.230	11.93	10.08	9.08	8.08	7.33	6.67	6.02
0.233	12.55	11.35	10.35	9.35	8.67	7.95	7.31
0.235	13.15	12.00	11.00	10.00	9.33	8.67	8.07
0.236	13.67	12.59	11.59	10.59	9.92	9.26	8.77
0.237	14.17	13.36	12.36	11.36	10.72	10.07	9.42
0.238	14.67	13.88	12.88	11.88	11.23	10.58	10.23
0.240	15.17	14.04	13.04	12.04	11.47	10.88	10.53
0.244	15.67	14.56	13.56	12.56	11.97	11.33	11.03
0.246	16.17	15.16	14.16	13.16	12.47	11.83	11.53
0.248	16.67	15.67	14.67	13.67	12.97	12.33	12.03
0.250	17.17	16.17	15.17	14.17	13.47	12.83	12.53
0.254	17.67	16.67	15.67	14.67	14.07	13.43	13.13
0.256	18.17	17.17	16.17	15.17	14.47	13.83	13.53
0.260	18.67	17.67	16.67	15.67	14.97	14.23	13.93
0.264	19.17	18.17	17.17	16.17	15.47	14.63	14.33
0.268	19.67	18.67	17.67	16.67	16.07	15.23	14.93
0.272	20.17	19.17	18.17	17.17	16.67	15.63	15.33
0.276	20.67	19.67	18.67	17.67	17.07	16.23	15.93
0.280	21.17	20.17	19.17	18.17	17.47	16.63	16.33
0.284	21.67	21.17	20.17	19.17	18.07	17.23	16.73
0.288	22.17	22.17	21.17	20.17	18.67	17.83	17.13
0.292	22.67	23.17	22.17	21.17	19.07	18.43	17.73
0.296	23.17	23.67	22.67	21.67	19.67	19.03	18.33
0.300	23.67	24.17	23.17	22.17	19.67	19.63	18.93
0.304	24.17	24.67	23.67	22.67	20.07	19.23	18.53
0.308	24.67	25.17	24.17	23.17	20.67	19.83	18.13
0.312	25.17	25.67	24.67	23.67	21.07	20.43	18.73
0.316	25.67	26.17	25.17	24.17	21.47	21.03	19.33
0.320	26.17	26.67	25.67	24.67	21.87	21.63	20.13
0.324	26.67	27.17	26.17	25.17	22.27	22.23	20.73
0.328	27.17	27.67	26.67	25.67	22.67	23.03	21.33
0.332	27.67	28.17	27.17	26.17	23.07	23.83	21.93
0.336	28.17	28.67	27.67	26.67	23.47	24.63	22.53
0.340	28.67	29.17	28.17	27.17	23.87	25.43	23.13
0.344	29.17	29.67	28.67	27.67	24.27	26.23	23.73
0.348	29.67	30.17	29.17	28.17	24.67	27.03	24.33
0.352	30.17	30.67	29.67	28.67	25.07	27.83	24.93
0.356	30.67	31.17	29.67	28.67	25.47	28.63	25.53
0.360	31.17	31.67	30.17	29.17	25.87	29.43	26.13
0.364	31.67	32.17	31.17	30.17	26.27	30.23	26.73
0.368	32.17	32.67	31.67	30.67	26.67	31.03	27.33
0.372	32.67	33.17	32.17	31.17	27.07	31.83	27.93
0.376	33.17	33.67	32.67	31.67	27.47	32.63	28.53
0.380	33.67	34.17	33.17	32.17	27.87	33.43	29.13
0.384	34.17	34.67	33.67	32.67	28.27	34.23	29.73
0.388	34.67	35.17	34.17	33.17	28.67	35.03	30.33
0.392	35.17	35.67	34.67	33.67	29.07	35.83	30.93
0.396	35.67	36.17	35.17	34.17	29.47	36.63	31.53
0.400	36.17	36.67	35.67	34.67	29.87	37.43	32.13
0.404	36.67	37.17	36.17	35.17	30.27	38.23	32.73
0.408	37.17	37.67	36.67	35.67	30.67	39.03	33.33
0.412	37.67	38.17	37.17	36.17	31.07	39.83	33.93
0.416	38.17	38.67	37.67	36.67	31.47	40.63	34.53
0.420	38.67	39.17	38.17	37.17	31.87	41.43	35.13
0.424	39.17	39.67	38.67	37.67	32.27	42.23	35.73
0.428	39.67	40.17	39.17	38.17	32.67	43.03	36.33
0.432	40.17	40.67	39.67	38.67	33.07	43.83	36.93
0.436	40.67	41.17	40.17	39.17	33.47	44.63	37.53
0.440	41.17	41.67	40.67	39.67	33.87	45.43	38.13
0.444	41.67	42.17	41.17	40.17	34.27	46.23	38.73
0.448	42.17	42.67	41.67	40.67	34.67	47.03	39.33
0.452	42.67	43.17	42.17	41.17	35.07	47.83	39.93
0.456	43.17	43.67	42.67	41.67	35.47	48.63	40.53
0.460	43.67	44.17	43.17	42.17	35.87	49.43	41.13
0.464	44.17	44.67	43.67	42.67	36.27	50.23	41.73
0.468	44.67	45.17	44.17	43.17	36.67	51.03	42.33
0.472	45.17	45.67	44.67	43.67	37.07	51.83	42.93
0.476	45.67	46.17	45.17	44.17	37.47	52.63	43.53
0.480	46.17	46.67	45.67	44.67	37.87	53.43	44.13
0.484	46.67	47.17	46.17	45.17	38.27	54.23	44.73
0.488	47.17	47.67	46.67	45.67	38.67	55.03	45.33
0.492	47.67	48.17	47.17	46.17	39.07	55.83	45.93
0.496	48.17	48.67	47.67	46.67	39.47	56.63	46.53
0.500	48.67	49.17	48.17	47.17	39.87	57.43	47.13

TABLE 14
Path in Meters for Flight Altitude = 50 m

DELTA	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
0.20	1302.43	1061.37	895.68	774.78	682.68	610.19	551.66
0.24	11290.70	1053.59	845.62	770.61	679.45	605.25	547.07
0.28	11267.99	1045.50	835.53	766.53	676.39	600.20	544.07
0.32	11256.67	1041.43	831.42	762.43	672.24	596.15	543.97
0.36	11245.55	1037.35	827.31	758.24	668.10	592.03	543.87
0.40	11234.42	1033.27	823.20	754.04	664.87	587.90	543.77
0.44	11223.30	1029.19	819.09	749.84	660.64	582.77	543.67
0.48	11212.17	1025.11	814.97	745.64	656.41	577.64	543.57
0.52	11201.04	1020.99	810.85	741.44	652.18	572.51	543.47
0.56	11190.92	1016.81	806.73	737.24	647.95	567.38	543.37
0.60	11180.80	1012.63	802.61	733.04	643.72	562.25	543.27
0.64	11170.67	1008.45	798.49	728.84	639.49	557.12	543.17
0.68	11160.54	1004.27	794.37	724.64	635.26	551.99	543.07
0.72	11150.41	1000.10	790.25	720.44	631.03	546.86	542.96
0.76	11140.28	995.92	786.13	695.24	590.00	531.75	542.86
0.80	11130.15	990.75	782.01	690.04	585.80	526.62	542.75
0.84	11120.02	985.57	777.89	685.84	580.57	521.42	542.65
0.88	11110.89	980.39	773.77	681.64	575.34	516.19	542.55
0.92	11100.76	975.21	769.65	677.44	570.11	510.96	542.45
0.96	11090.63	970.03	765.53	673.24	564.88	505.73	542.35
1.00	11080.50	964.85	761.41	669.04	559.65	500.50	542.25
1.04	11070.37	959.67	757.29	664.84	554.42	495.27	542.15
1.08	11060.24	954.49	753.17	660.64	549.19	490.04	542.05
1.12	11050.11	949.31	749.05	656.44	543.92	484.81	541.95
1.16	11040.00	944.13	744.93	652.24	538.69	479.56	541.85
1.20	11030.87	938.95	740.81	648.04	533.46	474.33	541.75
1.24	11020.74	933.77	736.69	643.84	528.23	469.09	541.65
1.28	11010.61	928.59	732.57	639.64	523.00	463.86	541.55
1.32	11000.48	923.41	728.45	635.44	517.77	458.63	541.45
1.36	10990.35	918.23	724.33	631.24	512.54	453.40	541.35
1.40	10980.22	913.05	720.21	627.04	507.31	448.17	541.25
1.44	10970.09	907.87	716.09	622.84	502.08	442.94	541.15
1.48	10960.96	902.69	711.97	618.64	496.85	437.71	541.05
1.52	10950.83	897.51	707.85	614.44	491.62	432.49	540.95
1.56	10940.70	892.33	703.73	610.24	486.39	427.26	540.85
1.60	10930.57	887.15	699.61	596.04	481.16	421.93	540.75
1.64	10920.44	881.97	695.49	591.84	475.93	416.69	540.65
1.68	10910.31	876.79	691.37	587.64	470.70	411.46	540.55
1.72	10900.18	871.61	687.25	583.44	465.47	406.23	540.45
1.76	10890.05	866.43	683.13	579.24	460.24	401.00	540.35
1.80	10880.92	861.25	678.99	575.04	454.99	395.77	540.25
1.84	10870.79	856.07	674.87	570.84	449.76	390.54	540.15
1.88	10860.66	850.89	670.75	566.64	444.53	385.31	540.05
1.92	10850.53	845.71	666.63	562.44	439.30	380.08	539.95
1.96	10840.40	840.53	662.51	558.24	434.07	374.85	539.85
2.00	10830.27	835.35	658.39	554.04	428.84	369.62	539.75
2.04	10820.14	830.17	654.27	549.84	423.61	364.39	539.65
2.08	10810.01	824.99	650.15	545.64	418.38	359.16	539.55
2.12	10800.88	819.81	646.03	541.44	413.15	353.93	539.45
2.16	10790.75	814.63	641.91	537.24	407.92	348.70	539.35
2.20	10780.62	809.45	637.79	533.04	402.69	343.47	539.25
2.24	10770.49	804.27	633.67	528.84	397.46	338.24	539.15
2.28	10760.36	800.09	629.55	524.64	392.23	332.99	539.05
2.32	10750.23	794.91	625.43	520.44	386.99	327.76	538.95
2.36	10740.10	789.73	621.31	516.24	381.76	322.53	538.85
2.40	10730.97	784.55	617.19	512.04	376.53	317.30	538.75
2.44	10720.84	779.37	613.07	507.84	371.30	312.07	538.65
2.48	10710.71	774.19	608.95	503.64	366.07	306.84	538.55
2.52	10700.58	768.99	604.83	499.44	360.84	301.61	538.45
2.56	10690.45	763.81	600.71	495.24	355.61	296.38	538.35
2.60	10680.32	758.63	596.59	491.04	350.38	291.15	538.25
2.64	10670.19	753.45	592.47	486.84	345.15	285.92	538.15
2.68	10660.06	748.27	588.35	482.64	340.92	280.69	538.05
2.72	10650.93	743.09	584.23	478.44	335.69	275.46	537.95
2.76	10640.80	737.91	579.11	474.24	330.46	270.23	537.85
2.80	10630.67	732.73	574.99	469.94	325.23	264.99	537.75
2.84	10620.54	727.55	570.87	465.74	320.00	259.76	537.65
2.88	10610.41	722.37	566.75	461.54	314.77	254.53	537.55
2.92	10600.28	717.19	562.63	457.34	309.54	249.30	537.45
2.96	10590.15	711.99	558.51	453.14	304.31	244.07	537.35
3.00	10580.02	706.81	554.39	448.94	299.08	238.84	537.25
3.04	10570.89	698.63	550.27	444.74	293.85	233.61	537.15
3.08	10560.76	693.45	546.15	440.54	288.62	228.38	537.05
3.12	10550.63	688.27	542.03	436.34	283.39	223.15	536.95
3.16	10540.50	683.09	537.91	432.14	278.16	217.92	536.85
3.20	10530.37	677.91	533.79	427.94	272.93	212.69	536.75
3.24	10520.24	672.73	529.67	423.74	267.70	207.46	536.65
3.28	10510.11	667.55	525.55	419.54	262.47	202.23	536.55
3.32	10500.98	662.37	521.43	415.34	257.24	196.99	536.45
3.36	10490.85	657.19	517.31	411.14	251.99	191.76	536.35
3.40	10480.72	652.01	513.19	406.94	246.76	186.53	536.25
3.44	10470.59	646.83	509.07	402.74	241.53	181.30	536.15
3.48	10460.46	641.65	504.95	398.54	236.30	176.07	536.05
3.52	10450.33	636.47	500.83	394.34	231.07	170.84	535.95
3.56	10440.20	631.29	496.71	384.14	225.84	165.61	535.85
3.60	10430.07	626.11	492.59	379.94	220.61	160.38	535.75
3.64	10420.94	620.93	488.47	375.74	215.38	155.15	535.65
3.68	10410.81	615.75	484.35	371.54	210.15	150.92	535.55
3.72	10400.68	610.57	480.23	367.34	204.92	145.69	535.45
3.76	10390.55	605.39	476.11	363.14	199.69	140.46	535.35
3.80	10380.42	600.21	471.99	358.94	194.46	135.23	535.25
3.84	10370.29	595.03	467.87	354.74	189.23	130.00	535.15
3.88	10360.16	589.85	463.75	350.54	184.00	124.77	535.05
3.92	10350.03	584.67	459.63	346.34	178.77	119.54	534.95
3.96	10340.90	579.49	455.51	342.14	173.54	114.31	534.85
4.00	10330.77	574.31	451.39	337.94	168.31	109.08	534.75
4.04	10320.64	569.13	447.27	333.74	163.08	103.85	534.65
4.08	10310.51	563.95	443.15	329.54	157.85	98.62	534.55
4.12	10300.38	558.77	438.93	325.34	152.62	93.39	534.45
4.16	10290.25	553.59	434.71	321.14	147.39	88.16	534.35
4.20	10280.12	548.41	429.50	316.94	142.16	82.93	534.25
4.24	10270.00	543.23	424.28	312.74	136.93	77.70	534.15
4.28	10260.87	538.05	419.06	308.54	131.70	72.47	534.05
4.32	10250.74	532.87	413.84	304.34	126.47	67.24	533.95
4.36	10240.61	527.69	408.62	300.14	121.24	62.01	533.85
4.40	10230.48	522.51	403.40	295.94	115.91	56.78	533.75
4.44	10220.35	517.33	398.18	291.74	110.68	51.55	533.65
4.48	10210.22	512.15	392.96	287.54	105.45	46.32	533.55
4.52	10200.09	506.97	387.74	283.34	100.22	41.09	533.45
4.56	10190.96	501.79	382.52	279.14	94.99	35.86	533.35
4.60	10180.83	496.61	377.30	274.94	89.76	30.63	533.25
4.64	10170.70	491.43	372.08	270.74	84.53	25.40	533.15
4.68	10160.57	486.25	366.86	266.54	79.30	20.17	533.05
4.72	10150.44	481.07	361.64	262.34	74.07	14.94	532.95
4.76	10140.31	475.89	356.42	258.14	68.84	9.71	532.85
4.80	10130.18	470.71	351.20	253.94	63.61	4.48	532.75
4.84	10120.05	465.53	345.98	249.74	58.38	-0.55	532.65
4.88	10110.92	460.35	340.76	245.54	53.15	-5.32	532.55
4							

TABLE 15
Side of Instantaneous Field of View in Meters
for Flight Altitude = 60 m

DELTA	PHI=2.00	PHI=2.50	PHI=3.00	PHI=3.50	PHI=4.00	PHI=4.50	PHI=5.00
0.20	10.91	8.89	7.50	6.49	5.72	5.11	4.62
0.24	11.86	10.51	8.89	7.10	6.26	5.60	5.06
0.30	11.31	11.20	8.58	7.70	6.30	5.54	5.04
0.32	11.47	11.25	8.05	7.86	6.90	6.12	5.67
0.34	11.56	11.29	8.99	8.90	8.25	7.25	6.80
0.36	11.44	11.20	9.06	8.88	8.47	7.42	6.98
0.38	11.65	11.24	9.17	9.00	8.67	7.72	7.22
0.42	11.60	11.16	9.22	9.06	8.90	8.12	7.60
0.44	11.74	11.14	9.26	9.00	8.90	8.12	7.60
0.46	11.55	11.15	9.13	9.06	8.90	8.12	7.60
0.48	11.67	11.14	9.14	9.06	8.90	8.12	7.60
0.50	11.61	11.14	9.14	9.06	8.90	8.12	7.60
0.54	11.78	11.17	9.17	9.07	8.97	8.17	7.67
0.56	11.72	11.17	9.17	9.07	8.97	8.17	7.67
0.58	11.66	11.17	9.19	9.07	8.97	8.17	7.67
0.60	11.70	11.17	9.19	9.07	8.97	8.17	7.67
0.64	11.81	11.21	9.23	9.14	9.07	9.12	8.74
0.67	11.78	11.21	9.23	9.14	9.07	9.12	8.74
0.70	11.83	11.21	9.23	9.14	9.07	9.12	8.74
0.74	11.80	11.21	9.23	9.14	9.07	9.12	8.74
0.77	11.87	11.21	9.23	9.14	9.07	9.12	8.74
0.80	11.89	11.21	9.23	9.14	9.07	9.12	8.74
0.84	11.82	11.21	9.23	9.14	9.07	9.12	8.74
0.87	11.85	11.21	9.23	9.14	9.07	9.12	8.74
0.91	11.86	11.21	9.23	9.14	9.07	9.12	8.74
0.94	11.83	11.21	9.23	9.14	9.07	9.12	8.74
0.97	11.87	11.21	9.23	9.14	9.07	9.12	8.74
1.00	11.80	11.21	9.23	9.14	9.07	9.12	8.74
1.04	11.88	11.21	9.23	9.14	9.07	9.12	8.74
1.07	11.85	11.21	9.23	9.14	9.07	9.12	8.74
1.11	11.82	11.21	9.23	9.14	9.07	9.12	8.74
1.14	11.87	11.21	9.23	9.14	9.07	9.12	8.74
1.17	11.84	11.21	9.23	9.14	9.07	9.12	8.74
1.21	11.81	11.21	9.23	9.14	9.07	9.12	8.74
1.24	11.86	11.21	9.23	9.14	9.07	9.12	8.74
1.27	11.83	11.21	9.23	9.14	9.07	9.12	8.74
1.31	11.88	11.21	9.23	9.14	9.07	9.12	8.74
1.34	11.85	11.21	9.23	9.14	9.07	9.12	8.74
1.37	11.82	11.21	9.23	9.14	9.07	9.12	8.74
1.41	11.87	11.21	9.23	9.14	9.07	9.12	8.74
1.44	11.84	11.21	9.23	9.14	9.07	9.12	8.74
1.47	11.81	11.21	9.23	9.14	9.07	9.12	8.74
1.51	11.86	11.21	9.23	9.14	9.07	9.12	8.74
1.54	11.83	11.21	9.23	9.14	9.07	9.12	8.74
1.57	11.88	11.21	9.23	9.14	9.07	9.12	8.74
1.61	11.85	11.21	9.23	9.14	9.07	9.12	8.74
1.64	11.82	11.21	9.23	9.14	9.07	9.12	8.74
1.67	11.87	11.21	9.23	9.14	9.07	9.12	8.74
1.71	11.84	11.21	9.23	9.14	9.07	9.12	8.74
1.74	11.81	11.21	9.23	9.14	9.07	9.12	8.74
1.77	11.86	11.21	9.23	9.14	9.07	9.12	8.74
1.81	11.83	11.21	9.23	9.14	9.07	9.12	8.74
1.84	11.88	11.21	9.23	9.14	9.07	9.12	8.74
1.87	11.85	11.21	9.23	9.14	9.07	9.12	8.74
1.91	11.82	11.21	9.23	9.14	9.07	9.12	8.74
1.94	11.87	11.21	9.23	9.14	9.07	9.12	8.74
1.97	11.84	11.21	9.23	9.14	9.07	9.12	8.74
2.01	11.81	11.21	9.23	9.14	9.07	9.12	8.74
2.04	11.86	11.21	9.23	9.14	9.07	9.12	8.74
2.07	11.83	11.21	9.23	9.14	9.07	9.12	8.74
2.11	11.88	11.21	9.23	9.14	9.07	9.12	8.74
2.14	11.85	11.21	9.23	9.14	9.07	9.12	8.74
2.17	11.82	11.21	9.23	9.14	9.07	9.12	8.74
2.21	11.87	11.21	9.23	9.14	9.07	9.12	8.74
2.24	11.84	11.21	9.23	9.14	9.07	9.12	8.74
2.27	11.81	11.21	9.23	9.14	9.07	9.12	8.74
2.31	11.86	11.21	9.23	9.14	9.07	9.12	8.74
2.34	11.83	11.21	9.23	9.14	9.07	9.12	8.74
2.37	11.88	11.21	9.23	9.14	9.07	9.12	8.74
2.41	11.85	11.21	9.23	9.14	9.07	9.12	8.74
2.44	11.82	11.21	9.23	9.14	9.07	9.12	8.74
2.47	11.87	11.21	9.23	9.14	9.07	9.12	8.74
2.51	11.84	11.21	9.23	9.14	9.07	9.12	8.74
2.54	11.81	11.21	9.23	9.14	9.07	9.12	8.74
2.58	11.86	11.21	9.23	9.14	9.07	9.12	8.74
2.60	11.83	11.21	9.23	9.14	9.07	9.12	8.74

TABLE 16
Path in Meters for Flight Altitude = 60 m

DELTA	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
0.224	1562.91	1273.64	1074.81	929.73	819.21	732.23	661.99
0.226	1548.84	1264.31	1068.14	924.74	815.34	726.14	659.47
0.335	1535.05	1255.09	1055.07	914.91	809.50	717.09	654.46
0.443	1521.40	1245.04	1042.62	905.31	803.20	711.09	651.50
0.547	1508.02	1237.04	1032.90	895.05	800.31	705.24	647.23
0.643	1495.02	1228.04	1020.46	884.09	796.40	700.57	644.29
0.740	1482.67	1219.42	1010.46	874.09	789.20	699.57	641.54
0.837	1469.54	1202.10	1002.31	864.09	785.20	696.18	639.71
0.934	1457.14	1194.54	992.09	854.09	782.20	693.55	637.18
1.032	1444.87	1185.74	982.09	844.09	779.20	690.99	635.48
1.130	1432.30	1177.74	972.09	834.09	776.20	688.36	633.04
1.228	1419.64	1169.74	962.09	824.09	773.20	685.72	630.49
1.326	1406.97	1161.74	952.09	814.09	770.20	683.09	628.04
1.424	1394.29	1153.74	942.09	804.09	767.20	680.45	625.55
1.522	1381.62	1145.74	932.09	794.09	764.20	677.85	623.97
1.620	1368.95	1137.74	922.09	784.09	761.20	675.25	621.30
1.718	1356.28	1129.74	912.09	774.09	758.20	672.63	619.71
1.816	1343.61	1121.74	902.09	764.09	755.20	670.97	618.02
1.914	1330.94	1113.74	892.09	754.09	752.20	668.30	617.02
2.012	1318.27	1105.74	882.09	744.09	749.20	666.74	614.81
2.110	1305.60	1097.74	872.09	734.09	746.20	665.18	613.21
2.208	1292.93	1089.74	862.09	724.09	743.20	663.62	612.63
2.306	1280.26	1081.74	852.09	714.09	740.20	662.05	611.97
2.404	1267.59	1073.74	842.09	704.09	737.20	660.48	611.44
2.502	1254.92	1065.74	832.09	694.09	734.20	658.91	610.87
2.600	1242.25	1057.74	822.09	684.09	731.20	657.34	609.71
2.798	1229.58	1049.74	812.09	674.09	728.20	655.77	608.57
2.896	1216.91	1041.74	802.09	664.09	725.20	654.20	607.47
2.994	1204.24	1033.74	792.09	654.09	722.20	652.63	606.31
3.092	1191.57	1025.74	782.09	644.09	719.20	651.06	605.21
3.190	1178.90	1017.74	772.09	634.09	716.20	649.49	604.11
3.288	1166.23	1009.74	762.09	624.09	713.20	648.93	603.01
3.386	1153.56	999.74	752.09	614.09	710.20	648.36	601.91
3.484	1140.89	991.74	742.09	604.09	707.20	647.79	600.81
3.582	1128.22	983.74	732.09	594.09	704.20	647.22	599.71
3.680	1115.55	975.74	722.09	584.09	701.20	646.65	598.61
3.778	1102.88	967.74	712.09	574.09	698.20	646.08	597.51
3.876	1090.21	959.74	702.09	564.09	695.20	645.51	596.41
3.974	1077.54	951.74	692.09	554.09	692.20	644.94	595.31
4.072	1064.87	943.74	682.09	544.09	689.20	644.37	594.21
4.170	1052.20	935.74	672.09	534.09	686.20	643.79	593.11
4.268	1039.53	927.74	662.09	524.09	683.20	643.22	592.01
4.366	1026.86	919.74	652.09	514.09	680.20	642.65	590.91
4.464	1014.19	911.74	642.09	504.09	677.20	642.08	589.81
4.562	1001.52	903.74	632.09	494.09	674.20	641.51	588.71
4.660	988.85	895.74	622.09	484.09	671.20	640.94	587.61
4.758	976.18	887.74	612.09	474.09	668.20	640.37	586.51
4.856	963.51	879.74	602.09	464.09	665.20	639.79	585.41
4.954	950.84	871.74	592.09	454.09	662.20	639.22	584.31
5.052	938.17	863.74	582.09	444.09	659.20	638.65	583.21
5.150	925.50	855.74	572.09	434.09	656.20	638.08	582.11
5.248	912.83	847.74	562.09	424.09	653.20	637.51	581.01
5.346	899.16	839.74	552.09	414.09	650.20	636.94	580.91
5.444	886.49	831.74	542.09	404.09	647.20	636.37	579.81
5.542	873.82	823.74	532.09	394.09	644.20	635.79	578.71
5.640	861.15	815.74	522.09	384.09	641.20	635.22	577.61
5.738	848.48	807.74	512.09	374.09	638.20	634.65	576.51
5.836	835.81	799.74	502.09	364.09	635.20	634.08	575.41
5.934	823.14	791.74	492.09	354.09	632.20	633.51	574.31
6.032	810.47	783.74	482.09	344.09	629.20	632.94	573.21
6.130	797.80	775.74	472.09	334.09	626.20	632.37	572.11
6.228	785.13	767.74	462.09	324.09	623.20	631.80	571.01
6.326	772.46	759.74	452.09	314.09	620.20	631.23	570.91
6.424	759.79	751.74	442.09	304.09	617.20	630.66	569.81
6.522	747.12	743.74	432.09	294.09	614.20	630.10	568.71
6.620	734.45	735.74	422.09	284.09	611.20	629.53	567.61
6.718	721.78	727.74	412.09	274.09	608.20	628.96	566.51
6.816	709.11	719.74	402.09	264.09	605.20	628.39	565.41
6.914	696.44	711.74	392.09	254.09	602.20	627.82	564.31
7.012	683.77	703.74	382.09	244.09	599.20	627.25	563.21
7.110	671.10	695.74	372.09	234.09	596.20	626.68	562.11
7.208	658.43	687.74	362.09	224.09	593.20	626.11	561.01
7.306	645.76	679.74	352.09	214.09	590.20	625.54	560.91
7.404	633.09	671.74	342.09	204.09	587.20	624.97	559.81
7.492	620.42	663.74	332.09	194.09	584.20	624.40	558.71
7.590	607.75	655.74	322.09	184.09	581.20	623.83	557.61
7.688	595.08	647.74	312.09	174.09	578.20	623.26	556.51
7.786	582.41	639.74	302.09	164.09	575.20	622.69	555.41
7.884	569.74	631.74	292.09	154.09	572.20	622.12	554.31
7.982	557.07	623.74	282.09	144.09	569.20	621.55	553.21
8.080	544.40	615.74	272.09	134.09	566.20	620.98	552.11
8.178	531.73	607.74	262.09	124.09	563.20	620.41	551.01
8.276	519.06	599.74	252.09	114.09	560.20	619.84	549.91
8.374	506.39	591.74	242.09	104.09	557.20	619.27	548.81
8.472	493.72	583.74	232.09	94.09	554.20	618.70	547.71
8.570	481.05	575.74	222.09	84.09	551.20	618.13	546.61
8.668	468.38	567.74	212.09	74.09	548.20	617.56	545.51
8.766	455.71	559.74	202.09	64.09	545.20	616.99	544.41
8.864	443.04	551.74	192.09	54.09	542.20	616.42	543.31
8.962	430.37	543.74	182.09	44.09	539.20	615.85	542.21
9.060	417.70	535.74	172.09	34.09	536.20	615.28	541.11
9.158	405.03	527.74	162.09	24.09	533.20	614.71	540.01
9.256	392.36	519.74	152.09	14.09	530.20	614.14	538.91
9.354	379.69	511.74	142.09	04.09	527.20	613.57	537.81
9.452	367.02	503.74	132.09	-4.09	524.20	613.00	536.71
9.550	354.35	495.74	122.09	-14.09	521.20	612.43	535.61
9.648	341.68	487.74	112.09	-24.09	518.20	611.86	534.51
9.746	329.01	479.74	102.09	-34.09	515.20	611.29	533.41
9.844	316.34	471.74	92.09	-44.09	512.20	610.72	532.31
9.942	303.67	463.74	82.09	-54.09	509.20	610.15	531.21
10.040	291.00	455.74	72.09	-64.09	506.20	609.58	530.11
10.138	278.33	447.74	62.09	-74.09	503.20	609.01	529.01
10.236	265.66	439.74	52.09	-84.09	500.20	608.44	527.91
10.334	252.99	431.74	42.09	-94.09	497.20	607.87	526.81
10.432	240.32	423.74	32.09	-104.09	494.20	607.30	525.71
10.530	227.65	415.74	22.09	-114.09	491.20	606.73	524.61
10.628	214.98	407.74	12.09	-124.09	488.20	606.16	523.51
10.726	202.31	399.74	1.09	-134.09	485.20	605.59	522.41
10.824	189.64	391.74	-8.99	-144.09	482.20	605.02	521.31
10.922	176.97	383.74	-18.99	-154.09	479.20	604.45	520.21
11.020	164.30	375.74	-28.99	-164.09	476.20	603.88	519.11
11.118	151.63	367.74	-38.99	-174.09	473.20	603.31	518.01
11.216	138.96	359.74	-48.99	-184.09	470.20	602.74	516.91
11.314	126.29	351.74	-58.99	-194.09	467.20	602.17	515.81
11.412	113.62	343.74	-68.99	-204.09	464.20	601.60	514.71
11.510	100.95	335.74	-78.99	-214.09	461.20	601.03	513.61
11.608	88.28	327.74	-88.99	-224.09	458.20	590.46	512.51
11.706	75.61	319.74	-98.99	-234.09	455.20	589.89	511.41
11.794	62.94	311.74	-108.99</td				

TABLE 17
Side of Instantaneous Field of View in Meters
for Flight Altitude = 70 m

DELTA	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
0.20	12.73	10.37	8.75	7.57	6.67	5.96	5.39
0.24	13.73	11.34	9.58	8.29	7.93	7.42	6.93
0.28	14.73	12.34	10.47	9.09	8.43	7.88	7.43
0.32	15.73	13.34	11.34	10.07	9.39	8.89	8.43
0.36	16.73	14.34	12.21	11.04	10.01	9.47	9.00
0.40	17.73	15.34	13.07	11.91	10.68	9.80	9.32
0.44	18.73	16.34	13.94	12.80	11.35	10.14	9.61
0.48	19.73	17.34	14.80	13.67	12.02	10.47	9.89
0.52	20.73	18.34	15.65	14.52	12.67	10.77	10.14
0.56	21.73	19.34	16.50	15.37	13.32	11.04	10.42
0.60	22.73	20.34	17.35	16.20	14.07	11.34	10.71
0.64	23.73	21.34	18.20	17.04	14.81	11.61	10.99
0.68	24.73	22.34	19.05	17.87	15.55	12.87	11.24
0.72	25.73	23.34	19.89	18.70	16.30	13.74	11.51
0.76	26.73	24.34	20.73	19.53	17.05	14.57	11.78
0.80	27.73	25.34	21.57	20.36	17.80	15.34	12.02
0.84	28.73	26.34	22.41	21.19	18.55	16.11	12.29
0.88	29.73	27.34	23.25	22.02	19.30	16.88	12.56
0.92	30.73	28.34	24.09	22.85	19.05	17.64	13.32
0.96	31.73	29.34	24.93	23.68	19.80	18.40	14.08
1.00	32.73	30.34	25.77	24.51	20.55	19.17	14.82
1.04	33.73	31.34	26.61	25.34	21.30	19.94	15.59
1.08	34.73	32.34	27.45	26.17	22.05	20.71	16.46
1.12	35.73	33.34	28.29	26.99	22.80	21.48	17.24
1.16	36.73	34.34	29.13	27.82	23.55	22.25	18.01
1.20	37.73	35.34	29.97	28.65	24.30	22.97	18.78
1.24	38.73	36.34	30.81	29.48	25.05	23.74	19.55
1.28	39.73	37.34	31.65	30.31	25.80	24.51	20.32
1.32	40.73	38.34	32.49	31.14	26.55	25.28	21.09
1.36	41.73	39.34	33.33	31.97	27.30	26.05	21.86
1.40	42.73	40.34	34.17	32.80	28.05	26.82	22.63
1.44	43.73	41.34	35.01	33.63	28.80	27.59	23.40
1.48	44.73	42.34	35.85	34.46	29.55	28.36	24.17
1.52	45.73	43.34	36.69	35.29	30.30	29.13	24.94
1.56	46.73	44.34	37.53	36.12	31.05	29.90	25.71
1.60	47.73	45.34	38.37	36.95	31.80	30.67	26.48
1.64	48.73	46.34	39.21	37.78	32.55	31.44	27.25
1.68	49.73	47.34	39.95	38.61	33.30	32.21	28.02
1.72	50.73	48.34	40.79	39.44	34.05	32.98	28.79
1.76	51.73	49.34	41.63	40.27	34.80	33.75	29.56
1.80	52.73	50.34	42.47	41.10	35.55	34.52	30.33
1.84	53.73	51.34	43.31	41.93	36.30	35.29	31.10
1.88	54.73	52.34	44.15	42.76	37.05	36.06	31.87
1.92	55.73	53.34	44.99	43.59	37.80	36.83	32.64
1.96	56.73	54.34	45.83	44.42	38.55	37.60	33.41
2.00	57.73	55.34	46.67	45.25	39.30	38.37	34.18
2.04	58.73	56.34	47.51	46.08	39.05	39.14	34.95
2.08	59.73	57.34	48.35	46.91	39.80	39.91	35.72
2.12	60.73	58.34	49.19	47.74	40.55	40.68	36.49
2.16	61.73	59.34	50.03	48.57	41.30	41.45	37.26
2.20	62.73	60.34	50.87	49.40	42.05	42.22	38.03
2.24	63.73	61.34	51.71	49.23	42.80	42.99	38.80
2.28	64.73	62.34	52.55	49.06	43.55	43.76	39.57
2.32	65.73	63.34	53.39	48.89	44.30	44.53	40.34
2.36	66.73	64.34	54.23	48.72	45.05	45.30	41.11
2.40	67.73	65.34	55.07	48.55	45.80	46.07	41.88
2.44	68.73	66.34	55.91	48.38	46.55	46.84	42.65
2.48	69.73	67.34	56.75	48.21	47.30	47.61	43.42
2.52	70.73	68.34	57.59	48.04	48.05	48.38	44.19
2.56	71.73	69.34	58.43	47.87	48.80	49.15	44.96
2.60	72.73	70.34	59.27	47.70	49.55	49.92	45.73
2.64	73.73	71.34	60.11	47.53	50.30	50.69	46.50
2.68	74.73	72.34	60.95	47.36	51.05	51.46	47.27
2.72	75.73	73.34	61.79	47.19	51.80	52.23	48.04
2.76	76.73	74.34	62.63	47.02	52.55	52.99	48.81
2.80	77.73	75.34	63.47	46.85	53.30	53.76	49.58
2.84	78.73	76.34	64.31	46.68	54.05	54.53	50.35
2.88	79.73	77.34	65.15	46.51	54.80	55.30	51.12
2.92	80.73	78.34	65.99	46.34	55.55	56.07	51.89
2.96	81.73	79.34	66.83	46.17	56.30	56.84	52.66
3.00	82.73	80.34	67.67	45.00	57.05	57.61	53.43
3.04	83.73	81.34	68.51	44.83	57.80	58.38	54.20
3.08	84.73	82.34	69.35	44.66	58.55	59.15	54.97
3.12	85.73	83.34	70.19	44.49	59.30	59.92	55.74
3.16	86.73	84.34	71.03	44.32	60.05	60.69	56.51
3.20	87.73	85.34	71.87	44.15	60.80	61.46	57.28
3.24	88.73	86.34	72.71	43.98	61.55	62.23	58.05
3.28	89.73	87.34	73.55	43.81	62.30	63.00	58.82
3.32	90.73	88.34	74.39	43.64	63.05	63.77	59.59
3.36	91.73	89.34	75.23	43.47	63.80	64.54	60.36
3.40	92.73	90.34	76.07	43.30	64.55	65.31	61.13
3.44	93.73	91.34	76.91	43.13	65.30	66.08	61.90
3.48	94.73	92.34	77.75	42.96	66.05	66.85	62.67
3.52	95.73	93.34	78.59	42.79	66.80	67.62	63.44
3.56	96.73	94.34	79.43	42.62	67.55	68.39	64.21
3.60	97.73	95.34	80.27	42.45	68.30	69.16	64.98
3.64	98.73	96.34	81.11	42.28	69.05	69.93	65.75
3.68	99.73	97.34	81.95	42.11	69.80	70.70	66.52
3.72	100.73	98.34	82.79	41.94	70.55	71.47	67.29
3.76	101.73	99.34	83.63	41.77	71.30	72.24	68.06
3.80	102.73	100.34	84.47	41.60	72.05	73.01	68.83
3.84	103.73	101.34	85.31	41.43	72.80	73.78	69.60
3.88	104.73	102.34	86.15	41.26	73.55	74.55	70.37
3.92	105.73	103.34	86.99	41.09	74.30	75.32	71.14
3.96	106.73	104.34	87.83	40.92	75.05	76.09	71.91
4.00	107.73	105.34	88.67	40.75	75.80	76.86	72.68
4.04	108.73	106.34	89.51	40.58	76.55	77.63	73.45
4.08	109.73	107.34	90.35	40.41	77.30	78.40	74.22
4.12	110.73	108.34	91.19	40.24	78.05	79.17	74.99
4.16	111.73	109.34	92.03	40.07	78.80	79.94	75.76
4.20	112.73	110.34	92.87	39.90	79.55	80.71	76.53
4.24	113.73	111.34	93.71	39.73	80.30	81.48	77.30
4.28	114.73	112.34	94.55	39.56	81.05	82.25	78.07
4.32	115.73	113.34	95.39	39.39	81.80	83.02	78.84
4.36	116.73	114.34	96.23	39.22	82.55	83.79	79.61
4.40	117.73	115.34	97.07	39.05	83.30	84.56	80.38
4.44	118.73	116.34	97.91	38.88	84.05	85.33	81.15
4.48	119.73	117.34	98.75	38.71	84.80	86.10	81.92
4.52	120.73	118.34	99.59	38.54	85.55	86.87	82.69
4.56	121.73	119.34	100.43	38.37	86.30	87.64	83.46
4.60	122.73	120.34	101.27	38.20	87.05	88.41	84.23
4.64	123.73	121.34	102.11	38.03	87.80	89.18	84.99
4.68	124.73	122.34	102.95	37.86	88.55	89.95	85.76
4.72	125.73	123.34	103.79	37.69	89.30	90.72	86.53
4.76	126.73	124.34	104.63	37.52	90.05	91.49	87.30
4.80	127.73	125.34	105.47	37.35	90.80	92.26	88.07
4.84	128.73	126.34	106.31	37.18	91.55	93.03	88.84
4.88	129.73	127.34	107.15	37.01	92.30	93.80	89.61
4.92	130.73	128.34	107.99	36.84	93.05	94.57	90.38
4.96	131.73	129.34	108.83	36.67	93.80	95.34	91.15
5.00	132.73	130.34	109.67	36.50	94.55	96.11	91.92

TABLE 18
Path in Meters for Flight Altitude=70 m

DELTA	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
0.220	1823.40	1485.92	1253.95	1084.69	955.75	854.27	772.32
0.224	1806.89	1475.93	1246.97	1078.76	946.76	850.69	769.30

TABLE 19

Side of Instantaneous Field of View in Meters
for Flight Altitude = 80 m

DELTA	PHI=2.00	PHI=2.50	PHI=3.00	PHI=3.50	PHI=4.00	PHI=4.50	PHI=5.00
0.224	14.55	11.86	10.00	8.65	7.63	6.82	6.16
0.224	15.81	12.45	10.86	9.47	8.06	7.47	6.75
0.224	17.89	14.56	12.67	11.27	9.97	8.57	7.92
0.224	20.00	17.00	15.00	13.00	11.00	9.00	7.50
0.224	22.23	19.45	17.45	15.45	13.45	11.45	9.95
0.224	24.42	21.89	19.89	17.89	15.89	13.89	11.89
0.224	26.60	24.07	21.07	18.07	15.07	12.07	9.97
0.224	28.78	26.25	23.25	20.25	17.25	14.25	11.25
0.224	30.90	28.34	25.34	22.34	19.34	16.34	13.34
0.224	33.03	30.45	27.45	24.45	21.45	18.45	15.45
0.224	35.16	32.57	29.57	26.57	23.57	20.57	17.57
0.224	37.29	34.67	31.67	28.67	25.67	22.67	19.67
0.224	39.42	36.78	33.78	30.78	27.78	24.78	21.78
0.224	41.55	38.89	35.89	32.89	29.89	26.89	23.89
0.224	43.68	40.90	37.90	34.90	31.90	28.90	25.90
0.224	45.81	43.02	40.02	37.02	34.02	31.02	28.02
0.224	47.94	45.14	42.14	39.14	36.14	33.14	30.14
0.224	50.07	47.25	44.25	41.25	38.25	35.25	32.25
0.224	52.20	49.34	46.34	43.34	40.34	37.34	34.34
0.224	54.33	51.45	48.45	45.45	42.45	39.45	36.45
0.224	56.46	53.57	50.57	47.57	44.57	41.57	38.57
0.224	58.59	55.67	52.67	49.67	46.67	43.67	40.67
0.224	60.72	57.78	54.78	51.78	48.78	45.78	42.78
0.224	62.85	59.89	56.89	53.89	50.89	47.89	44.89
0.224	64.98	61.90	58.90	55.90	52.90	49.90	46.90
0.224	67.11	64.02	60.92	57.92	54.92	51.92	48.92
0.224	69.24	66.14	63.04	59.94	56.94	53.94	50.94
0.224	71.37	68.25	65.15	62.05	58.95	55.85	52.85
0.224	73.50	70.34	67.24	64.14	61.04	57.94	54.84
0.224	75.63	72.45	69.35	66.25	63.15	59.95	56.85
0.224	77.76	74.57	71.47	68.37	65.27	62.17	58.97
0.224	79.89	76.67	73.57	70.47	67.37	64.27	61.07
0.224	82.02	78.78	75.68	72.58	69.48	66.38	63.18
0.224	84.15	80.90	77.80	74.70	71.60	68.50	65.30
0.224	86.28	83.02	80.02	76.92	73.92	70.82	67.62
0.224	88.41	85.14	82.14	78.94	75.94	72.84	69.64
0.224	90.54	87.25	84.25	80.95	77.95	74.85	71.65
0.224	92.67	89.34	86.34	82.94	79.94	76.84	73.64
0.224	94.80	91.45	88.45	84.95	81.95	78.85	75.65
0.224	96.93	93.57	90.57	86.97	83.97	80.87	77.67
0.224	99.06	95.67	92.67	88.97	85.97	82.87	79.67
0.224	101.19	97.78	94.78	90.98	87.98	84.88	81.68
0.224	103.32	99.89	96.89	92.99	89.99	86.89	83.69
0.224	105.45	101.90	98.90	94.90	91.90	88.80	85.60
0.224	107.58	103.92	100.92	96.92	93.92	90.82	87.62
0.224	109.71	105.94	102.94	98.94	95.94	92.84	89.64
0.224	111.84	107.95	104.95	100.95	97.95	94.85	91.65
0.224	113.97	109.97	106.97	102.97	99.97	96.87	93.67
0.224	116.10	111.98	108.98	105.98	102.98	99.88	96.68
0.224	118.23	114.00	111.00	107.00	104.00	101.00	97.80
0.224	120.36	116.02	113.02	109.02	106.02	103.02	99.82
0.224	122.49	118.04	115.04	111.04	108.04	105.04	102.04
0.224	124.62	120.06	117.06	113.06	110.06	107.06	104.06
0.224	126.75	122.08	119.08	115.08	112.08	109.08	106.08
0.224	128.88	124.10	121.10	117.10	114.10	111.10	108.10
0.224	131.01	126.12	123.12	119.12	116.12	113.12	110.12
0.224	133.14	128.14	125.14	121.14	118.14	115.14	112.14
0.224	135.27	130.16	127.16	123.16	120.16	117.16	114.16
0.224	137.40	132.18	129.18	125.18	122.18	119.18	116.18
0.224	139.53	134.20	131.20	127.20	124.20	121.20	118.20
0.224	141.66	136.22	133.22	129.22	126.22	123.22	120.22
0.224	143.79	138.24	135.24	131.24	128.24	125.24	122.24
0.224	145.92	140.26	137.26	133.26	130.26	127.26	124.26
0.224	148.05	142.28	139.28	135.28	132.28	129.28	126.28
0.224	150.18	144.30	141.30	137.30	134.30	131.30	128.30
0.224	152.31	146.32	143.32	139.32	136.32	133.32	130.32
0.224	154.44	148.34	145.34	141.34	138.34	135.34	132.34
0.224	156.57	150.36	147.36	143.36	140.36	137.36	134.36
0.224	158.70	152.38	149.38	145.38	142.38	139.38	136.38
0.224	160.83	154.40	151.40	147.40	144.40	141.40	138.40
0.224	162.96	156.42	153.42	149.42	146.42	143.42	140.42
0.224	165.09	158.44	155.44	151.44	148.44	145.44	142.44
0.224	167.22	160.46	157.46	153.46	150.46	147.46	144.46
0.224	169.35	162.48	159.48	155.48	152.48	149.48	146.48
0.224	171.48	164.50	161.50	157.50	154.50	151.50	148.50
0.224	173.61	166.52	163.52	159.52	156.52	153.52	150.52
0.224	175.74	168.54	165.54	161.54	158.54	155.54	152.54
0.224	177.87	170.56	167.56	163.56	160.56	157.56	154.56
0.224	180.00	172.58	169.58	165.58	162.58	159.58	156.58
0.224	182.13	174.60	171.60	167.60	164.60	161.60	158.60
0.224	184.26	176.62	173.62	169.62	166.62	163.62	160.62
0.224	186.39	178.64	175.64	171.64	168.64	165.64	162.64
0.224	188.52	180.66	177.66	173.66	170.66	167.66	164.66
0.224	190.65	182.68	179.68	175.68	172.68	169.68	166.68
0.224	192.78	184.70	181.70	177.70	174.70	171.70	168.70
0.224	194.91	186.72	183.72	179.72	176.72	173.72	170.72
0.224	197.04	188.74	185.74	181.74	178.74	175.74	172.74
0.224	199.17	190.76	187.76	183.76	180.76	177.76	174.76
0.224	201.30	192.78	189.78	185.78	182.78	179.78	176.78
0.224	203.43	194.80	191.80	187.80	184.80	181.80	178.80
0.224	205.56	196.82	193.82	189.82	186.82	183.82	180.82
0.224	207.69	198.84	195.84	191.84	188.84	185.84	182.84
0.224	209.82	200.86	197.86	193.86	190.86	187.86	184.86
0.224	211.95	202.88	199.88	195.88	192.88	189.88	186.88
0.224	214.08	204.90	201.90	197.90	194.90	191.90	188.90
0.224	216.21	206.92	203.92	199.92	196.92	193.92	190.92
0.224	218.34	208.94	205.94	201.94	198.94	195.94	192.94
0.224	220.47	210.96	207.96	203.96	200.96	197.96	194.96
0.224	222.60	212.98	209.98	205.98	202.98	199.98	196.98
0.224	224.73	214.99	211.99	207.99	204.99	201.99	198.99
0.224	226.86	216.99	213.99	209.99	206.99	203.99	200.99
0.224	228.99	218.99	215.99	211.99	208.99	205.99	202.99
0.224	231.12	220.99	217.99	213.99	210.99	207.99	204.99
0.224	233.25	222.99	219.99	215.99	212.99	209.99	206.99
0.224	235.38	224.99	221.99	217.99	214.99	211.99	208.99
0.224	237.51	226.99	223.99	219.99	216.99	213.99	210.99
0.224	239.64	228.99	225.99	221.99	218.99	215.99	212.99
0.224	241.77	230.99	227.99	223.99	220.99	217.99	214.99
0.224	243.90	232.99	229.99	225.99	222.99	219.99	216.99
0.224	246.03	234.99	231.99	227.99	224.99	221.99	218.99
0.224	248.16	236.99	233.99	229.99	226.99	223.99	220.99
0.224	250.29	238.99	235.99	231.99	228.99	225.99	222.99
0.224	252.42	240.99	237.99	233.99	230.99	227.99	224.99
0.224	254.55	242.99	239.99	235.99	232.99	229.99	226.99
0.224	256.68	244.99	241.99	237.99	234.99	231.99	228.99
0.224	258.81	246.99	243.99	239.99	236.99	233.99	230.99
0.224	260.94	248.99	245.99	241.99	238.99	235.99	232.99
0.224	263.07	250.99	247.99	243.99	240.99	237.99	234.99
0.224	265.20	252.99	249.99	245.99	242.99	239.99	236.99
0.224	267.33	254.99	251.99	247.99	244.99	241.99	238.99
0.224	269.46	256.99	253.99	249.99	246.99	243.99	240.99
0.224	271.59	258.99	255.99	251.99	248.99	245.99	242.99
0.224	273.72	260.99	257.99	253.99	250.99	247.99	244.99
0.224	275.85	262.99	25				

TABLE 20
Path in Meters for Flight Altitude = 80 m

TABLE 21
Side of Instantaneous Field of View in Meters
for Flight Altitude = 90 m

	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
DELTA	0.20	1.6.37	1.4.34	1.1.26	9.74	8.58	7.67
	0.22	1.7.29	1.5.57	1.2.30	9.65	8.39	7.60
	0.24	2.0.71	1.5.77	1.3.34	10.20	9.12	8.46
	0.26	2.2.89	1.6.96	1.4.36	11.12	9.84	9.56
	0.28	2.4.20	1.8.14	1.6.55	11.14	10.56	10.20
	0.30	2.6.22	1.9.23	1.7.67	11.16	11.24	11.93
	0.32	2.8.24	2.0.43	1.8.00	11.18	11.33	11.97
	0.34	3.0.26	2.1.56	1.9.57	11.20	11.55	12.21
	0.36	3.2.28	2.2.67	2.0.67	11.22	11.89	12.57
	0.38	3.4.30	2.3.78	2.1.80	11.24	12.00	13.24
	0.40	3.6.32	2.4.89	2.2.82	11.26	12.11	13.58
	0.42	3.8.34	2.5.00	2.3.85	11.28	12.22	13.74
	0.44	4.0.36	2.6.11	2.4.88	11.30	12.33	13.91
	0.46	4.2.38	2.7.22	2.5.91	11.32	12.44	14.09
	0.48	4.4.40	2.8.33	2.6.94	11.34	12.55	14.26
	0.50	4.6.42	2.9.44	2.7.97	11.36	12.67	14.43
	0.52	4.8.44	3.0.55	2.8.00	11.38	12.79	14.60
	0.54	5.0.46	3.1.66	2.9.03	11.40	12.91	14.77
	0.56	5.2.48	3.2.77	3.0.06	11.42	13.03	14.94
	0.58	5.4.50	3.3.88	3.1.09	11.44	13.15	15.11
	0.60	5.6.52	3.4.99	3.2.12	11.46	13.27	15.28
	0.62	5.8.54	3.5.10	3.3.15	11.48	13.39	15.45
	0.64	6.0.56	3.6.21	3.4.18	11.50	13.51	15.62
	0.66	6.2.58	3.7.32	3.5.21	11.52	13.63	15.79
	0.68	6.4.60	3.8.43	3.6.24	11.54	13.75	15.96
	0.70	6.6.62	3.9.54	3.7.27	11.56	13.87	16.13
	0.72	6.8.64	4.0.65	3.8.30	11.58	13.99	16.30
	0.74	7.0.66	4.1.76	3.9.33	11.60	14.11	16.47
	0.76	7.2.68	4.2.87	4.0.36	11.62	14.23	16.64
	0.78	7.4.70	4.3.98	4.1.39	11.64	14.35	16.81
	0.80	7.6.72	4.5.09	4.2.42	11.66	14.47	16.98
	0.82	7.8.74	4.6.20	4.3.45	11.68	14.59	17.15
	0.84	8.0.76	4.7.31	4.4.48	11.70	14.71	17.32
	0.86	8.2.78	4.8.42	4.5.51	11.72	14.83	17.49
	0.88	8.4.80	4.9.53	4.6.54	11.74	14.95	17.66
	0.90	8.6.82	5.0.64	4.7.57	11.76	15.07	17.83
	0.92	8.8.84	5.1.75	4.8.60	11.78	15.19	17.99
	0.94	9.0.86	5.2.86	4.9.63	11.80	15.31	18.16
	0.96	9.2.88	5.3.97	5.0.66	11.82	15.43	18.33
	0.98	9.4.90	5.4.09	5.1.69	11.84	15.55	18.50
	1.00	9.6.92	5.5.20	5.2.72	11.86	15.67	18.67

TABLE 22
Path in Meters for Flight Altitude=90 m

DELTA	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
0.22	2344.37	1910.46	1612.22	1394.60	1228.82	1098.35	992.99
0.22	2323.24	1896.46	1592.22	1387.72	1223.02	1093.57	989.20
0.22	2302.24	1882.45	1572.23	1375.03	1217.24	1087.48	985.17
0.22	2282.24	1877.45	1552.23	1367.03	1215.54	1080.88	987.74
0.22	2262.24	1872.45	1532.23	1359.03	1213.84	1074.00	980.07
0.22	2242.24	1867.45	1512.23	1351.03	1212.14	1067.00	977.07
0.22	2222.24	1862.45	1492.23	1343.03	1209.34	1059.00	974.05
0.22	2202.24	1857.45	1472.23	1335.03	1207.64	1052.00	971.03
0.22	2182.24	1852.45	1452.23	1327.03	1205.94	1045.00	968.02
0.22	2162.24	1847.45	1432.23	1319.03	1204.24	1038.00	965.00
0.22	2142.24	1842.45	1412.23	1311.03	1202.54	1031.00	962.00
0.22	2122.24	1837.45	1392.23	1303.03	1200.84	1024.00	959.00
0.22	2102.24	1832.45	1372.23	1295.03	1199.14	1017.00	956.00
0.22	2082.24	1827.45	1352.23	1287.03	1197.44	1010.00	953.00
0.22	2062.24	1822.45	1332.23	1279.03	1195.74	1003.00	950.00
0.22	2042.24	1817.45	1312.23	1271.03	1194.04	996.00	947.00
0.22	2022.24	1812.45	1292.23	1263.03	1192.34	989.00	944.00
0.22	2002.24	1807.45	1272.23	1255.03	1190.64	982.00	941.00
0.22	1982.24	1802.45	1252.23	1247.03	1189.94	975.00	938.00
0.22	1962.24	1797.45	1232.23	1239.03	1188.24	968.00	935.00
0.22	1942.24	1792.45	1212.23	1231.03	1187.54	961.00	932.00
0.22	1922.24	1787.45	1192.23	1223.03	1186.84	954.00	929.00
0.22	1902.24	1782.45	1172.23	1215.03	1186.14	947.00	926.00
0.22	1882.24	1777.45	1152.23	1207.03	1185.44	940.00	923.00
0.22	1862.24	1772.45	1132.23	1200.03	1184.74	933.00	920.00
0.22	1842.24	1767.45	1112.23	1192.03	1184.04	926.00	917.00
0.22	1822.24	1762.45	1092.23	1184.03	1183.34	919.00	916.00
0.22	1802.24	1757.45	1072.23	1176.03	1182.64	912.00	919.00
0.22	1782.24	1752.45	1052.23	1168.03	1181.94	905.00	916.00
0.22	1762.24	1747.45	1032.23	1160.03	1181.24	900.00	913.00
0.22	1742.24	1742.45	1012.23	1152.03	1180.54	895.00	910.00
0.22	1722.24	1737.45	992.23	1144.03	1180.04	890.00	907.00
0.22	1702.24	1732.45	972.23	1136.03	1179.34	885.00	904.00
0.22	1682.24	1727.45	952.23	1128.03	1178.64	880.00	901.00
0.22	1662.24	1722.45	932.23	1120.03	1178.04	875.00	898.00
0.22	1642.24	1717.45	912.23	1112.03	1177.34	870.00	895.00
0.22	1622.24	1712.45	892.23	1104.03	1176.64	865.00	892.00
0.22	1602.24	1707.45	872.23	1096.03	1176.04	860.00	889.00
0.22	1582.24	1702.45	852.23	1088.03	1175.34	855.00	886.00
0.22	1562.24	1697.45	832.23	1080.03	1174.64	850.00	883.00
0.22	1542.24	1692.45	812.23	1072.03	1174.04	845.00	880.00
0.22	1522.24	1687.45	792.23	1064.03	1173.34	840.00	877.00
0.22	1502.24	1682.45	772.23	1056.03	1172.64	835.00	874.00
0.22	1482.24	1677.45	752.23	1048.03	1172.04	830.00	871.00
0.22	1462.24	1672.45	732.23	1040.03	1171.34	825.00	868.00
0.22	1442.24	1667.45	712.23	1032.03	1170.64	820.00	865.00
0.22	1422.24	1662.45	692.23	1024.03	1170.04	815.00	862.00
0.22	1402.24	1657.45	672.23	1016.03	1169.34	810.00	859.00
0.22	1382.24	1652.45	652.23	1008.03	1168.64	805.00	856.00
0.22	1362.24	1647.45	632.23	999.03	1168.04	800.00	853.00
0.22	1342.24	1642.45	612.23	991.03	1167.34	795.00	850.00
0.22	1322.24	1637.45	592.23	983.03	1166.64	790.00	847.00
0.22	1302.24	1632.45	572.23	975.03	1166.04	785.00	844.00
0.22	1282.24	1627.45	552.23	967.03	1165.34	780.00	841.00
0.22	1262.24	1622.45	532.23	959.03	1164.64	775.00	838.00
0.22	1242.24	1617.45	512.23	951.03	1164.04	770.00	835.00
0.22	1222.24	1612.45	492.23	943.03	1163.34	765.00	832.00
0.22	1202.24	1607.45	472.23	935.03	1162.64	760.00	829.00
0.22	1182.24	1602.45	452.23	927.03	1162.04	755.00	826.00
0.22	1162.24	1597.45	432.23	919.03	1161.34	750.00	823.00
0.22	1142.24	1592.45	412.23	911.03	1160.64	745.00	820.00
0.22	1122.24	1587.45	392.23	903.03	1160.04	740.00	817.00
0.22	1102.24	1582.45	372.23	895.03	1159.34	735.00	814.00
0.22	1082.24	1577.45	352.23	887.03	1158.64	730.00	811.00
0.22	1062.24	1572.45	332.23	879.03	1158.04	725.00	808.00
0.22	1042.24	1567.45	312.23	871.03	1157.34	720.00	805.00
0.22	1022.24	1562.45	292.23	863.03	1156.64	715.00	802.00
0.22	1002.24	1557.45	272.23	855.03	1156.04	710.00	800.00
0.22	982.24	1552.45	252.23	847.03	1155.34	705.00	797.00
0.22	962.24	1547.45	232.23	839.03	1154.64	700.00	794.00
0.22	942.24	1542.45	212.23	831.03	1154.04	695.00	791.00
0.22	922.24	1537.45	192.23	823.03	1153.34	690.00	788.00
0.22	902.24	1532.45	172.23	815.03	1152.64	685.00	785.00
0.22	882.24	1527.45	152.23	807.03	1152.04	680.00	782.00
0.22	862.24	1522.45	132.23	799.03	1151.34	675.00	779.00
0.22	842.24	1517.45	112.23	791.03	1150.64	670.00	776.00
0.22	822.24	1512.45	92.23	783.03	1150.04	665.00	773.00
0.22	802.24	1507.45	72.23	775.03	1149.34	660.00	770.00
0.22	782.24	1502.45	52.23	767.03	1148.64	655.00	767.00
0.22	762.24	1497.45	32.23	759.03	1148.04	650.00	764.00
0.22	742.24	1492.45	12.23	751.03	1147.34	645.00	761.00
0.22	722.24	1487.45	-3.23	743.03	1146.64	640.00	758.00
0.22	702.24	1482.45	-13.23	735.03	1146.04	635.00	755.00
0.22	682.24	1477.45	-33.23	727.03	1145.34	630.00	752.00
0.22	662.24	1472.45	-53.23	719.03	1144.64	625.00	749.00
0.22	642.24	1467.45	-73.23	711.03	1144.04	620.00	746.00
0.22	622.24	1462.45	-93.23	703.03	1143.34	615.00	743.00
0.22	602.24	1457.45	-113.23	695.03	1142.64	610.00	740.00
0.22	582.24	1452.45	-133.23	687.03	1142.04	605.00	737.00
0.22	562.24	1447.45	-153.23	679.03	1141.34	600.00	734.00
0.22	542.24	1442.45	-173.23	671.03	1140.64	595.00	731.00
0.22	522.24	1437.45	-193.23	663.03	1140.04	590.00	728.00
0.22	502.24	1432.45	-213.23	655.03	1139.34	585.00	725.00
0.22	482.24	1427.45	-233.23	647.03	1138.64	580.00	722.00
0.22	462.24	1422.45	-253.23	639.03	1138.04	575.00	719.00
0.22	442.24	1417.45	-273.23	631.03	1137.34	570.00	716.00
0.22	422.24	1412.45	-293.23	623.03	1136.64	565.00	713.00
0.22	402.24	1407.45	-313.23	615.03	1136.04	560.00	710.00
0.22	382.24	1402.45	-333.23	607.03	1135.34	555.00	707.00
0.22	362.24	1397.45	-353.23	599.03	1134.64	550.00	704.00
0.22	342.24	1392.45	-373.23	591.03	1134.04	545.00	701.00
0.22	322.24	1387.45	-393.23	583.03	1133.34	540.00	698.00
0.22	302.24	1382.45	-413.23	575.03	1132.64	535.00	695.00
0.22	282.24	1377.45	-433.23	567.03	1132.04	530.00	692.00
0.22	262.24	1372.45	-453.23	559.03	1131.34	525.00	689.00
0.22	242.24	1367.45	-473.23	551.03	1130.64	520.00	686.00
0.22	222.24	1362.45	-493.23	543.03	1130.04	515.00	683.00
0.22	202.24	1357.45	-513.23	535.03	1129.34	510.00	680.00
0.22	182.24	1352.45	-533.23	527.03	1128.64	505.00	677.00
0.22	162.24	1347.45	-553.23	519.03	1128.04	500.00	674.00
0.22	142.24	1342.45	-573.23	511.03	1127.34	495.00	671.00
0.22	122.24	1337.45	-593.23	503.03	1126.64	490.00	668.00
0.22	102.24	1332.45	-613.23	495.03	1126.04	485.00	665.00
0.22	82.24	1327.45	-633.23	487.03	1125.34	480.00	662.00
0.22	62.24	1322.45	-653.23	479.03	1124.64	475.00	659.00
0.22	42.24	1317.45	-673.23	471.03	1124.04	470.00	656.00
0.22	22.24	1312.45	-693.23	463.03	1123.34	465.00	653.00
0							

TABLE 23
Side of Instantaneous Field of View in Meters
for Flight Altitude=100 m

DELTA	PHI = 2.00	PHI = 2.50	PHI = 3.00	PHI = 3.50	PHI = 4.00	PHI = 4.50	PHI = 5.00
0.20	18.19	14.82	12.51	10.82	9.53	8.52	7.70
0.24	19.43	16.17	13.67	11.94	10.44	9.33	8.44
0.28	20.07	17.47	14.97	12.47	10.97	9.97	9.07
0.32	20.67	18.07	15.57	13.07	11.57	10.57	9.67
0.36	21.29	18.67	16.17	13.67	12.17	11.17	10.29
0.40	21.93	19.29	16.79	14.29	12.79	11.79	10.83
0.44	22.57	19.89	17.39	14.89	13.39	12.39	11.47
0.48	23.23	20.53	18.03	15.53	14.03	12.53	11.67
0.52	23.90	21.17	18.67	16.17	14.67	13.17	12.23
0.56	24.59	21.82	19.32	16.82	15.32	13.82	12.89
0.60	25.29	22.47	19.97	17.47	15.97	14.47	13.59
0.64	25.99	23.12	20.62	18.12	16.62	15.12	14.29
0.68	26.71	23.77	21.27	18.77	17.27	15.77	14.91
0.72	27.44	24.42	21.92	19.42	17.92	16.42	15.56
0.76	28.18	25.07	22.57	20.07	18.57	17.07	16.21
0.80	28.93	25.72	23.22	20.72	19.22	17.72	16.86
0.84	29.69	26.37	23.87	21.37	19.87	18.37	17.51
0.88	30.46	27.02	24.52	21.02	19.52	18.02	17.16
0.92	31.24	27.67	25.17	21.67	20.17	18.67	17.81
0.96	32.03	28.32	25.82	22.32	20.82	19.32	18.46
1.00	32.83	28.97	26.57	23.07	21.57	20.07	19.21

TABLE 24
Path in Meters for Flight Altitude=100 m

APPENDIX C

TABLES WITH DATA USED FOR LOWTRAN 6

TABLE 25
Data Used for Month October

CARD 1

MODEL	:	2	:	Midlatitude Summer
IATYPE	:	2	:	Slant Path Between Two Altitudes
IEMSCT	:	0	:	Program Execution in Transmittance Mode
M1	:	2	:	Midlatitude Summer T and P Profile
M2	:	2	:	Midlatitude Summer Water Vapor Profile
M3	:	2	:	Midlatitude Summer Ozone Profile
IM	:	0	:	Normal Operation of Program
NOPRT	:	0	:	Normal Operation of Program
TBOUND	:	0	:	Normal Operation of Program
SALB	:	0	:	Normal Operation of Program

CARD 2

IHAZE	:	3	:	Navy Maritime Extinction
ISEASN	:	2	:	Fall - Winter
IVULCN	:	0	:	No Stratospheric Background
ICSTL	:	5	:	Medium Continental Influence
ICIR	:	0	:	No Cirrus
IVSA	:	0	:	Not Used
VIS	:	10	:	Meteorological Range (Km)
WSS	:	2.09	:	Current Wind Speed (m/s)
WHH	:	2.09	:	24 hours Average Wind Speed (m/s)
RAINRT	:	0	:	Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude		
H2	:	From Table 22 : Final Altitude		
ANGLE	:	--	:	Not Necessary to Define
RANGE	:	From Table 22 : Path Length		
BETA	:	--	:	Not Necessary to Define
RO	:	--	:	Not Necessary to Define
LEN	:	0	:	Normal Operation

CARD 4

V1	:	2255	:	Initial Frequency
V2	:	2000	:	Final Frequency
DV	:	10	:	Frequency Increment

TABLE 26
Average Transmittances for Month October

INITIAL ALTITUDE (Km)	FINAL ALTITUDE (Km)	Average Transmittance for Month October (Km)			
0.004 (Km)	0.4718	0.4715	0.4712	0.4712	0.4711
0.007 (Km)	0.4719	0.4716	0.4713	0.4713	0.4712
0.012 (Km)	0.4720	0.4717	0.4715	0.4715	0.4714
0.016 (Km)	0.4722	0.4719	0.4716	0.4716	0.4715
0.018 (Km)	0.4722	0.4719	0.4717	0.4717	0.4716
0.021 (Km)	0.4723	0.4720	0.4718	0.4718	0.4717
0.022 (Km)	0.4723	0.4720	0.4718	0.4718	0.4717
Average per Altitude	0.4721	0.4718	0.4716	0.4716	0.4715
Total Average of the Month = 0.4717					

TABLE 27
Data Used for Month November

CARD 1

MODEL	:	2	: Midlatitude Summer
IATYPE	:	2	: Slant Path Between Two Altitudes
IEMSCT	:	0	: Program Execution in Transmittance Mode
M1	:	2	: Midlatitude Summer T and P Profile
M2	:	2	: Midlatitude Summer Water Vapor Profile
M3	:	2	: Midlatitude Summer Ozone Profile
IM	:	0	: Normal Operation of Program
NOPRT	:	0	: Normal Operation of Program
TBOUND	:	0	: Normal Operation of Program
SALB	:	0	: Normal Operation of Program

CARD 2

IHAZE	:	3	: Navy Maritime Extinction
ISEASN	:	2	: Fall - Winter
IVULCN	:	0	: No Stratospheric Background
ICSTL	:	5	: Medium Continental Influence
ICIR	:	0	: No Cirrus
IVSA	:	0	: Not Used
VIS	:	10	: Meteorological Range (Km)
WSS	:	2.09	: Current Wind Speed(m/s)
WHH	:	2.09	: 24 hours Average Wind Speed (m/s)
RAINRT	:	0	: Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude	
H2	:	From Table 22 : Final Altitude	
ANGLE	:	--	: Not Necessary to Define
RANGE	:	From Table 22 : Path Length	
BETA	:	--	: Not Necessary to Define
RO	:	--	: Not Necessary to Define
LEN	:	0	: Normal Operation

CARD 4

V1	:	2255	: Initial Frequency
V2	:	2000	: Final Frequency
DV	:	10	: Frequency Increment

TABLE 28
Average Transmittances for Month November

FINAL ALTITUDE	0.050 (Km)	0.060 (Km)	0.070 (Km)	0.080 (Km)	0.090 (Km)	0.100 (Km)
INITIAL ALTITUDE						
0.004 (Km)	0.4705	0.4702	0.4699	0.4699	0.4698	0.4698
0.007 (Km)	0.4706	0.4703	0.4700	0.4700	0.4699	0.4699
0.012 (Km)	0.4708	0.4705	0.4702	0.4702	0.4701	0.4701
0.016 (Km)	0.4709	0.4706	0.4703	0.4703	0.4702	0.4702
0.018 (Km)	0.4710	0.4707	0.4704	0.4704	0.4703	0.4703
0.021 (Km)	0.4711	0.4708	0.4705	0.4705	0.4704	0.4704
0.022 (Km)	0.4711	0.4708	0.4705	0.4705	0.4704	0.4704
Average per Altitude	0.4709	0.4706	0.4703	0.4703	0.4702	0.4702

Total Average of the Month = 0.4704

TABLE 29
Data Used for Month December

CARD 1

MODEL	:	2	: Midlatitude Summer
ITYPE	:	2	: Slant Path Between Two Altitudes
IEMSCT	:	0	: Program Execution in Transmittance Mode
M1	:	2	: Midlatitude Summer T and P Profile
M2	:	2	: Midlatitude Summer Water Vapor Profile
M3	:	2	: Midlatitude Summer Ozone Profile
IM	:	0	: Normal Operation of Program
NOPRT	:	0	: Normal Operation of Program
TBOUND	:	0	: Normal Operation of Program
SALB	:	0	: Normal Operation of Program

CARD 2

IHAZE	:	3	: Navy Maritime Extinction
ISEASN	:	2	: Fall - Winter
IVULCN	:	0	: No Stratospheric Background
ICSTL	:	5	: Medium Continental Influence
ICIR	:	0	: No Cirrus
IVSA	:	0	: Not Used
VIS	:	10	: Meteorological Range
WSS	:	2.05	: Current Wind Speed
WHH	:	2.05	: 24 hours Average Wind Speed (m/s)
RAINRT	:	0	: Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude	
H2	:	From Table 22 : Final Altitude	
ANGLE	:	--	: Not Necessary to Define
RANGE	:	From Table 22 : Path Length	
BETA	:	--	: Not Necessary to Define
RO	:	--	: Not Necessary to Define
LEN	:	0	: Normal Operation

CARD 4

V1	:	2255	: Initial Frequency
V2	:	2000	: Final Frequency
DV	:	10	: Frequency Increment

TABLE 30
Average Transmittances for Month December

FINAL ALTITUDE	0.050 (Km)	0.060 (Km)	0.070 (Km)	0.080 (Km)	0.090 (Km)	0.100 (Km)
INITIAL ALTITUDE						
0.014 (Km)	0.4705	0.4702	0.4699	0.4699	0.4698	0.4698
0.007 (Km)	0.4706	0.4703	0.4700	0.4700	0.4699	0.4699
0.012 (Km)	0.4708	0.4705	0.4702	0.4702	0.4701	0.4701
0.016 (Km)	0.4709	0.4706	0.4703	0.4703	0.4702	0.4702
0.018 (Km)	0.4710	0.4707	0.4704	0.4704	0.4703	0.4703
0.021 (Km)	0.4711	0.4708	0.4705	0.4705	0.4704	0.4704
0.022 (Km)	0.4711	0.4708	0.4705	0.4705	0.4704	0.4704
Average per Altitude	0.4709	0.4706	0.4703	0.4703	0.4702	0.4702

Total Average of the Month = 0.4704

TABLE 31
Data Used for Month January

CARD 1

MODEL	:	2	:	Midlatitude Summer
ITYPE	:	2	:	Slant Path Between Two Altitudes
IEMSCT	:	0	:	Program Execution in Transmittance Mode
M1	:	2	:	Midlatitude Summer T and P Profile
M2	:	2	:	Midlatitude Summer Water Vapor Profile
M3	:	2	:	Midlatitude Summer Ozone Profile
IM	:	0	:	Normal Operation of Program
NOPRT	:	0	:	Normal Operation of Program
TBOUND	:	0	:	Normal Operation of Program
SALB	:	0	:	Normal Operation of Program

CARD 2

IHAZE	:	3	:	Navy Maritime Extinction
ISEASN	:	2	:	Fall - Winter
IVULCN	:	0	:	No Stratospheric Background
ICSTL	:	5	:	Medium Continental Influence
ICIR	:	0	:	No Cirrus
IVSA	:	0	:	Not Used
VIS	:	15	:	Meteorological Range
WSS	:	2.06	:	Current Wind Speed
WHH	:	2.06	:	24 hours Average Wind Speed (m/s)
RAINRT	:	0	:	Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude		
H2	:	From Table 22 : Final Altitude		
ANGLE	:	--	:	Not Necessary to Define
RANGE	:	From Table 22 : Path Length		
BETA	:	--	:	Not Necessary to Define
RO	:	--	:	Not Necessary to Define
LEN	:	0	:	Normal Operation

CARD 4

V1	:	2255	:	Initial Frequency
V2	:	2000	:	Final Frequency
DV	:	10	:	Frequency Increment

TABLE 32

Average Transmittances for Month January

INITIAL ALTITUDE (Km)	FINAL ALTITUDE 0.050 (Km)	0.060 (Km)	0.070 (Km)	0.080 (Km)	0.090 (Km)	0.100 (Km)
0.004 (Km)	0.4705	0.4702	0.4699	0.4699	0.4698	0.4698
0.007 (Km)	0.4706	0.4703	0.4700	0.4700	0.4699	0.4699
0.012 (Km)	0.4708	0.4705	0.4702	0.4702	0.4701	0.4701
0.016 (Km)	0.4709	0.4706	0.4703	0.4703	0.4702	0.4702
0.018 (Km)	0.4710	0.4707	0.4704	0.4704	0.4703	0.4703
0.021 (Km)	0.4711	0.4708	0.4705	0.4705	0.4704	0.4704
0.022 (Km)	0.4711	0.4708	0.4705	0.4705	0.4704	0.4704
Average per Altitude	0.4709	0.4706	0.4703	0.4703	0.4702	0.4702

Total Average of the Month = 0.4704

TABLE 33
Data Used for Month February

CARD 1

MODEL	:	2	: Midlatitude Summer
ITYPE	:	2	: Slant Path Between Two Altitudes
IEMSCT	:	0	: Program Execution in Transmittance Mode
M1	:	2	: Midlatitude Summer T and P Profile
M2	:	2	: Midlatitude Summer Water Vapor Profile
M3	:	2	: Midlatitude Summer Ozone Profile
IM	:	0	: Normal Operation of Program
NOPRT	:	0	: Normal Operation of Program
TBOUND	:	0	: Normal Operation of Program
SALB	:	0	: Normal Operation of Program

CARD 2

IHAZE	:	3	: Navy Maritime Extinction
ISEASN	:	2	: Fall - Winter
IVULCN	:	0	: No Stratospheric Background
ICSTL	:	5	: Medium Continental Influence
ICIR	:	0	: No Cirrus
IVSA	:	0	: Not Used
VIS	:	15	: Meteorological Range (Km)
WSS	:	2.22	: Current Wind Speed (m/s)
WHH	:	2.22	: 24 hours Average Wind Speed (m/s)
RAINRT	:	0	: Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude	
H2	:	From Table 22 : Final Altitude	
ANGLE	:	--	: Not Necessary to Define
RANGE	:	From Table 22 : Path Length	
BETA	:	--	: Not Necessary to Define
RO	:	--	: Not Necessary to Define
LEN	:	0	: Normal Operation

CARD 4

V1	:	2255	: Initial Frequency
V2	:	2000	: Final Frequency
DV	:	10	: Frequency Increment

TABLE 34
Average Transmittances for Month February

FINAL ALTITUDE (Km)	0 . 0 5 0	0 . 0 6 0	0 . 0 7 0	0 . 0 8 0	0 . 0 9 0	0 . 1 0 0
INITIAL ALTITUDE (Km)				(Km)		(Km)
0 . 0 0 4 (Km)	0 . 4 7 1 1	0 . 4 7 0 8	0 . 4 7 0 5	0 . 4 7 0 4	0 . 4 7 0 4	0 . 4 7 0 4
0 . 0 0 7 (Km)	0 . 4 7 1 2	0 . 4 7 0 9	0 . 4 7 0 6	0 . 4 7 0 5	0 . 4 7 0 5	0 . 4 7 0 5
0 . 0 1 2 (Km)	0 . 4 7 1 4	0 . 4 7 1 1	0 . 4 7 0 8	0 . 4 7 0 7	0 . 4 7 0 7	0 . 4 7 0 7
0 . 0 1 6 (Km)	0 . 4 7 1 5	0 . 4 7 1 2	0 . 4 7 0 9	0 . 4 7 0 8	0 . 4 7 0 8	0 . 4 7 0 8
0 . 0 1 8 (Km)	0 . 4 7 1 6	0 . 4 7 1 3	0 . 4 7 1 0	0 . 4 7 0 9	0 . 4 7 0 9	0 . 4 7 0 9
0 . 0 2 1 (Km)	0 . 4 7 1 7	0 . 4 7 1 4	0 . 4 7 1 2	0 . 4 7 1 0	0 . 4 7 1 0	0 . 4 7 1 0
0 . 0 2 2 (Km)	0 . 4 7 1 7	0 . 4 7 1 4	0 . 4 7 1 1	0 . 4 7 1 0	0 . 4 7 1 0	0 . 4 7 1 0
Average per Altitude	0 . 4 7 1 5	0 . 4 7 1 2	0 . 4 7 0 9	0 . 4 7 0 8	0 . 4 7 0 8	0 . 4 7 0 8

Total Average of the Month = 0 . 4 7 1 0

TABLE 35
Data Used for Month March

CARD 1

MODEL	:	2	: Midlatitude Summer
IATYPE	:	2	: Slant Path Between Two Altitudes
IEMSCT	:	0	: Program Execution in Transmittance Mode
M1	:	2	: Midlatitude Summer T and P Profile
M2	:	2	: Midlatitude Summer Water Vapor Profile
M3	:	2	: Midlatitude Summer Ozone Profile
IM	:	0	: Normal Operation of Program
NOPRT	:	0	: Normal Operation of Program
TBOUND	:	0	: Normal Operation of Program
SALB	:	0	: Normal Operation of Program

CARD 2

IHAZE	:	3	: Navy Maritime Extinction
ISEASN	:	2	: Fall - Winter
IVULCN	:	0	: No Stratospheric Background
ICSTL	:	5	: Medium Continental Influence
ICIR	:	0	: No Cirrus
IVSA	:	0	: Not Used
VIS	:	15	: Meteorological Range (Km)
WSS	:	2.13	: Current Wind Speed (m/s)
WHH	:	2.13	: 24 hours Average Wind Speed (m/s)
RAINRT	:	0	: Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude	
H2	:	From Table 22 : Final Altitude	
ANGLE	:	--	: Not Necessary to Define
RANGE	:	From Table 22 : Path Length	
BETA	:	--	: Not Necessary to Define
RO	:	--	: Not Necessary to Define
LEN	:	0	: Normal Operation

CARD 4

V1	:	2255	: Initial Frequency
V2	:	2000	: Final Frequency
DV	:	10	: Frequency Increment

TABLE 36
Average Transmittances for Month March

FINAL ALTITUDE	0.050 (Km)	0.060 (Km)	0.070 (Km)	0.080 (Km)	0.090 (Km)	0.100 (Km)
INITIAL ALTITUDE						
0.004 (Km)	0.4718	0.4713	0.4712	0.4711	0.4711	0.4711
0.007 (Km)	0.4719	0.4714	0.4713	0.4712	0.4712	0.4712
0.012 (Km)	0.4720	0.4716	0.4715	0.4714	0.4714	0.4714
0.016 (Km)	0.4722	0.4717	0.4716	0.4715	0.4715	0.4715
0.018 (Km)	0.4722	0.4718	0.4717	0.4716	0.4716	0.4716
0.021 (Km)	0.4723	0.4719	0.4718	0.4717	0.4717	0.4717
0.022 (Km)	0.4723	0.4719	0.4718	0.4717	0.4717	0.4717
Average per Altitude	0.4721	0.4717	0.4716	0.4715	0.4715	0.4715

Total Average of the Month = 0.4713

TABLE 37
Data Used for Month April

CARD 1

MODEL	:	2	: Midlatitude Summer
ITYPE	:	2	: Slant Path Between Two Altitudes
IEMSCT	:	0	: Program Execution in Transmittance Mode
M1	:	2	: Midlatitude Summer T and P Profile
M2	:	2	: Midlatitude Summer Water Vapor Profile
M3	:	2	: Midlatitude Summer Ozone Profile
IM	:	0	: Normal Operation of Program
NOPRT	:	0	: Normal Operation of Program
TBOUND	:	0	: Normal Operation of Program
SALB	:	0	: Normal Operation of Program

CARD 2

IHAZE	:	3	: Navy Maritime Extinction
ISEASN	:	2	: Fall - Winter
IVULCN	:	0	: No Stratospheric Background
ICSTL	:	5	: Medium Continental Influence
ICIR	:	0	: No Cirrus
IVSA	:	0	: Not Used
VIS	:	15	: Meteorological Range (Km)
WSS	:	2.12	: Current Wind Speed (m/s)
WHH	:	2.12	: 24 hours Average Wind Speed (m/s)
RAINRT	:	0	: Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude	
H2	:	From Table 22 : Final Altitude	
ANGLE	:	--	: Not Necessary to Define
RANGE	:	From Table 22 : Path Length	
BETA	:	--	: Not Necessary to Define
RO	:	--	: Not Necessary to Define
LEN	:	0	: Normal Operation

CARD 4

V1	:	2255	: Initial Frequency
V2	:	2000	: Final Frequency
DV	:	10	: Frequency Increment

TABLE 38
Average Transmittances for Month April

FINAL ALTITUDE (Km)	0.050 (Km)	0.060 (Km)	0.070 (Km)	0.080 (Km)	0.090 (Km)	0.100 (Km)
INITIAL ALTITUDE (Km)						
0.004 (Km)	0.4718	0.4713	0.4712	0.4711	0.4711	0.4711
0.007 (Km)	0.4719	0.4714	0.4713	0.4712	0.4712	0.4712
0.012 (Km)	0.4720	0.4716	0.4715	0.4714	0.4714	0.4714
0.016 (Km)	0.4722	0.4717	0.4716	0.4715	0.4715	0.4715
0.018 (Km)	0.4722	0.4718	0.4717	0.4716	0.4716	0.4716
0.021 (Km)	0.4723	0.4719	0.4718	0.4717	0.4717	0.4717
0.022 (Km)	0.4723	0.4719	0.4718	0.4717	0.4717	0.4717
Average per Altitude	0.4721	0.4717	0.4716	0.4715	0.4715	0.4715
Total Average of the Month						0.4713

TABLE 39
Data Used for Month May

CARD 1

MODEL	:	1	:	Tropical Atmosphere
IATYPE	:	2	:	Slant Path Between Two Altitudes
IEMSCT	:	0	:	Program Execution in Transmittance Mode
M1	:	1	:	Tropical Temperature and Pressure Profile
M2	:	1	:	Tropical Water Vapor Profile
M3	:	1	:	Tropical Ozone Profile
IM	:	0	:	Normal Operation of Program
NOPRT	:	0	:	Normal Operation of Program
TBOUND	:	0	:	Normal Operation of Program
SALB	:	0	:	Normal Operation of Program

CARD 2

IHAZE	:	3	:	Navy Maritime Extinction
ISEASN	:	1	:	Spring - Summer
IVULCN	:	0	:	No Stratospheric Background
ICSTL	:	5	:	Medium Continental Influence
ICIR	:	0	:	No Cirrus
IVSA	:	0	:	Not Used
VIS	:	15	:	Meteorological Range (Km)
WSS	:	2.02	:	Current Wind Speed (m/s)
WHH	:	2.02	:	24 hours Average Wind Speed (m/s)
RAINRT	:	0	:	Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude		
H2	:	From Table 22 : Final Altitude		
ANGLE	:	-- : Not Necessary to Define		
RANGE	:	From Table 22 : Path Length		
BETA	:	-- : Not Necessary to Define		
RO	:	-- : Not Necessary to Define		
LEN	:	0	:	Normal Operation

CARD 4

V1	:	2255	:	Initial Frequency
V2	:	2000	:	Final Frequency
DV	:	10	:	Frequency Increment

TABLE 40
Average Transmittances for Month May

FINAL ALTITUDE (Km)	0.050	0.060 (Km)	0.070 (Km)	0.080 (Km)	0.090 (Km)	0.100 (Km)
INITIAL ALTITUDE (Km)						
0.004 (Km)	0.4270	0.4267	0.4264	0.4262	0.4263	0.4263
0.007 (Km)	0.4271	0.4268	0.4265	0.4263	0.4264	0.4264
0.012 (Km)	0.4273	0.4270	0.4267	0.4265	0.4266	0.4266
0.016 (Km)	0.4274	0.4271	0.4268	0.4267	0.4268	0.4268
0.018 (Km)	0.4275	0.4272	0.4269	0.4268	0.4269	0.4269
0.021 (Km)	0.4276	0.4273	0.4270	0.4269	0.4270	0.270
0.022 (Km)	0.4276	0.4273	0.4270	0.4269	0.4270	0.4270
Average per Altitude	0.4273	0.4271	0.4268	0.4266	0.4267	0.4267
Total Average of the Month = 0.4269						

TABLE 41
Data Used for Month June

CARD 1

MODEL	:	1	:	Tropical Atmosphere
I TYPE	:	2	:	Slant Path Between Two Altitudes
I EMSCT	:	0	:	Program Execution in Transmittance Mode
M1	:	1	:	Tropical Temperature and Pressure Profile
M2	:	1	:	Tropical Water Vapor Profile
M3	:	1	:	Tropical Ozone Profile
IM	:	0	:	Normal Operation of Program
NOPRT	:	0	:	Normal Operation of Program
TBOUND	:	0	:	Normal Operation of Program
SALB	:	0	:	Normal Operation of Program

CARD 2

I HAZE	:	3	:	Navy Maritime Extinction
I SEASN	:	1	:	Spring - Summer
I VULCN	:	0	:	No Stratospheric Background
I C STL	:	5	:	Medium Continental Influence
I CIR	:	0	:	No Cirrus
I VSA	:	0	:	Not Used
VIS	:	12	:	Meteorological Range (Km)
WSS	:	2.15	:	Current Wind Speed (m/s)
WHH	:	2.15	:	24 hours Average Wind Speed (m/s)
RAINRT	:	0	:	Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude		
H2	:	From Table 22 : Final Altitude		
ANGLE	:	--	:	Not Necessary to Define
RANGE	:	From Table 22 : Path Length		
BETA	:	--	:	Not Necessary to Define
RO	:	--	:	Not Necessary to Define
LEN	:	0	:	Normal Operation

CARD 4

V1	:	2255	:	Initial Frequency
V2	:	2000	:	Final Frequency
DV	:	10	:	Frequency Increment

TABLE 42
Average Transmittances for Month June

FINAL ALTITUDE	Average Transmittances for Month June			0.100 (Km)
	0.050 (Km)	0.060 (Km)	0.070 (Km)	
0.004 (Km)	0.4264	0.4261	0.4258	0.4257
0.007 (Km)	0.4265	0.4262	0.4259	0.4258
0.012 (Km)	0.4267	0.4264	0.4261	0.4260
0.016 (Km)	0.4268	0.4265	0.4262	0.4262
0.018 (Km)	0.4269	0.4266	0.4263	0.4262
0.021 (Km)	0.4270	0.4267	0.4264	0.4263
0.022 (Km)	0.4270	0.4267	0.4264	0.4263
Average per Altitude	0.4267	0.4264	0.4261	0.4260

Total Average of the Month = 0.4262

TABLE 43
Data Used for Month July

CARD 1

MODEL	:	1	:	Tropical Atmosphere
I TYPE	:	2	:	Slant Path Between Two Altitudes
I EMSCT	:	0	:	Program Execution in Transmittance Mode
M1	:	1	:	Tropical Temperature and Pressure Profile
M2	:	1	:	Tropical Water Vapor Profile
M3	:	1	:	Tropical Ozone Profile
IM	:	0	:	Normal Operation of Program
NOPRT	:	0	:	Normal Operation of Program
TBOUND	:	0	:	Normal Operation of Program
SALB	:	0	:	Normal Operation of Program

CARD 2

I HAZE	:	3	:	Navy Maritime Extinction
I SEASN	:	1	:	Spring - Summer
I VULCN	:	0	:	No Stratospheric Background
I C STL	:	5	:	Medium Continental Influence
I CIR	:	0	:	No Cirrus
I VSA	:	0	:	Not Used
VIS	:	12	:	Meteorological Range (Km)
WSS	:	2.15	:	Current Wind Speed (m/s)
WHH	:	2.15	:	24 hours Average Wind Speed (m/s)
RAINRT	:	0	:	Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude		
H2	:	From Table 22 : Final Altitude		
ANGLE	:	--	:	Not Necessary to Define
RANGE	:	From Table 22 : Path Length		
BETA	:	--	:	Not Necessary to Define
RO	:	--	:	Not Necessary to Define
LEN	:	0	:	Normal Operation

CARD 4

V1	:	2255	:	Initial Frequency
V2	:	2000	:	Final Frequency
DV	:	10	:	Frequency Increment

TABLE 44
Average Transmittances for Month July

FINAL ALTITUDE	0.050	0.060	0.070	0.080	0.090	0.100
INITIAL ALTITUDE	(Km.)	(Km.)	(Km.)	(Km.)	(Km.)	(Km.)
0.004 (Km.)	0.4264	0.4261	0.4258	0.4257	0.4257	0.4257
0.007 (Km.)	0.4265	0.4262	0.4259	0.4258	0.4258	0.4258
0.012 (Km.)	0.4267	0.4264	0.4261	0.4260	0.4260	0.4260
0.016 (Km.)	0.4268	0.4265	0.4262	0.4262	0.4262	0.4262
0.018 (Km.)	0.4269	0.4266	0.4263	0.4262	0.4262	0.4262
0.021 (Km.)	0.4270	0.4267	0.4264	0.4263	0.4263	0.4263
0.022 (Km.)	0.4270	0.4267	0.4264	0.4263	0.4263	0.4263
Average per Altitude	0.4267	0.4264	0.4261	0.4260	0.4260	0.4260

Total Average of the Month = 0.4262

TABLE 45
Data Used for Month August

CARD 1

MODEL	:	1	:	Tropical Atmosphere
IATYPE	:	2	:	Slant Path Between Two Altitudes
IEMSCT	:	0	:	Program Execution in Transmittance Mode
M1	:	1	:	Tropical Temperature and Pressure Profile
M2	:	1	:	Tropical Water Vapor Profile
M3	:	1	:	Tropical Ozone Profile
IM	:	0	:	Normal Operation of Program
NOPRT	:	0	:	Normal Operation of Program
TBOUND	:	0	:	Normal Operation of Program
SALB	:	0	:	Normal Operation of Program

CARD 2

IHAZE	:	3	:	Navy Maritime Extinction
ISEASN	:	1	:	Spring - Summer
IVULCN	:	0	:	No Stratospheric Background
ICSTL	:	5	:	Medium Continental Influence
ICIR	:	0	:	No Cirrus
IVSA	:	0	:	Not Used
VIS	:	10	:	Meteorological Range (Km)
WSS	:	2.20	:	Current Wind Speed (m/s)
WHH	:	2.20	:	24 hours Average Wind Speed (m/s)
RAINRT	:	0	:	Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II ; Initial Altitude		
H2	:	From Table 22 : Final Altitude		
ANGLE	:	--	:	Not Necessary to Define
RANGE	:	From Table 22 : Path Length		
BETA	:	--	:	Not Necessary to Define
RO	:	--	:	Not Necessary to Define
LEN	:	0	:	Normal Operation

CARD 4

V1	:	2255	:	Initial Frequency
V2	:	2000	:	Final Frequency
DV	:	10	:	Frequency Increment

TABLE 46
Average Transmittances for Month August

INITIAL ALTITUDE (Km)	FINAL ALTITUDE (Km)	0.050			0.060			0.070			0.080			0.090			0.100		
		(Km)																	
0.004 (Km)	0.4258	0.4255	0.4252	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251	0.4251		
0.007 (Km)	0.4259	0.4256	0.4253	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252	0.4252		
0.012 (Km)	0.4261	0.4258	0.4255	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253	0.4253		
0.016 (Km)	0.4262	0.4259	0.4256	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255	0.4255		
0.018 (Km)	0.4263	0.4260	0.4257	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256	0.4256		
0.021 (Km)	0.4264	0.4261	0.4258	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257		
0.022 (Km)	0.4264	0.4261	0.4258	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257	0.4257		
Average per Altitude	0.4261	0.4258	0.4255	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254	0.4254		
Total Average of the Month																			

TABLE 47
Data Used for Month September

CARD 1

MODEL	:	1	:	Tropical Atmosphere
I TYPE	:	2	:	Slant Path Between Two Altitudes
I EMSCT	:	0	:	Program Execution in Transmittance Mode
M1	:	1	:	Tropical Temperature and Pressure Profile
M2	:	1	:	Tropical Water Vapor Profile
M3	:	1	:	Tropical Ozone Profile
IM	:	0	:	Normal Operation of Program
NOPRT	:	0	:	Normal Operation of Program
TBOUND	:	0	:	Normal Operation of Program
SALB	:	0	:	Normal Operation of Program

CARD 2

I HAZE	:	3	:	Navy Maritime Extinction
I SEASN	:	1	:	Spring - Summer
I VULCN	:	0	:	No Stratospheric Background
I C STL	:	5	:	Medium Continental Influence
I CIR	:	0	:	No Cirrus
I VSA	:	0	:	Not Used
VIS	:	8	:	Meteorological Range (Km)
WSS	:	2.17	:	Current Wind Speed (m/s)
WHH	:	2.17	:	24 hours Average Wind Speed (m/s)
RAINRT	:	0	:	Rain Rate (mm/h)

CARD 3

H1	:	From Section G of Chapter II : Initial Altitude		
H2	:	From Table 22 : Final Altitude		
ANGLE	:	--	:	Not Necessary to Define
RANGE	:	From Table 22 : Path Length		
BETA	:	--	:	Not Necessary to Define
RO	:	--	:	Not Necessary to Define
LEN	:	0	:	Normal Operation

CARD 4

V1	:	2255	:	Initial Frequency
V2	:	2000	:	Final Frequency
DV	:	10	:	Frequency Increment

TABLE 48
Average Transmittances for Month September

INITIAL ALTITUDE (Km)	FINAL ALTITUDE (Km)			INITIAL ALTITUDE (Km)	FINAL ALTITUDE (Km)		
	0.050	0.060	0.070		0.080	0.090	0.100
0.004 (Km)	0.4249	0.4246	0.4243		0.4242	0.4242	0.4242
0.007 (Km)	0.4250	0.4247	0.4244		0.4243	0.4243	0.4243
0.012 (Km)	0.4252	0.4249	0.4246		0.4245	0.4245	0.4245
0.016 (Km)	0.4253	0.4250	0.4247		0.4246	0.4246	0.4246
0.018 (Km)	0.4254	0.4251	0.4248		0.4247	0.4247	0.4247
0.021 (Km)	0.4255	0.4252	0.4249		0.4248	0.4248	0.4248
0.022 (Km)	0.4255	0.4252	0.4249		0.4248	0.4248	0.4248
Average per Altitude	0.4252	0.4249	0.4246		0.4245	0.4245	0.4245

Total Average of the Month = 0.4247

APPENDIX D
PLOTS OF S/N RATIO

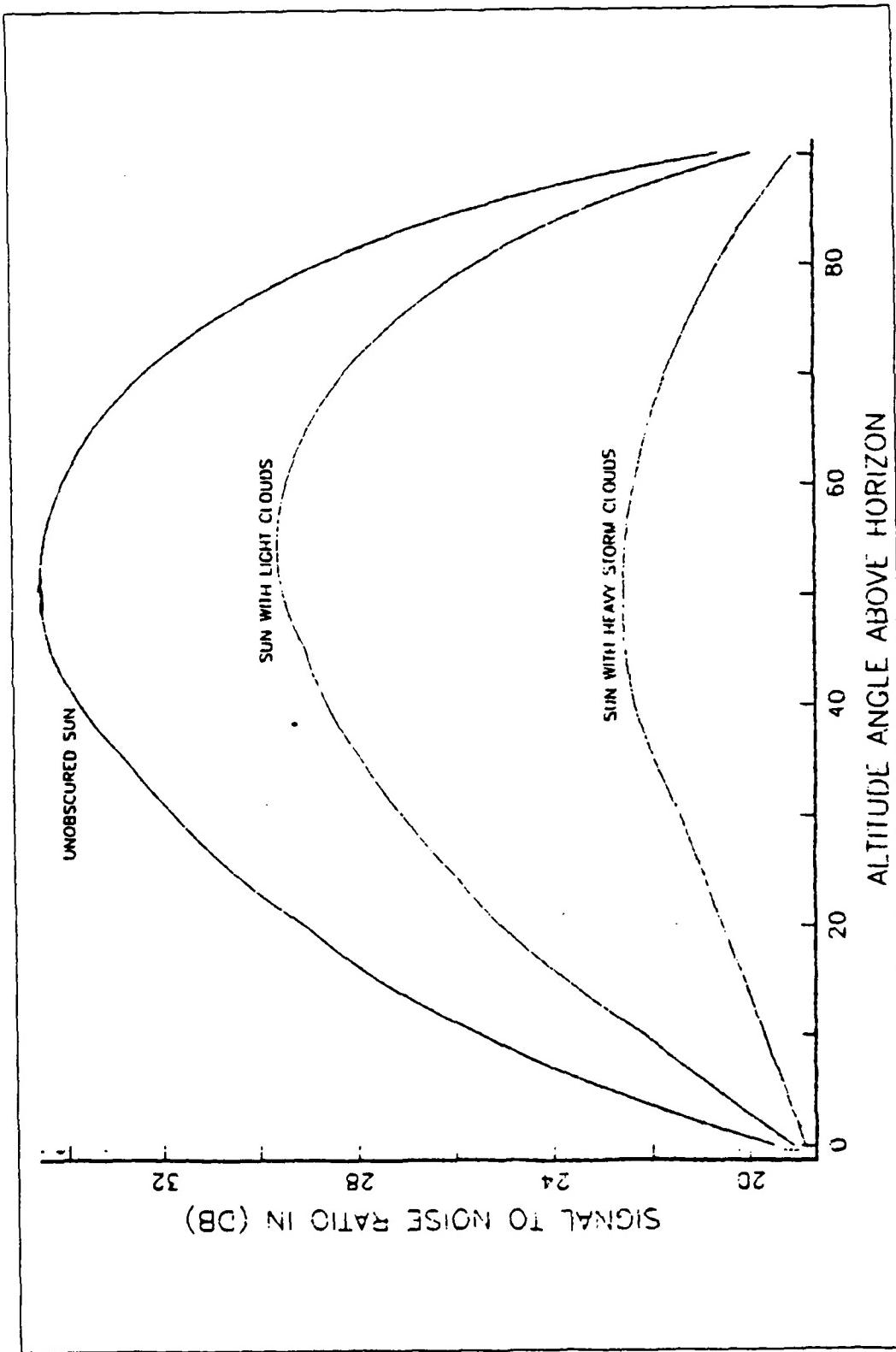


Figure D.1 S/N Ratio for Midlatitude Summer, $\epsilon_0=0.80$
and Flight Altitude 50m

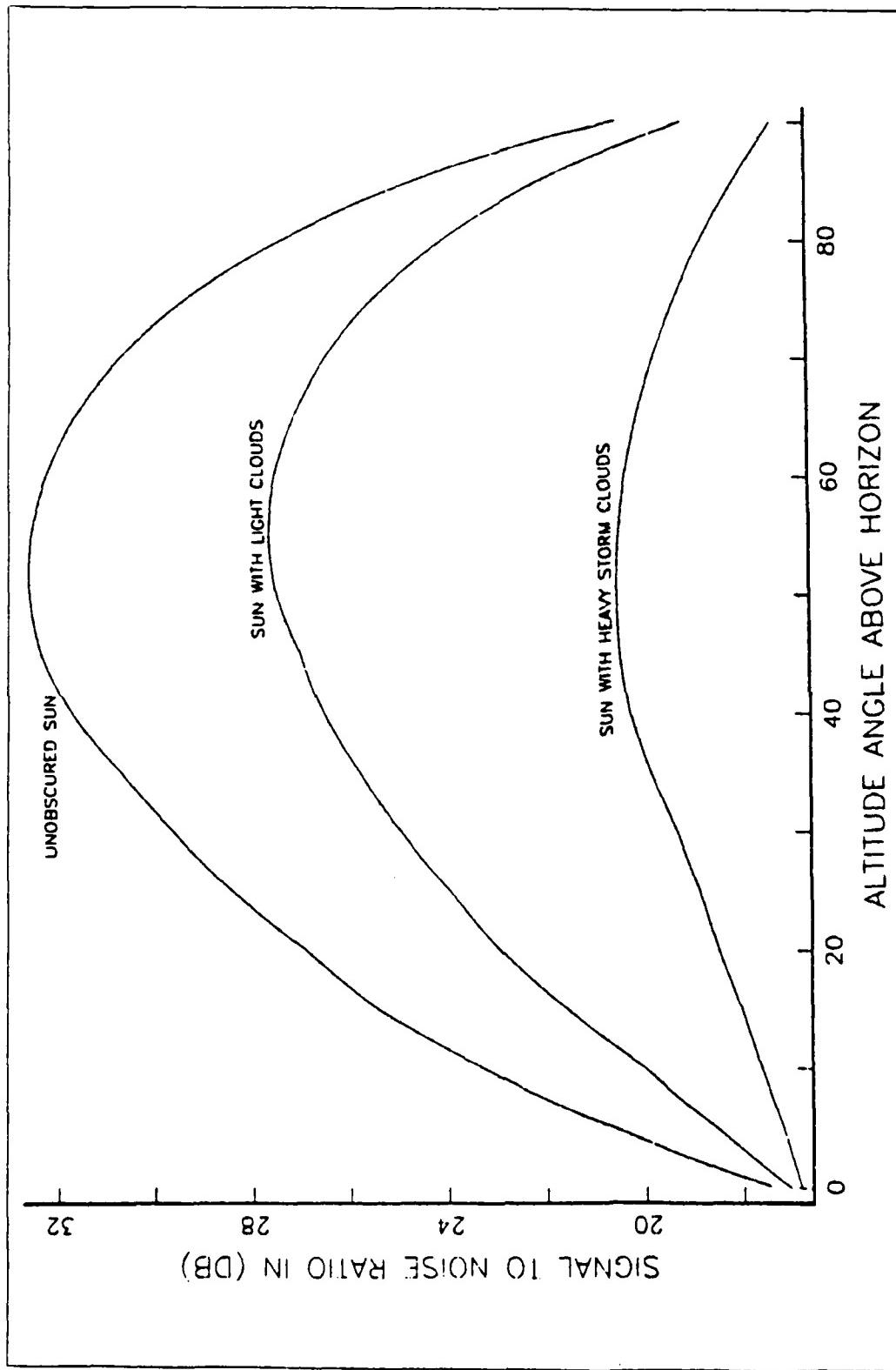


Figure D.2 S/N Ratio for Midlatitude Summer, $\epsilon_s = 0.80$ and Flight Altitude 100m

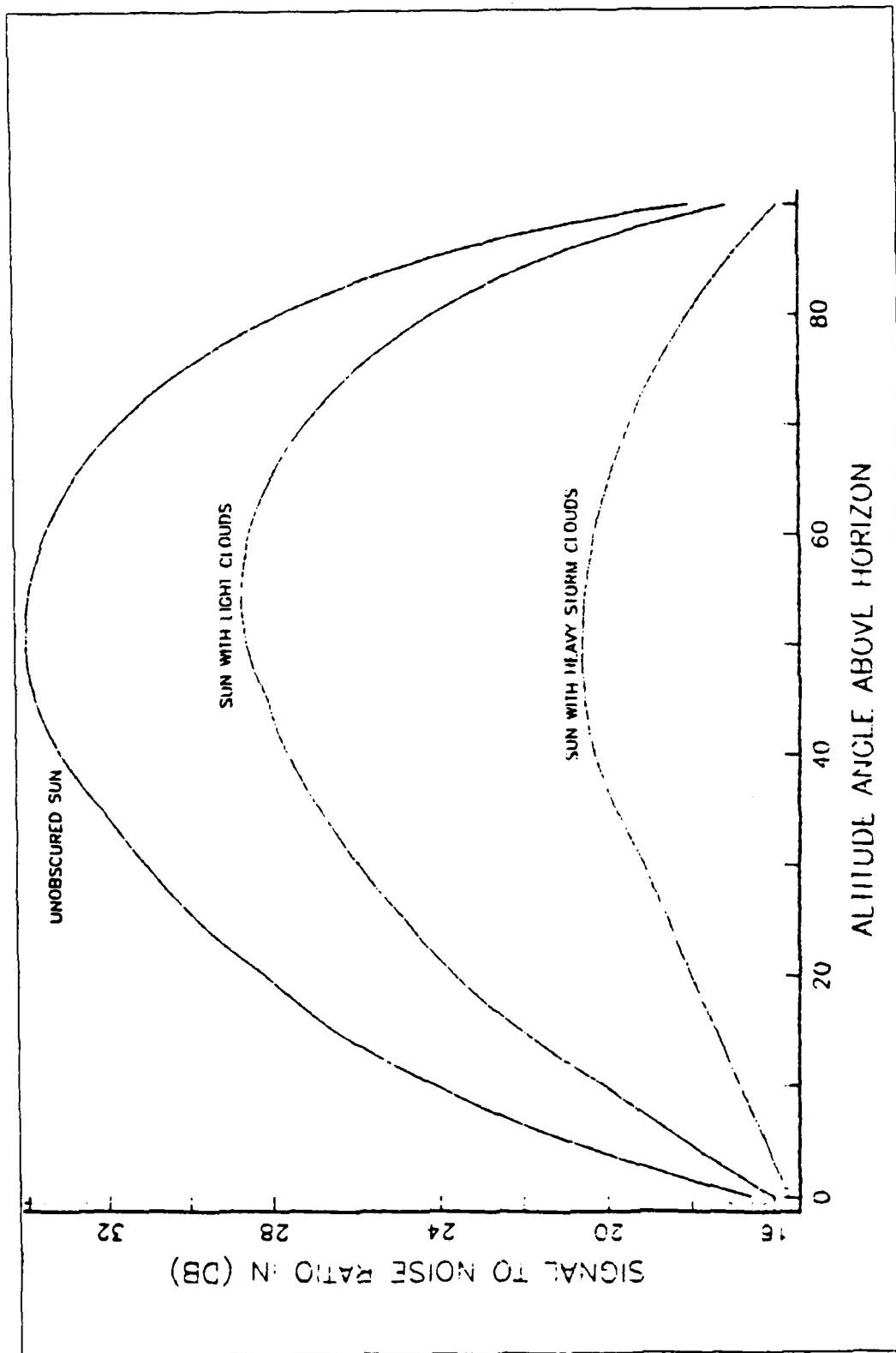


Figure D.3 S/N Ratio for Midlatitude Summer, $\epsilon_0=0.55$ and Flight Altitude 50m

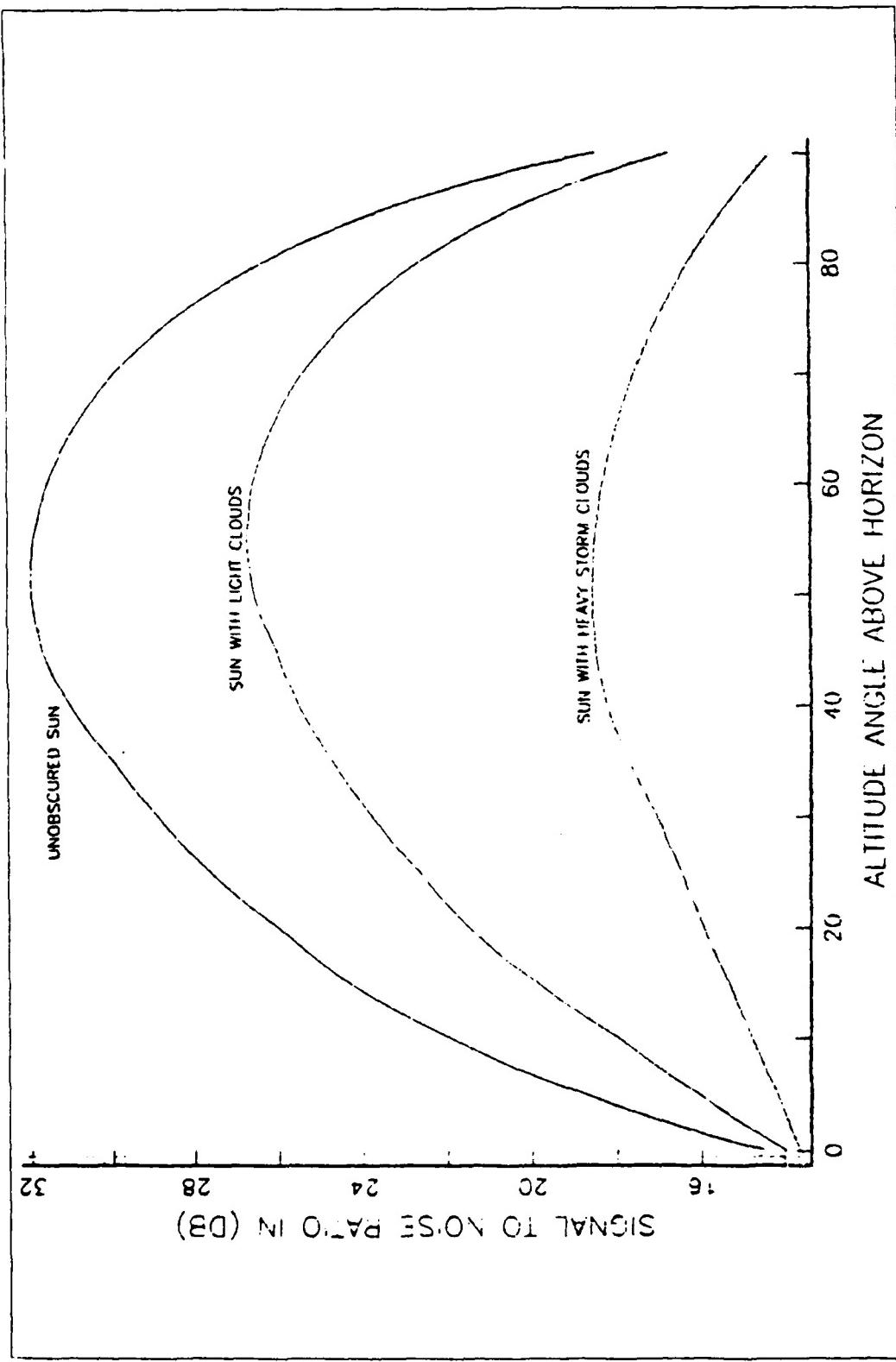


Figure D.4 S/N Ratio for Midlatitude Summer, $\delta = 0.55$
and Flight Altitude 100m

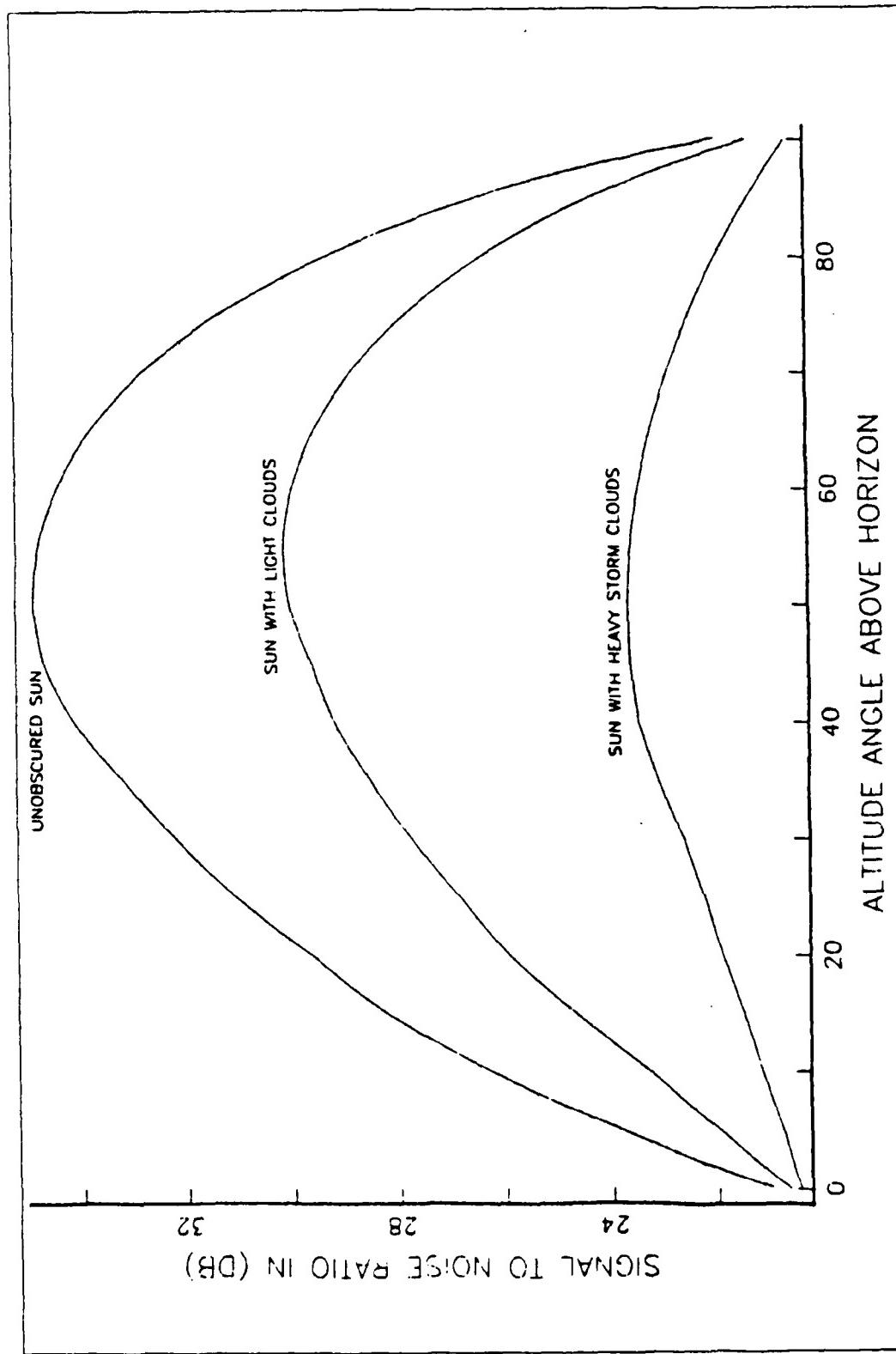


Figure D.5 S/N Ratio for Tropical, $\epsilon_0 = 0.80$ and Flight Altitude 50m

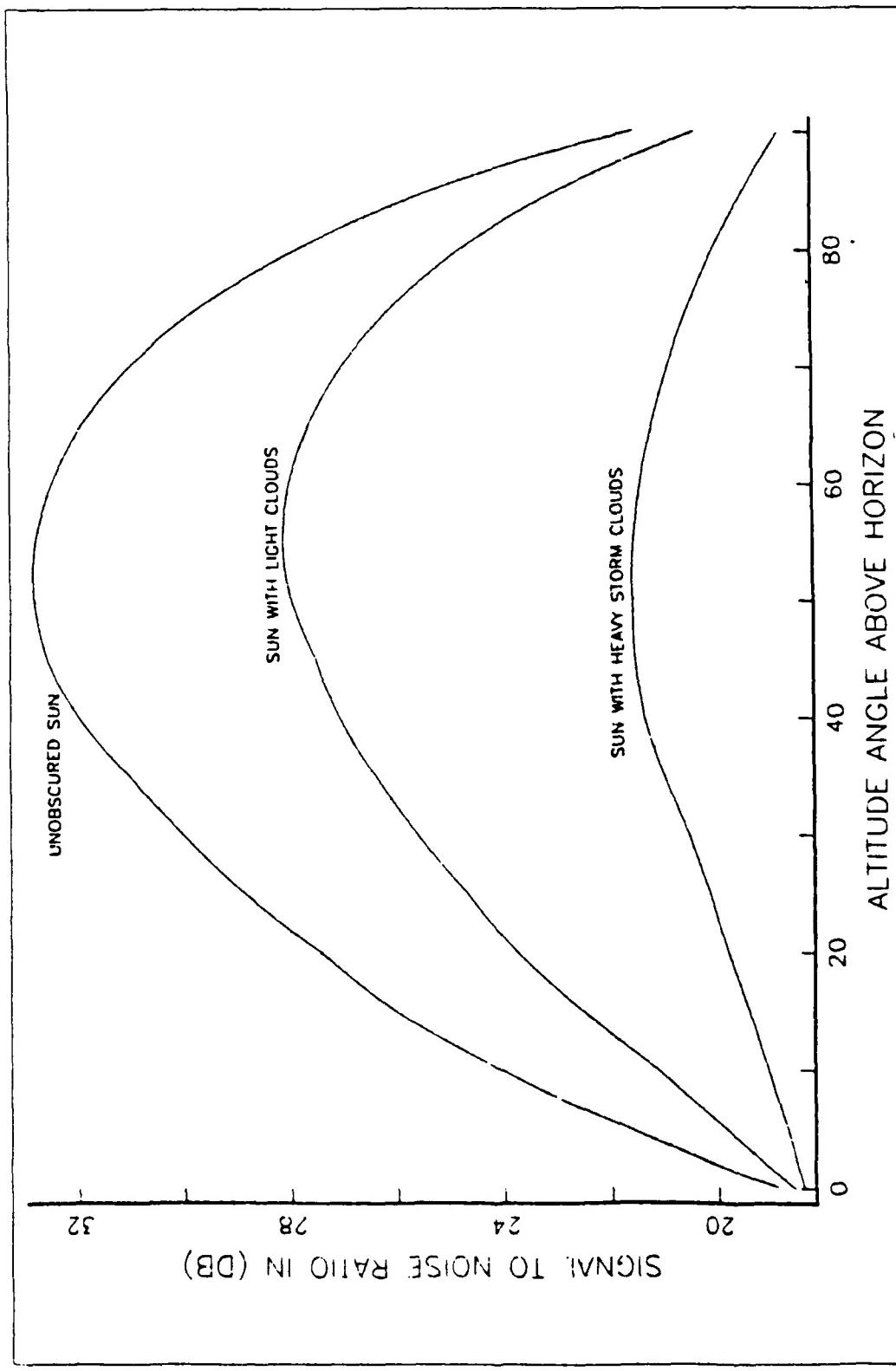


Figure D.6 S/N Ratio for Tropical, $\theta_0=0.80$ and Flight Altitude 100m

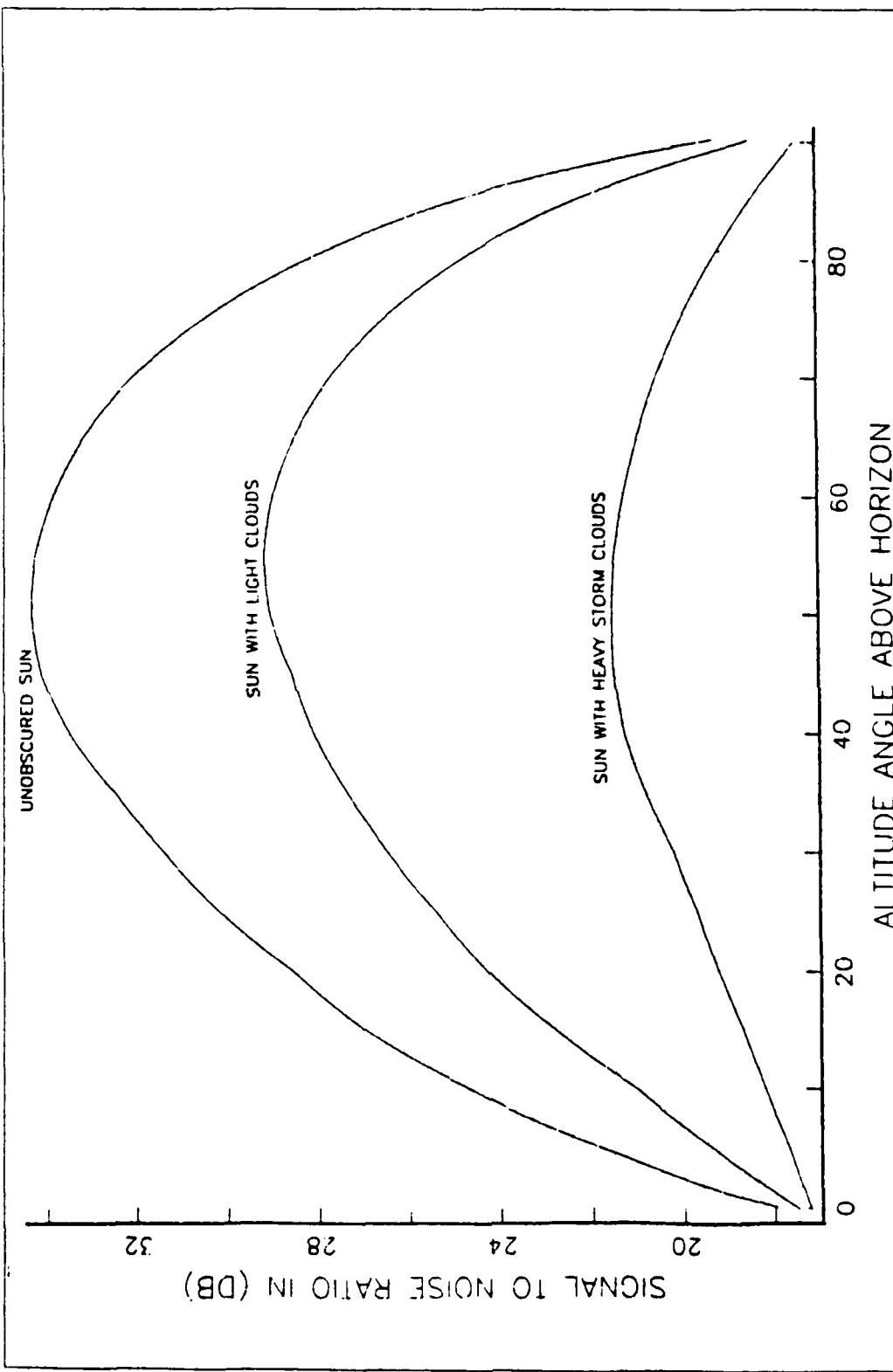


Figure D.7 and S/N Ratio for Tropical, $\epsilon_s = 0.55$
and Flight Altitude 50m

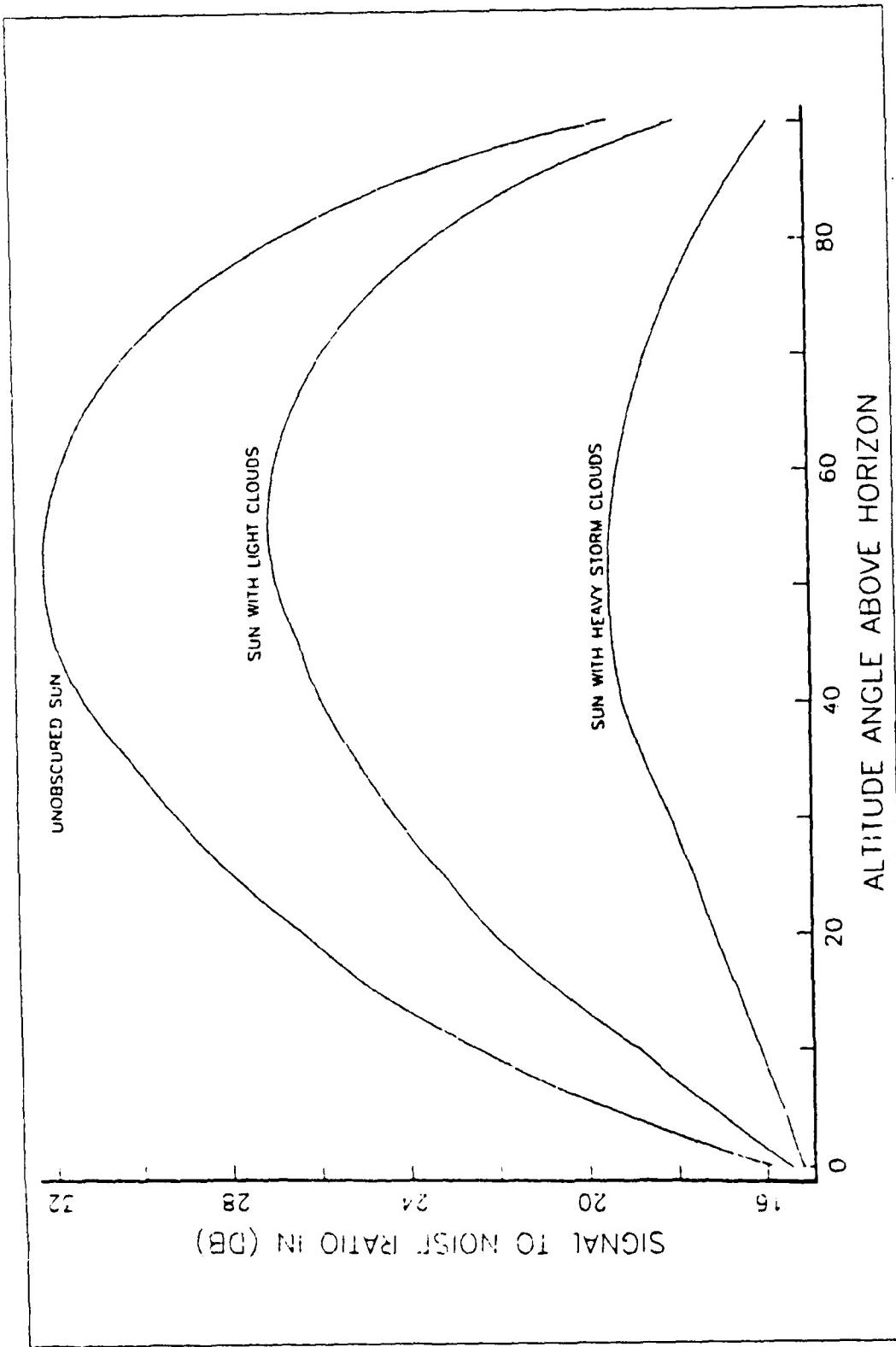


Figure D.8 S/N Ratio for Tropical, $\tau = 0.55$
and Flight Altitude 100m

APPENDIX E
COMPUTER PROGRAMS

```

$JOB
*****  

PROGRAM SHIP1
*****  

C   CALCULATION OF SIDE OF INSTANTANEOUS FIELD OF VIEW  

C   DIMENSION H(6),PHI(7),DELTA(21),SIDE(6,7,21),PATH(6,7,21)  

C   DATA (H(I), I=1,6)/50.,60.,70.,80.,90.,100./  

C   DATA (PHI(K), K=1,7)/2.0,2.5,3.0,3.5,4.0,4.5,5.0/  

C   DATA (DELTA(L), L=1,21)/32.,34.,36.,38.,40.,42.,44.,46.,48.,50.,52.,54.,56.,58.,60./  

C  

C   DO 50 I=1,6  

C      DO 45 K=1,7  

C         DO 40 L=1,21  

C            TH=(90.-PHI(K)-DELTA(L))*3.141592/100.  

C            B=I(I)*TAN(TH)  

C            C=(H(I)*H(I)+B*B)**.5  

C            D=DELTA(L)*3.141592/100.  

C            SIDE(I,K,L)=2.*C*SIN(D)  

C            PATH(I,K,L)=SIDE(I,K,L)/(2.*TAN(D))  

C        CONTINUE  

C 45    CONTINUE  

C        WRITE(6,39) H(I)  

39    FORMAT(6,1X,'SIDE OF INSTANTANEOUS FIELD OF VIEW IN METERS'//  

15X,'FOR FLIGHT HEIGHT=' ,F4.0,1X,'METERS'//)  

38    WRITE(6,38) (PHI(K), K=1,7)  

38    FORMAT(2X,7(1X,F4.2,2X)//)  

34    DO 35 L=1,21  

34    WRITE(6,34) DELTA(L), (SIDE(I,K,L), K=1,7)  

34    FORMAT(6,1X,F4.2,1X,7F10.2)  

35    CONTINUE  

C  

C 49    WRITE(6,49) H(I)  

49    FORMAT(6,1X,'PATH IN METERS'//5X,'FOR FLIGHT HEIGHT=' ,F4.0,  

1      1X,'METERS'//)  

48    WRITE(6,48) (PHI(K), K=1,7)  

48    FORMAT(2X,7(1X,F4.2,2X)//)  

47    DO 47 L=1,21  

47    WRITE(6,47) DELTA(L), (PATH(I,K,L), K=1,7)  

47    FORMAT(6,1X,F4.2,1X,7F10.2)  

50    CONTINUE  

50    CONTINUE  

50    STOP  

50    END
SENTRY

```

5 JOS

PROGRAM SHIP THERMAL SIGNALS

```

C 10.2046,J.143d,J.1127,J.0024/
C
C      DC 3 K=1,5
C      DC 2 I=1,18
C      JN=5*I+1
C      N=5*(I-1)+1
C      DO 1 L=N,JN
C      ES(K,L)=ES(K,N,I)-(ES(K,JN)-ES(K,N))/5*FLOAT(JN-L)/5.
C      D(L)=D(JN)-(J(N)-D(I))*FLOAT(N-L)/5.
C      CONTINUE
C      CONTINUE
C
C      CALCULATION OF NORMAL COMPONENTS OF ES
C
C      DO 6 I=1,2
C      DC 5 K=1,5
C      DC 4 L=1,11
C      TH=D(L)*E1/1E0
C      IF(I.EQ.1) A=3*IN(I)
C      IF(I.EQ.2) A=3*IN(I)
C      ESN(I,K,L)=ES(K,L)
C      HDLR(I,K,L)=(9.34E-07)*ESN(I,K,L)
C      CONTINUE
C      CONTINUE
C      CONTINUE
C
C      WRITE(6,7)
C      FORMAT(1X,5X,'SCALAR CONSTANT IS IN W/E2'//)
C      WRITE(5,8)
C      FORMAT(1X,'ALBEDO',1X,'UNCOLOURED SUN',2X,'SUN WITH LIGHT CLOUDS')
C      15S*,2X,5X,'WITH HEAVY CLOUDS')
C      DO 10 L=1,11
C      WRITE(6,9) D(L),ES(I,L),K=1,5
C      FORMAT(1X,F0.5,5X,F0.2,5X,F10.2)
C      CONTINUE
C      WRITE(5,11)
C      FCH4A1(1X,'NORMAL COCOMPONENT OF RADIANCE DUE TO REFLECTED SCALAR EN
C      ERGY',//1X,'IN 4/(C12*SE) IS THE WINDOW OF 5 MICROMETRES'//)
C      WRITE(5,12)
C      FORMAT(1X,'UNCOLOURED SUN',5X,'SUN WITH LIGHT CLOUDS',2X,
C      'SUN WITH HEAVY CLOUDS')
C      WRITE(6,13)
C      FORMAT(1X,'ALTITUDE',1X,5('HORIZONTAL',2X,'VERTICAL',2X))
C      WRITE(5,14)
C      FORMAT(1X,'ANGLE',1X,5('FACE',5X,'SAFCE',5X)//)
C      DO 16 L=1,11
C      WRITE(6,15) D(L),(DLR(I,L),I=1,4),(HDLR(I,L),I=1,2),
C      15 FORMAT(1X,F4.0,5X,5F0.4,12,F10.4,1X)
C      16 CONTINUE
C
C      CALCULATION OF SHIP BODY TEMPERATURE
C
C      DO 20 I=1,2
C      DC 19 J=1,2
C      DO 18 K=1,5
C      DO 17 L=1,11
C      T04=TIN(J)**4
C      ESN(I,K,L)=(A/3.)*ESN(I,K,L)
C      A=ESN(I,K,L)*AC+ES*33*T04+EIN*SE*MIN4
C      D=(20+EL1)*SE
C      XK=(A/2)**0.23
C      T4=AK**4
C      P1=SE*SE*(T4-T04)+EIN*SE*(T4-TIN4)+HCINV*(XK-10)-ESN(I,K,L)*AC
C      XA=KK+F1
C      T4=XA**4
C      P2=SE*SE*(T4-T04)+EIN*SE*(X4-TIN4)+HCINV*(KA-T0)-ESN(I,K,L)*AC
C      XK1=XK-F1*(F2-F1)
C      DI2=ABS(XK1-XK)
C      XK=XK1

```

```

      IF(DIF.GT.000001) GO TO 2000
      T(I,J,K,L)=XK
      CONTINUE
17    CONTINUE
18    CONTINUE
19    CONTINUE
20    CONTINUE

C
C      DO 27 J=1,2
C      WRITE(6,21) TIN(J)
C 21    FORMAT(1X,'SHIP BODY TEMPERATURE IN K FOR TIN=',F7.2,1X,
C           1X///)
C 22    FORMAT(1X,'X,'UNOCCURED SUN',JX,'SUN WITH LIGHT CLOUDS',2X,
C           1X,'SUN WITH HEAVY SURFACE CLOUDS')
C 23    FORMAT(1X,'ALTITUDE',1X,3('MATERIAL',2X,'VERTICAL',2X))
C 24    FORMAT(1X,'ANGLE',4X,3('PACE',3X,'FACE',3X)//)
C 25    DO 26 L=1,31
C 26    WRITE(6,25) J(L), (T(I,J,L), I=1,2), (T(I,J,2,L), I=1,2),
C           1      FCBSAT(1X,F6.1,3(F10.2,1X,F10.2,1X))
C 27    CONTINUE
C
C      CALCULATION OF NORMAL COMPONENT OF SPECTRAL RADIANCE IN THE
C      WINDOW OF 5 MICRAMETER (I.E. 4.435-5.000)
C
C      DO J2, I=1,2
C      DO 31 J=1,2
C          DO 30 K=L,1
C              DO 29 L=1,J
C                  S1=0.0
C                  S2=0.0
C                  X1=C2/(T(I,J,K,L)*4.335E-04)
C                  X2=C2/(T(I,J,K,L)*5.000E-04)
C                  DC=28-S1
C                  A1=FLUAT(S1)*X1
C                  A2=FLICAT(S1)*X2
C                  B=FLICAT(S1)**4.
C                  S1=S1+EXP(-A1)*(A1**3.+3.*A1*A1+b.*A1+b.)/B
C                  S2=S2+EXP(-A2)*(A2**3.+3.*A2*A2+c.*A2+c.)/B
C 23      CONTINUE
C                  HDL(I,J,K,L)=C1*EW*(T(I,J,K,L)**4.)*(S2-S1)/(2I*(C2**4.))
C                  HDL(I,J,K,L)=HDL(I,J,K,L)+HDL(I,K,L)
C 29      CONTINUE
C 30      CONTINUE
C 31      CONTINUE
C 32      CONTINUE

C
C      CALCULATION OF NORMAL COMPONENT OF SEA RADIANCE IN THE
C      WINDOW OF 5 MICRAMETER (I.E. 4.435-5.000)
C
C      S1=0.0
C      S2=0.0
C      X1=C2/(TW**4.335E-04)
C      X2=C2/(TW**5.000E-04)
C      DO 128 M=1,8
C          DO 129 N=1,M
C              A1=FLUAT(M)*X1
C              A2=FLUAT(N)*X2
C              B=FLICAT(S1)**4.
C              S1=S1+EXP(-A1)*(A1**3.+3.*A1*A1+b.*A1+b.)/B
C              S2=S2+EXP(-A2)*(A2**3.+3.*A2*A2+c.*A2+c.)/B
C 128      CONTINUE
C              HDLw=C1*EW*(TW**4.)*(S2-S1)/(FI*(C2**4.))

C
C      DO 40 J=1,2
C      WRITE(6,33)
C 33    FORMAT(1X,'NORMAL COMPONENT OF RADIANCE IN W/(CM2*SR)',//1X,
C           1      'DUE TO THERMAL RADIATION')

```

```

C      WRITE(6,34),FIN(J)
C 34   FORMAT(2X,'THE WINDOW OF 5 MICROMETER FOR TIM=' ,F7.2,1X,'E'///)
C      WRITE(6,35)
C 35   FORMAT(10' 4X,'UNOBSCURED SUN',3X,'SUN WITH LIGHT CLOUD',2X,
C           1,'SUN WITH HEAVY STORM CLOUDS')
C      WRITE(6,36)
C 36   FORMAT(1X,3('HORIZONTAL',2X,'VERTICAL',2X))
C      WRITE(6,37)
C 37   FORMAT(1Z,4X,'ANGLE',4X,3('FACE',3X,'FACE',6X)//)
C      DO 39 L=1,91
C 38   1   WRITE(6,38) D(L),(HDL(I,J,1,L),I=1,2),(HDL(I,J,2,L),I=1,2),
C           1   FORMAT(14,F6.1,2X,3(E15.4,1X,E15.4,1X))
C 39   CONTINUE
C 40   CONTINUE
C
C      WRITE(6,138),TW,HDL
C 138  FORMAT(1',2X,'SEA RADIANCE FOR TW=' ,F6.2,2X,E15.4)
C
C      CALCULATION OF SHIP SURFACES AND THEIR CORRESPONDING MEAN ELEVATIONS
C      FROM THE SEA INSIDE THE INSTANTANEOUS DETECTOR FIELD OF VIEW IN THE
C      REGION OF THE BIGGEST RADIANT INTENSITY.
C
C      A1=S1*H1
C      A11=(SIDE-H1)*H1
C      H11=H1/2.
C
C      A2=S1*H2
C      A22=(SIDE-H1)*H2
C      H22=H1
C
C      A3=SIDE*H3
C      H33=H1+H3/2.
C
C      H44=H1+H3+H4/2.
C
C      A5=H5*H5
C      H55=H1+H3+H4+H5/2.
C
C      H66=H1+H3+H4+H5
C
C      A7=SIDE*H7-P1*W4**4/4.
C      H77=H1+H3
C
C      CALCULATION OF THE RADIANT INTENSITY IN (W/SR) OF THE
C      SURFACES (A1+A11), (A2+A22), A3, A4, A5, A7 IN THE WINDOW OF
C      5 MICROMETERS, IN THE DIRECTION SHIP-MISSILE
C
C      DO 42 K=1,3
C          DO 41 L=1,91
C              RI1N=HDL(2,2,K,L)*A1+HDL(2,1,K,L)*A11
C              C11=(HEIGH1-HM1)/PAIH
C              C1=(-C11*C11)**0.5
C              BI1(K,L)=RI1N*C1
C              RI2N=HDL(1,2,K,L)*A2+HDL(1,1,K,L)*A22
C              C22=(HEIGH1-HM2)/PAIH
C              RI2(K,L)=RI2N*C22
C              RI3N=HDL(2,1,K,L)*A3
C              C33=(HEIGH1-HM3)/PAIH
C              C3=(-C33*C33)**0.5
C              BI3(K,L)=RI3N*C3
C              RI4N=HDL(2,2,K,L)*A4
C              C44=(HEIGH1-HM4)/PAIH
C              C4=(-C44*C44)**0.5
C              RI4(K,L)=RI4N*C4
C              RI5N=HDL(2,1,K,L)*A5
C              C55=(HEIGH1-HM5)/PAIH
C              C5=(-C55*C55)**0.5
C              RI5(K,L)=RI5N*C5
C              C66=(HEIGH1-HM6)/PAIH
C              RI6=d16N*C66
C              RI7N=HDL(1,1,K,L)*A7

```

```

C77=(HEIGHT-847)/PATH
41    CONTINUE
42    CONTINUE
C   CALCULATION OF SHIP SIGNAL VOLIAGE
C
DO 50 K=1,3
DO 49 L=1,1
VS(K,L)=RI1(K,L)+RI2(K,L)+RI3(S,L)+RI4(K,L)+RI5(K,L)+RI6+
1      RI7(K,L))+TRANS*RESPON*0.41*APER/(PATH+FAITH)
49    CONTINUE
50    CONTINUE
C   CALCULATION OF NCISE VOLTAGE
C
UMINST=SIDE*SIDE/(PATH*FAITH)
AD=UMINST*FCCL*FOLL
OMTOT=(SIDE/PATH)*ALOG((PATH+XTOT)/(PATH-XTOT))
DFREL=XTOT/(2.*UMINST*EN*TIME)
VNOP1=2*SECN*(AD*DFREL)**.5)/LEFECT
VB=DDL*AREAS*(TRANS*RESPON*J.A1*AFEST)/(PATHS*PATHS)
VNTOT=VNOP1*VB
C   CALCULATION OF SIGNAL TO NOISE RATIO
C
DO 60 K=1,1
DO 59 L=1,1
SNR(K,L)=20.*4LOG10(VS(K,L)/VNTOT)
59    CONTINUE
60    CONTINUE
C   TABULATION OF THE DETECTOR SIGNAL TO NCISE RATIO
C
WRITE(6,70)
70  FORMAT(1X,'DETECTOR SIGNAL TO NOISE RATIO',
1/8X,'FOR FLIGHT HEIGHT=50 MEIERS AND SIDE=21.08 METERS//')
WRITE(6,71)
71  FORMAT(1X,'ALT. ANGLE',3X,'OBSCURED SUN',3X,'SUN WITH LIGHT CLOS
1D.',3X,'SON WITH HEAVY STORM CLOUDS//')
DO 73 L=1,21
    WRITE(6,72) L,(SNR(K,L),K=1,3)
72  FORMAT(1A,F0.0,3X,F10.3,7X,F10.3,13X,F10.3)
73  CONTINUE
STOP
END
SENTRY

```

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